

Engineering/Design of the Welded Inner Leg Interfaces
for the NCSX Modular Coils A-A, A-B, B-C PDR

Presented by
T Brown, M Cole, K Freudenberg, D Williamson, and M Viola
October 18, 2007

- Are the requirements defined? What is the proposed design?
- What is the status of welding trials?
- Is the analysis consistent with proposed design?
- What is the plan to complete final design?
- Have prior design review chits been addressed?
- Have all technical, cost, schedule, and safety risks been addressed?

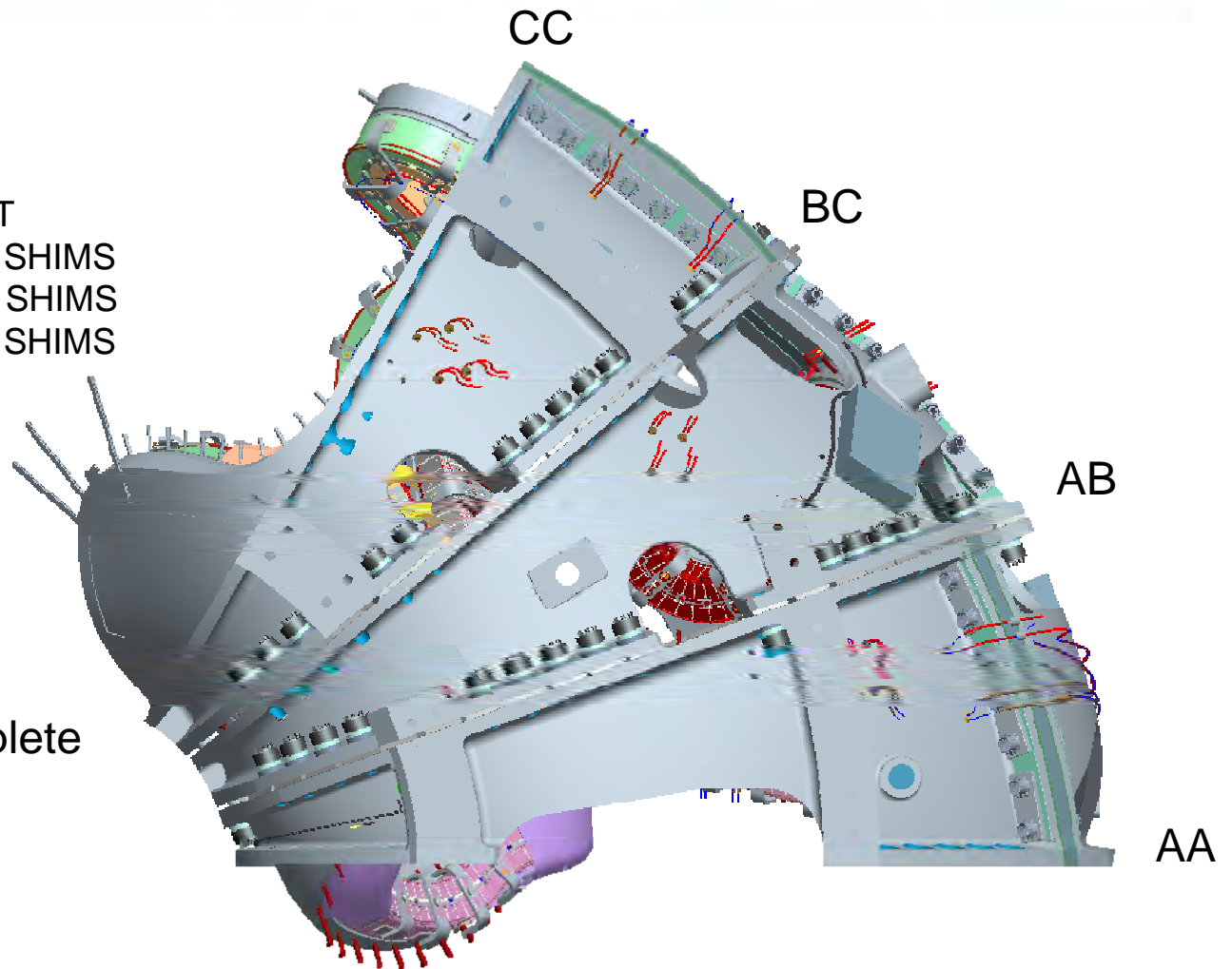
- This review-
AA/AB/BC interface:

SE140-046, SHIM LAYOUT
SE140-052, AB INBOARD SHIMS
SE140-053, BC INBOARD SHIMS
SE140-054, AA INBOARD SHIMS

- Upcoming reviews:

AA/AB/BC Interface
FDR 11/22/07

CC Interface
PDR 8/7/07 Complete
FDR 1/7/08



Requirements are derived from the Modular Coil Asm Specification (NCSX-CSPEC-14-05-01) and the Station-2 Asm Specification (in progress).

Electrical

- Partial Toroidal electrical breaks shall be provided between adjacent modular coils within a field period (AA, AB, BC).
- Electrical breaks are required between adjacent modular coils in adjacent field periods (CC). [Ref. GRD Section 3.2.1.5.2b to be revised]
- Toroidal electrical breaks must be able to withstand an applied voltage of 150 V (ref. GRD Section 3.2.1.5.3.6).

Structural

- Carry compressive and shear loads

Assembly

- Position the coil fiducials to +/- .020 inch at the half period assembly

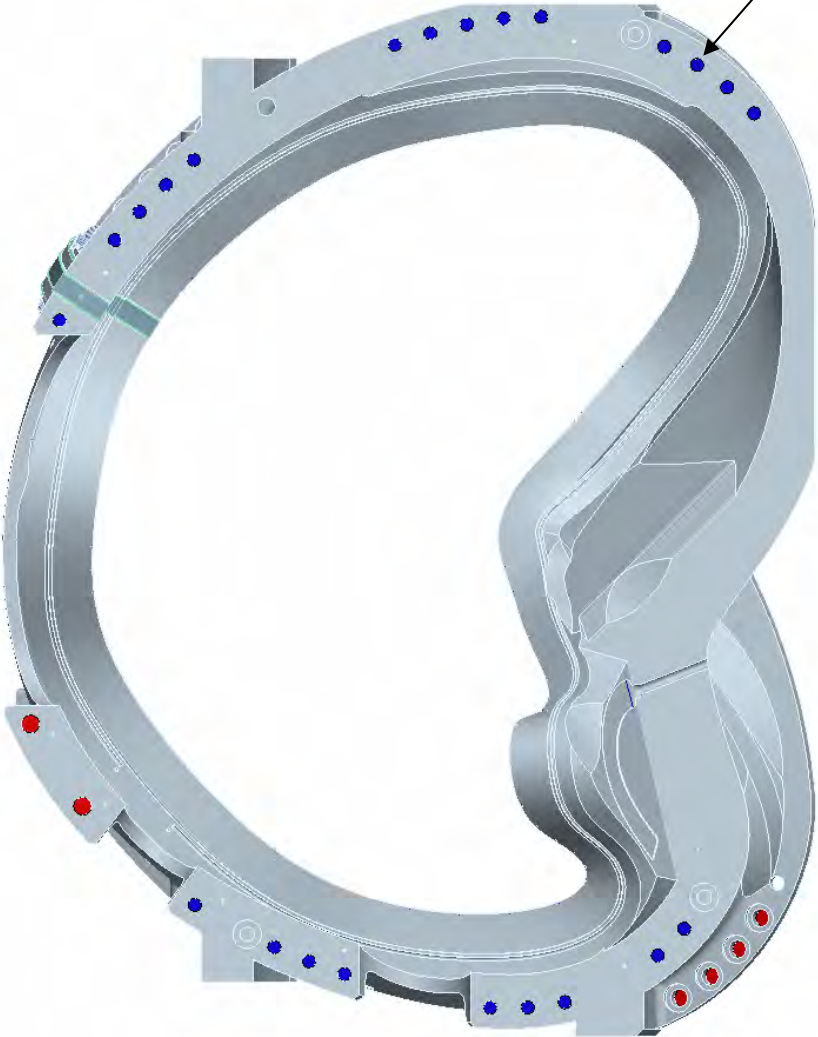
Interface A-B



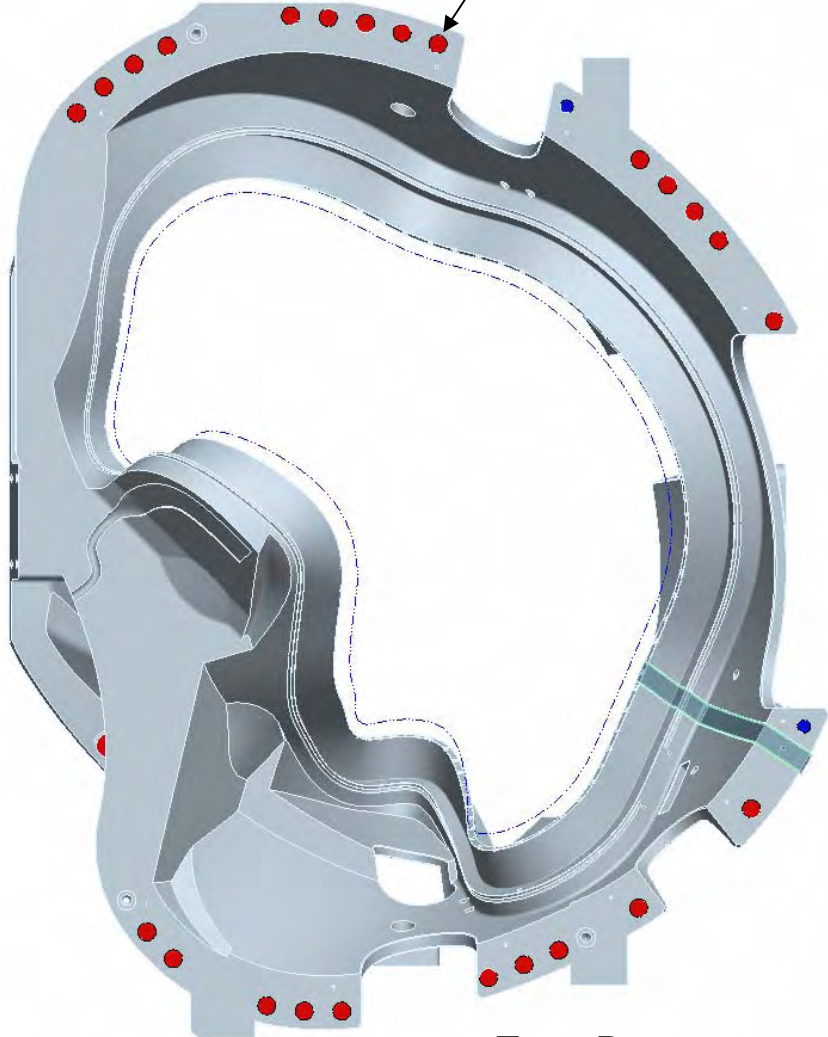
- 25 tapped holes, most on Type-A
- 1 through hole

$\varnothing 1.375-6UNC$ THRU OR
 $\varnothing 1.375-6UNC \times 1.5$ MIN
FOR FLANGE THK > 1.5
 $\varnothing .06$ M A D

$\varnothing 1.885 \pm .003$ THRU
 $\varnothing 3.00$ SPOTFACE BACKSIDE
MINIMUM TO CLEAN UP
 $\varnothing .06$ M A D



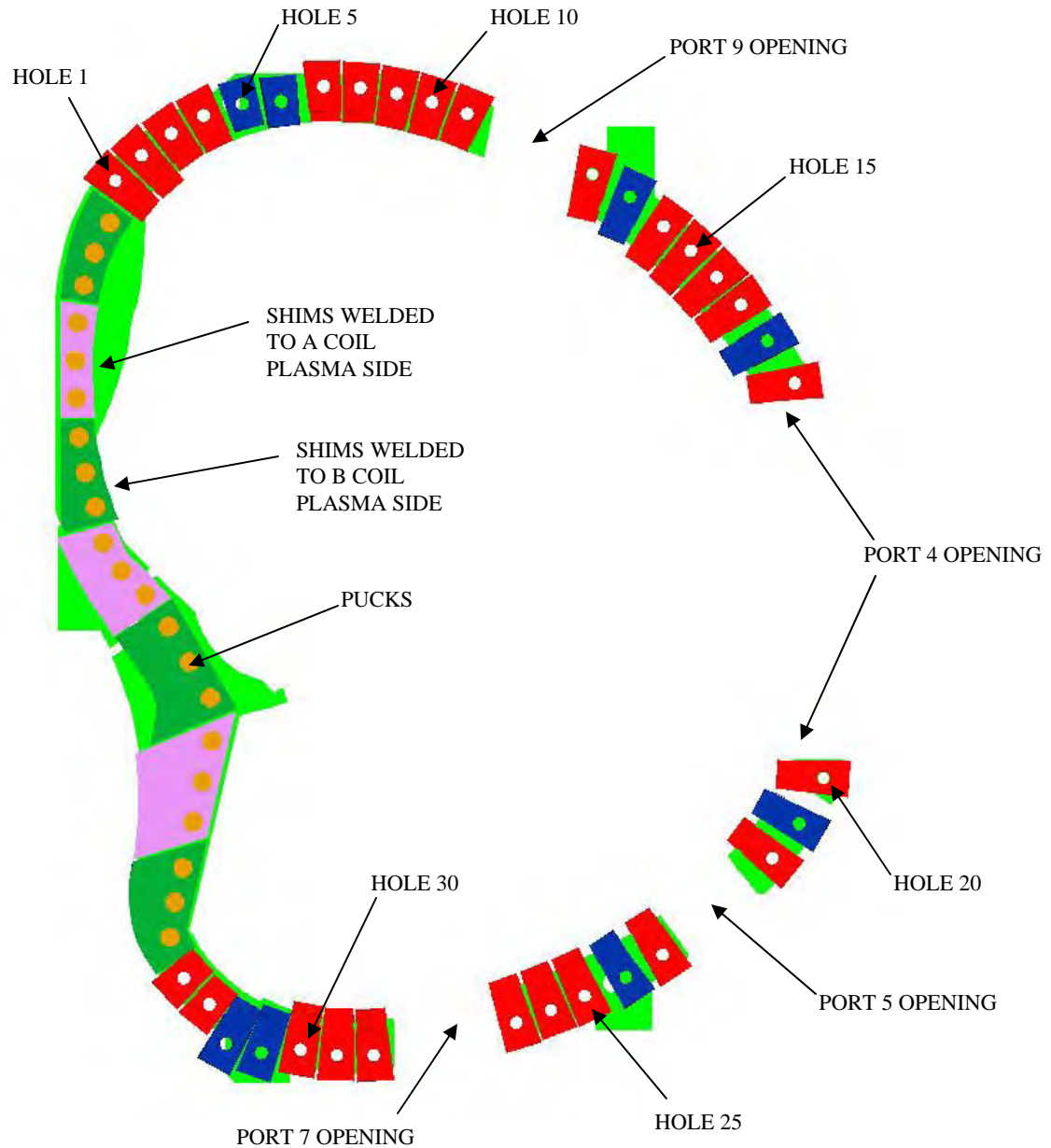
Type-A



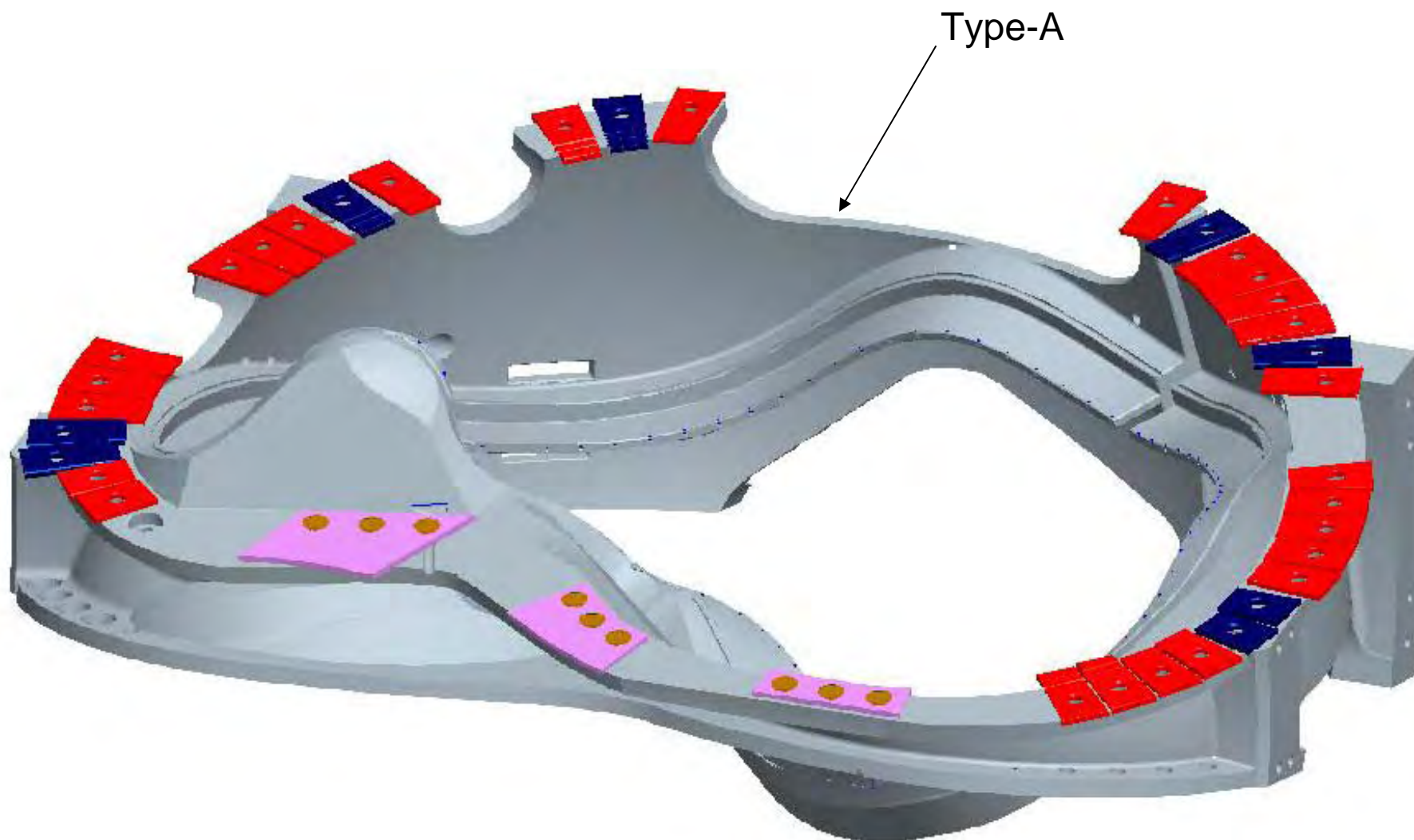
Type-B

A-B FLANGE

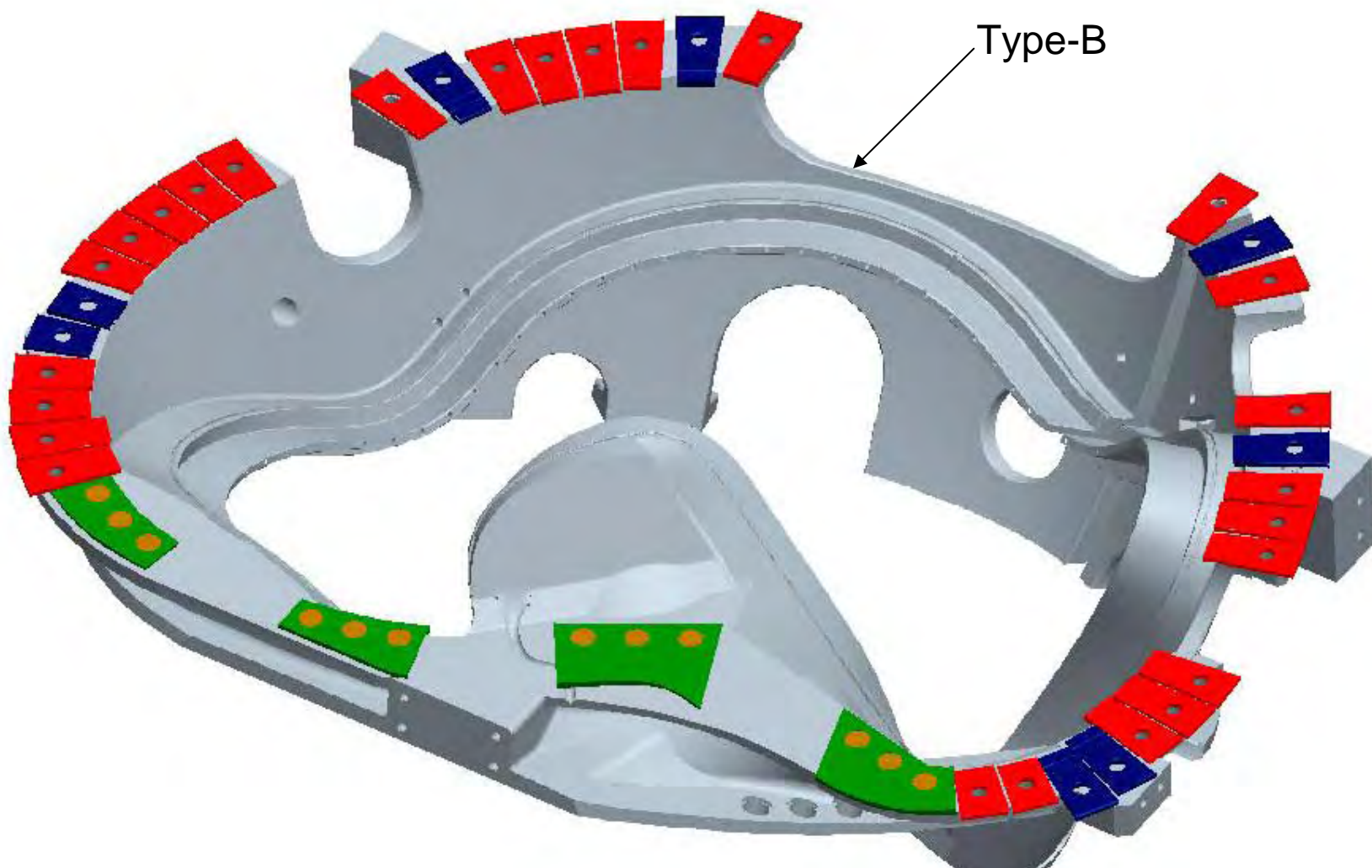
AB Hole #	Shim Length Hole to Bottom	No Bolt Shim
1	5.00	
2	5.00	
3	3.75	
4	3.75	
5		2.75
6		2.75
7	3.75	
8	3.75	
9	3.75	
10	3.75	
11	3.75	
12	5.00	
13		5.00
14	5.00	
15	5.00	
16	5.00	
17	5.00	
18		5.00
19	5.00	
20	5.00	
21		5.00
22	5.00	
23	5.00	
24		5.00
25	5.00	
26	5.00	
27	5.00	
28	5.00	
29	5.00	
30	5.00	
31		5.00
32		5.00
33	2.75	
34	2.75	

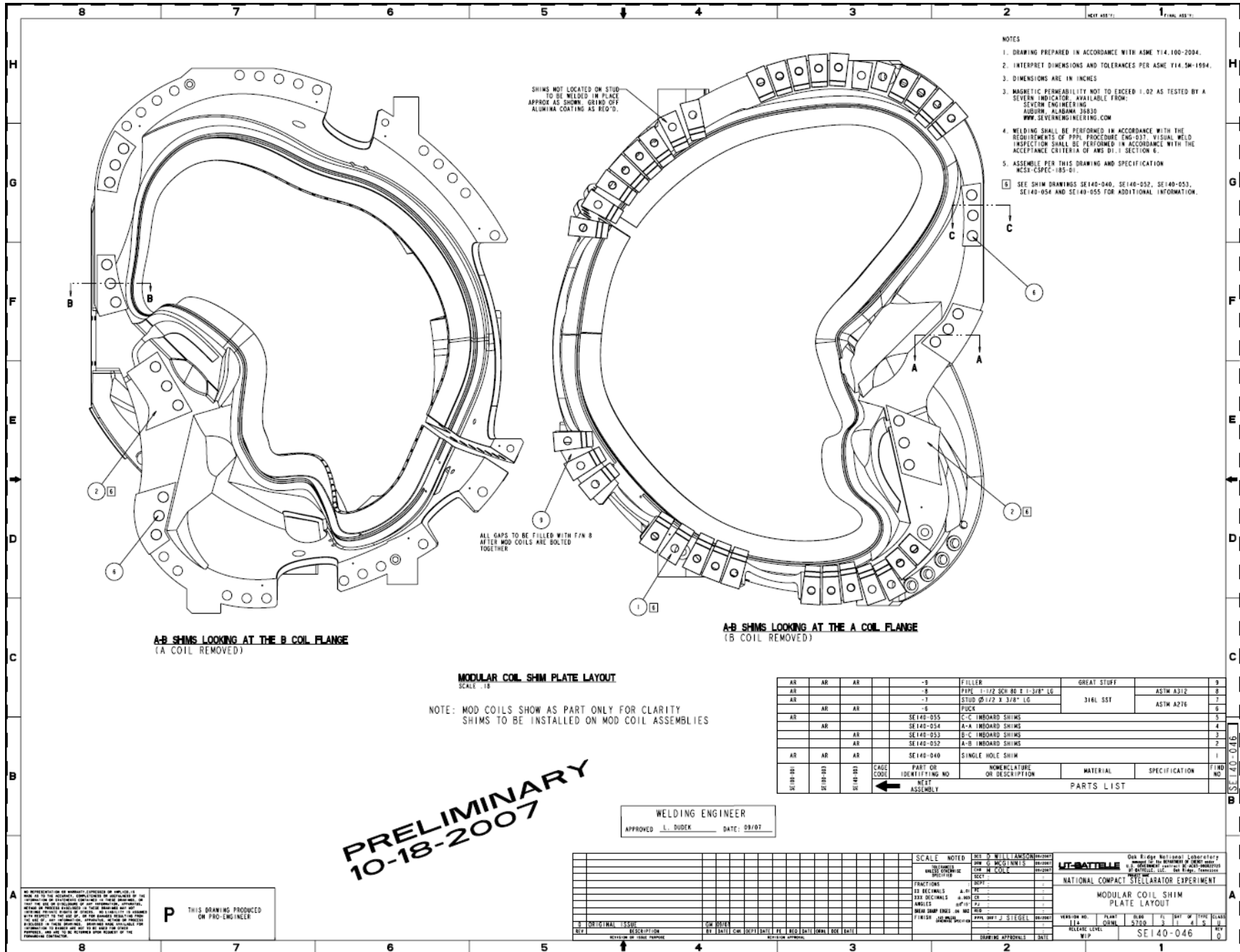


AB Inboard Welded Shims

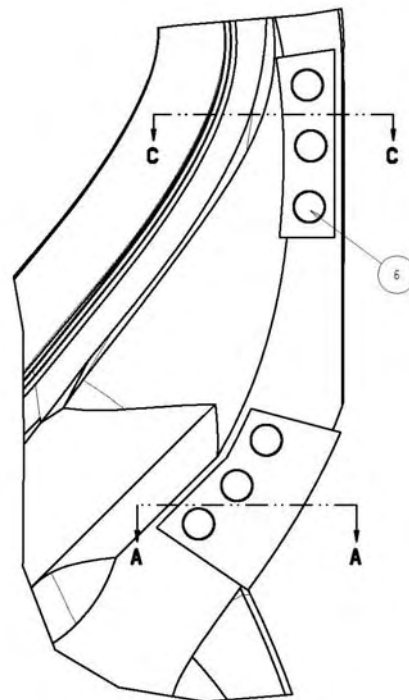
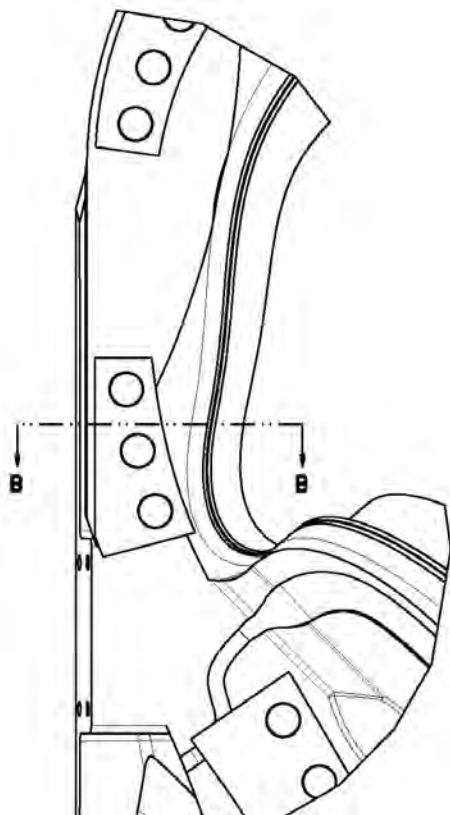
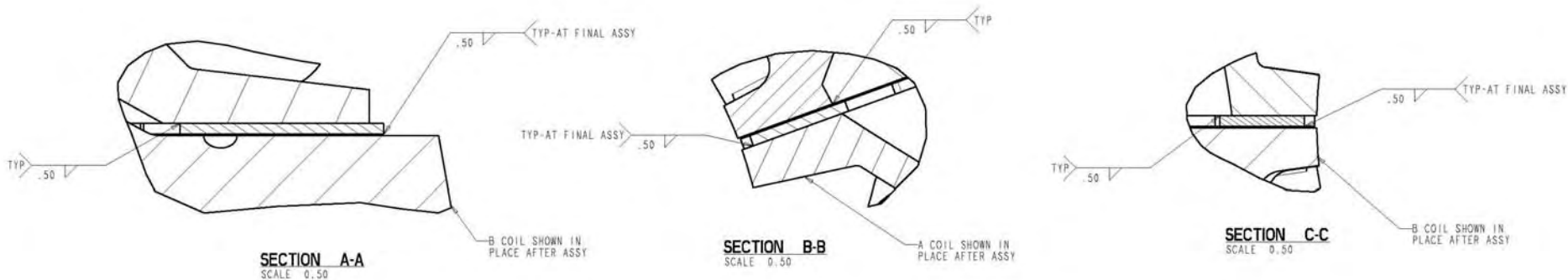


AB Inboard Welded Shims

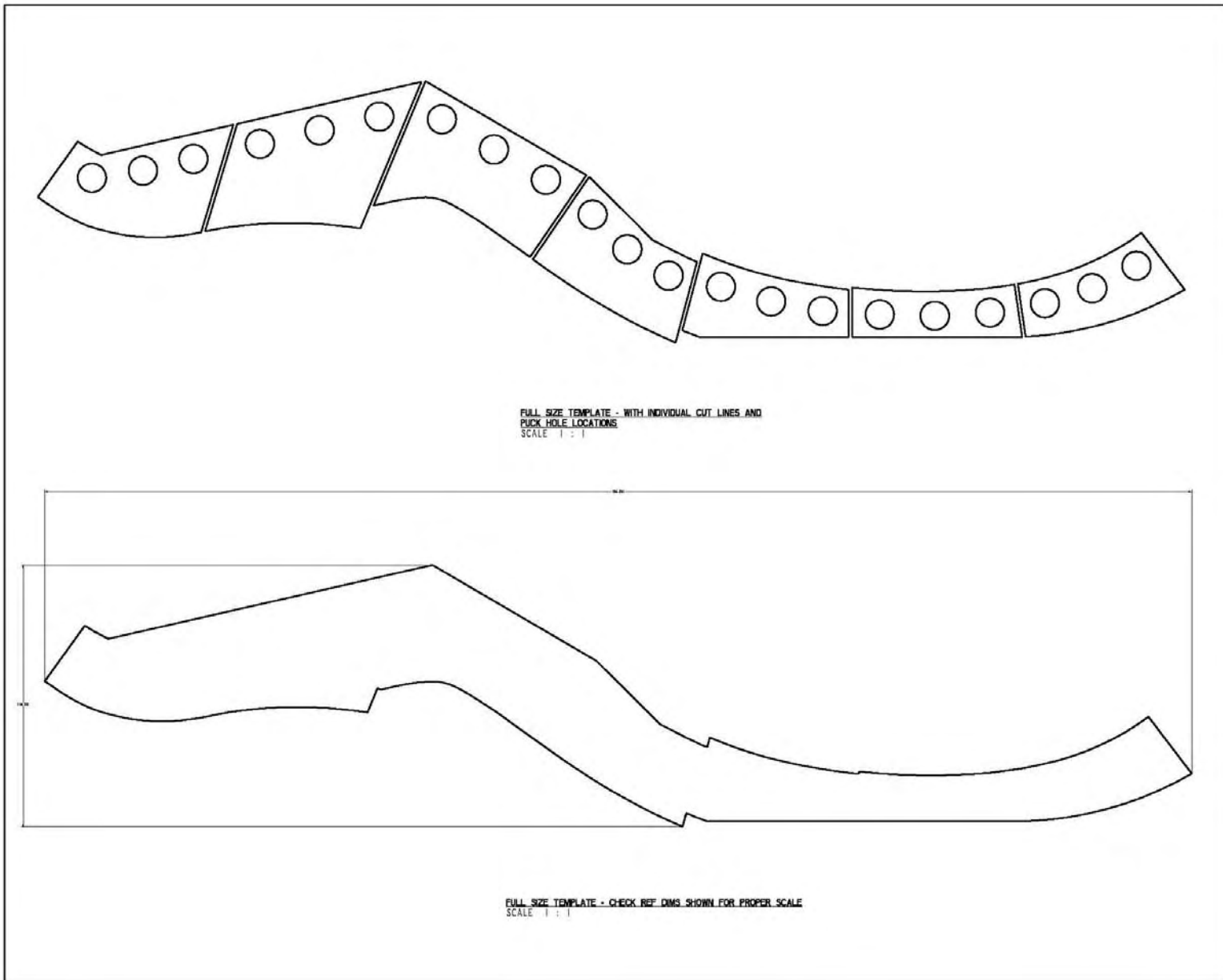




AB Inboard Welded Shims



AB Inboard Welded Shims (SE140-052)



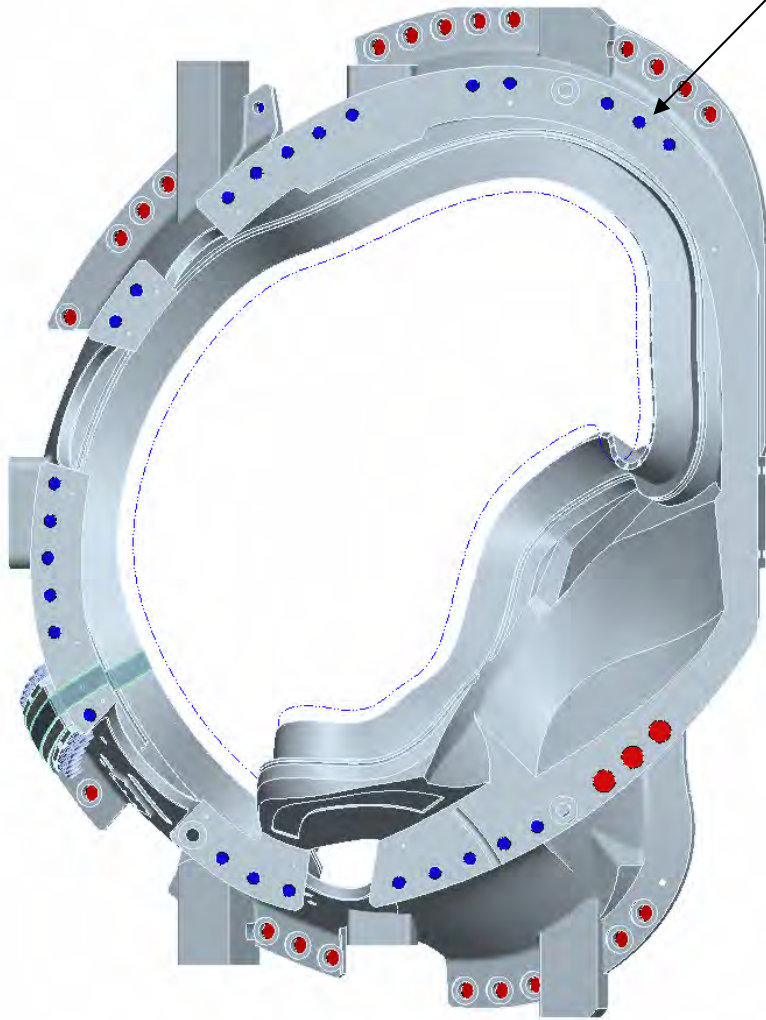
- 29 tapped holes, most on Type-B

$\varnothing 1.375-6UNC$ THRU OR
 $\varnothing 1.375-6UNC \times 1.5$ MIN
 FOR FLANGE THK > 1.5

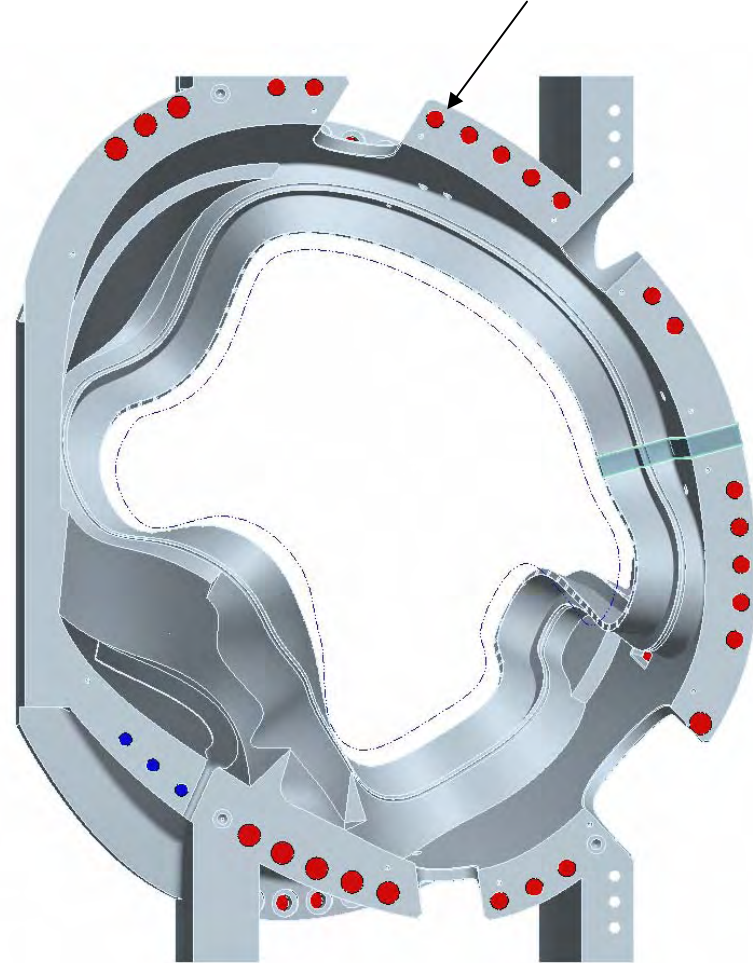
\varnothing	.06	M	A	D
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$\varnothing 1.885 \pm .003$ THRU
 $\sqsupset \varnothing 3.00$ SPOTFACE BACKSIDE
 MINIMUM TO CLEAN UP

\varnothing	.06	M	A	D
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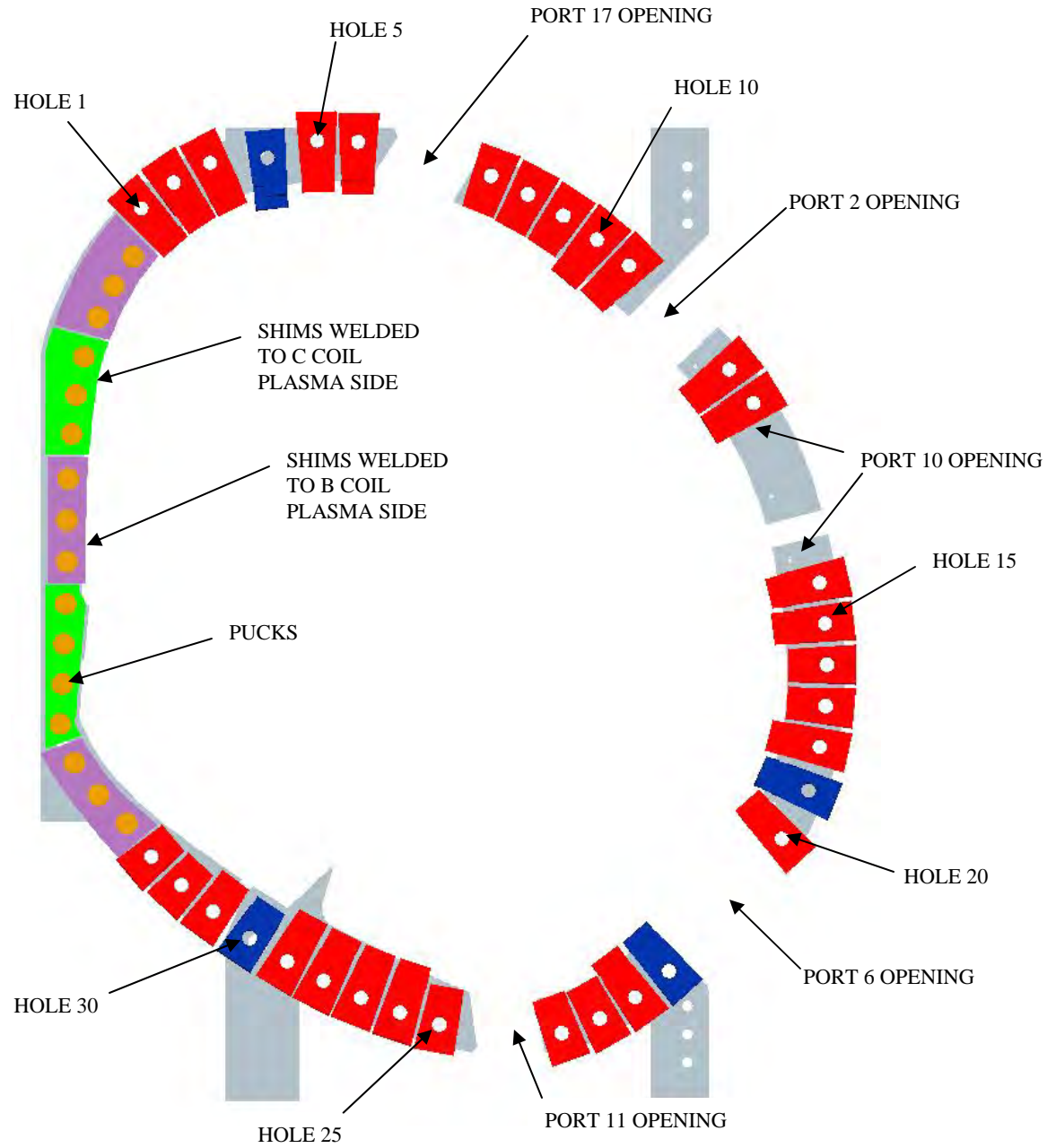
Type-B

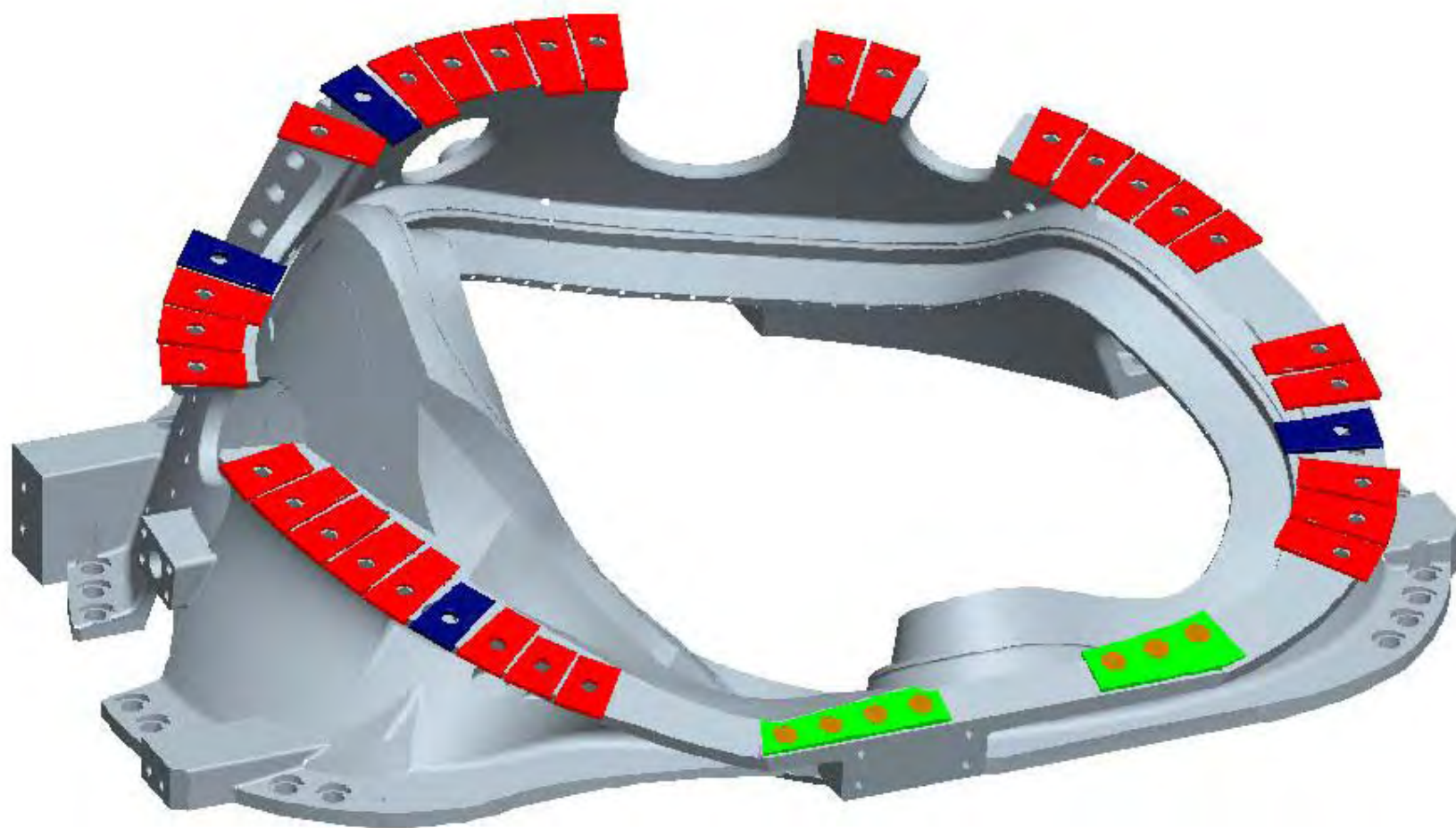


Type-C

B-C FLANGE

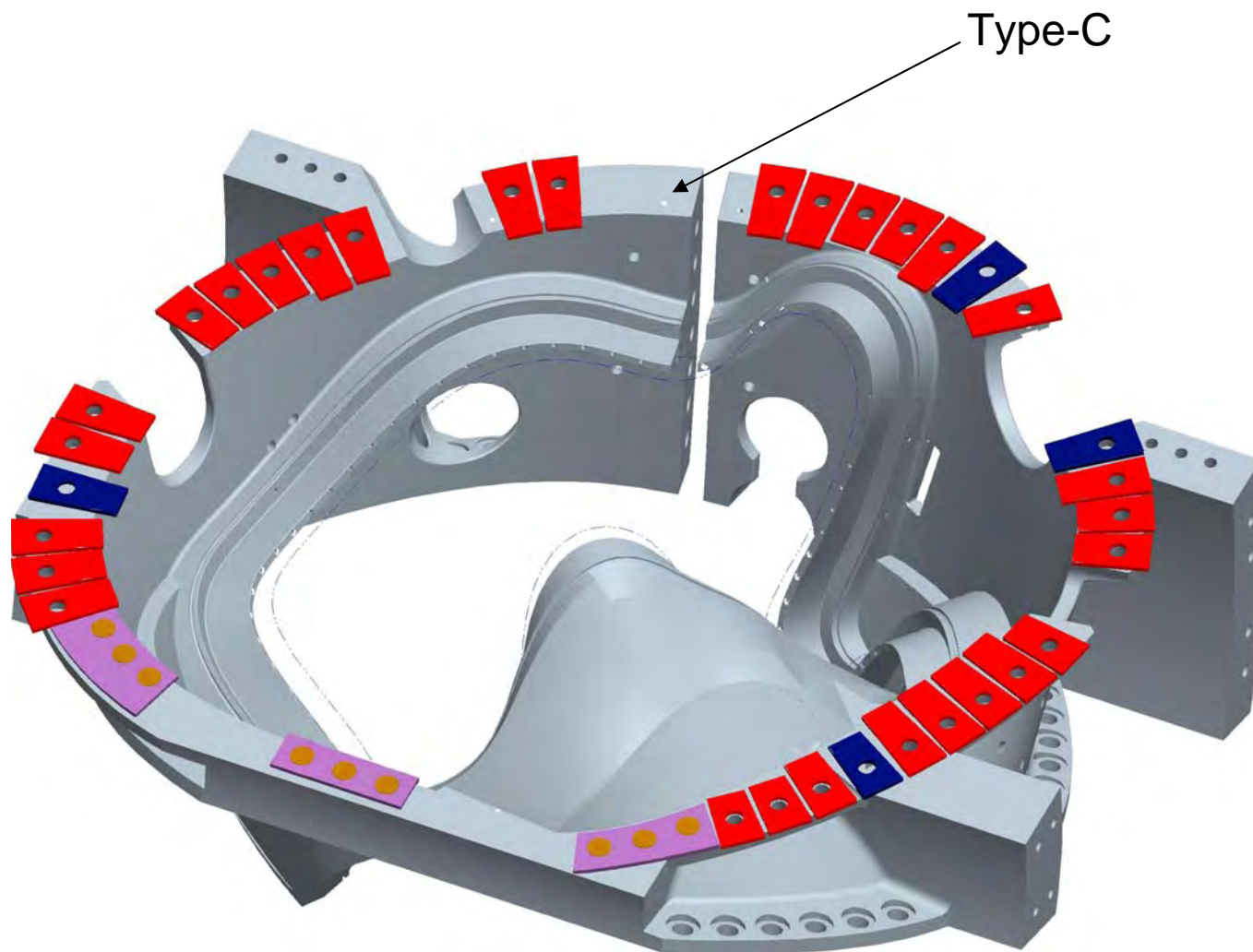
BC Hole #	Shim Length Hole to Bottom	No Bolt Shim
1	5.00	
2	5.00	
3	5.00	
4		5.00
5	5.00	
6	5.00	
7	3.75	
8	3.75	
9	3.75	
10	5.00	
11	5.00	
12	5.00	
13	5.00	
14	5.00	
15	5.00	
16	3.75	
17	3.75	
18	5.00	
19		5.00
20	5.00	
21		5.00
22	5.00	
23	3.75	
24	3.75	
25	3.75	
26	5.00	
27	5.00	
28	5.00	
29	5.00	
30		3.75
31	3.75	
32	2.75	
33	2.75	



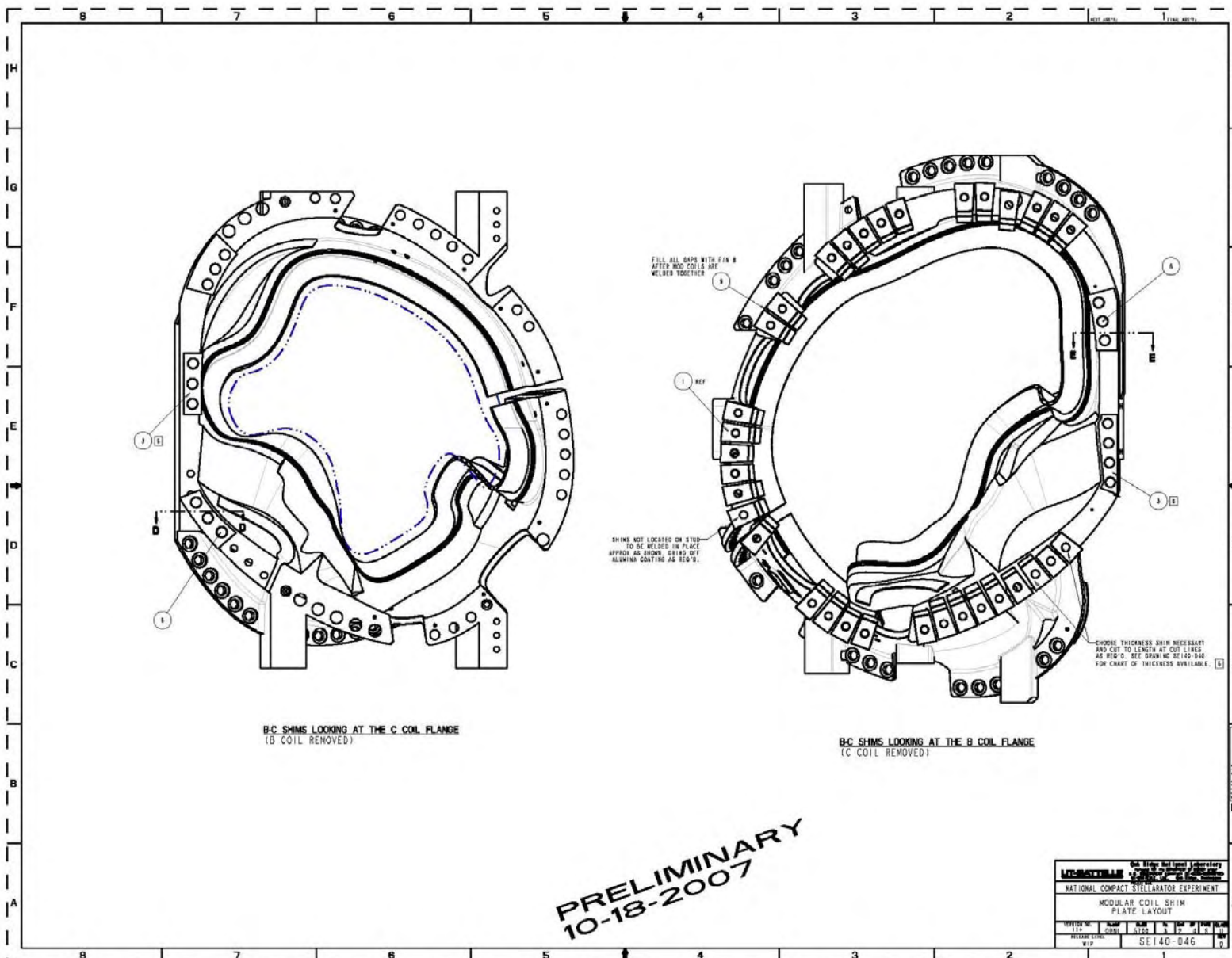


Type-B

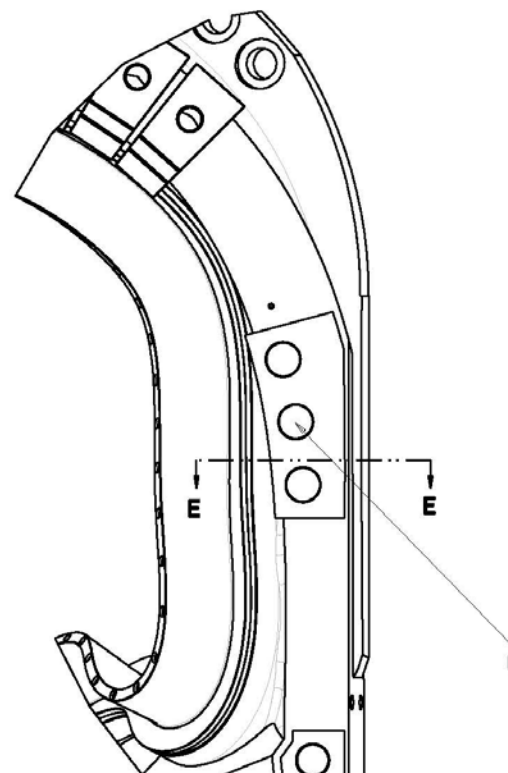
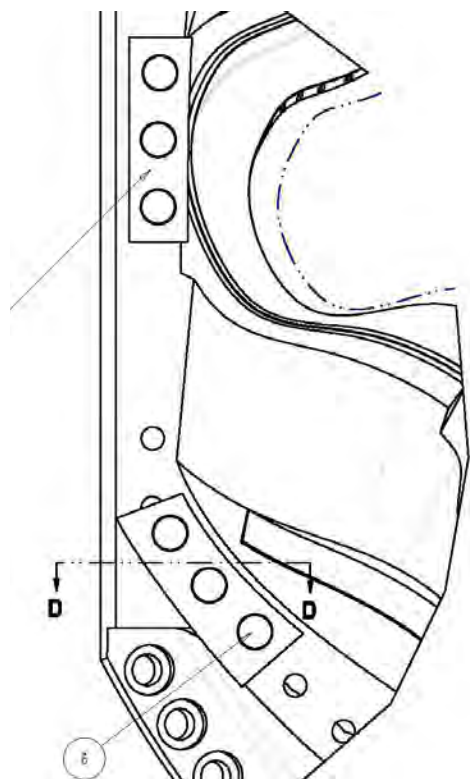
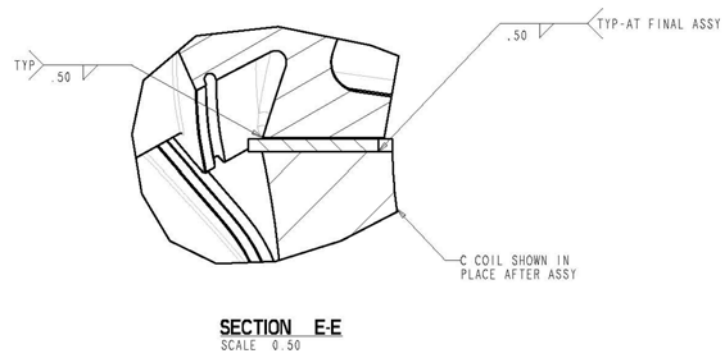
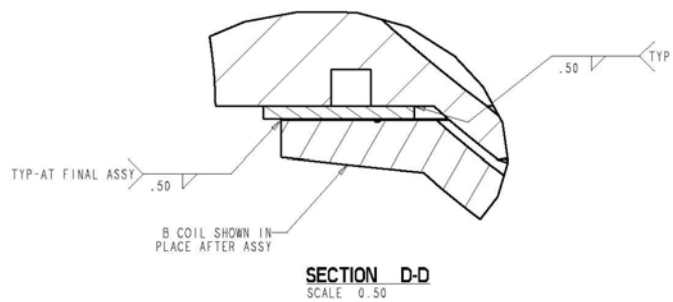
BC Inboard Welded Shims



BC Inboard Welded Shims



BC Inboard Welded Shims



Interface A-A

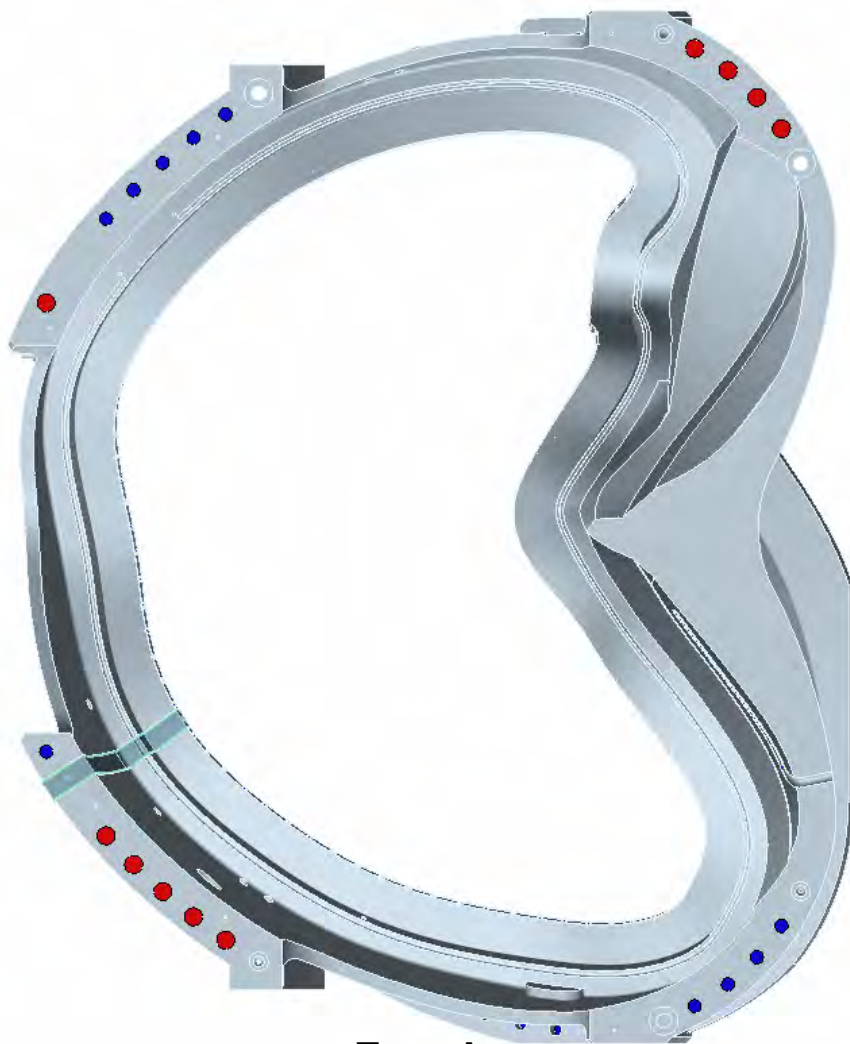
- 20 tapped holes

$\varnothing 1.375-6UNC$ THRU OR
 $\varnothing 1.375-6UNC \times 1.5$ MIN
 FOR FLANGE THK > 1.5

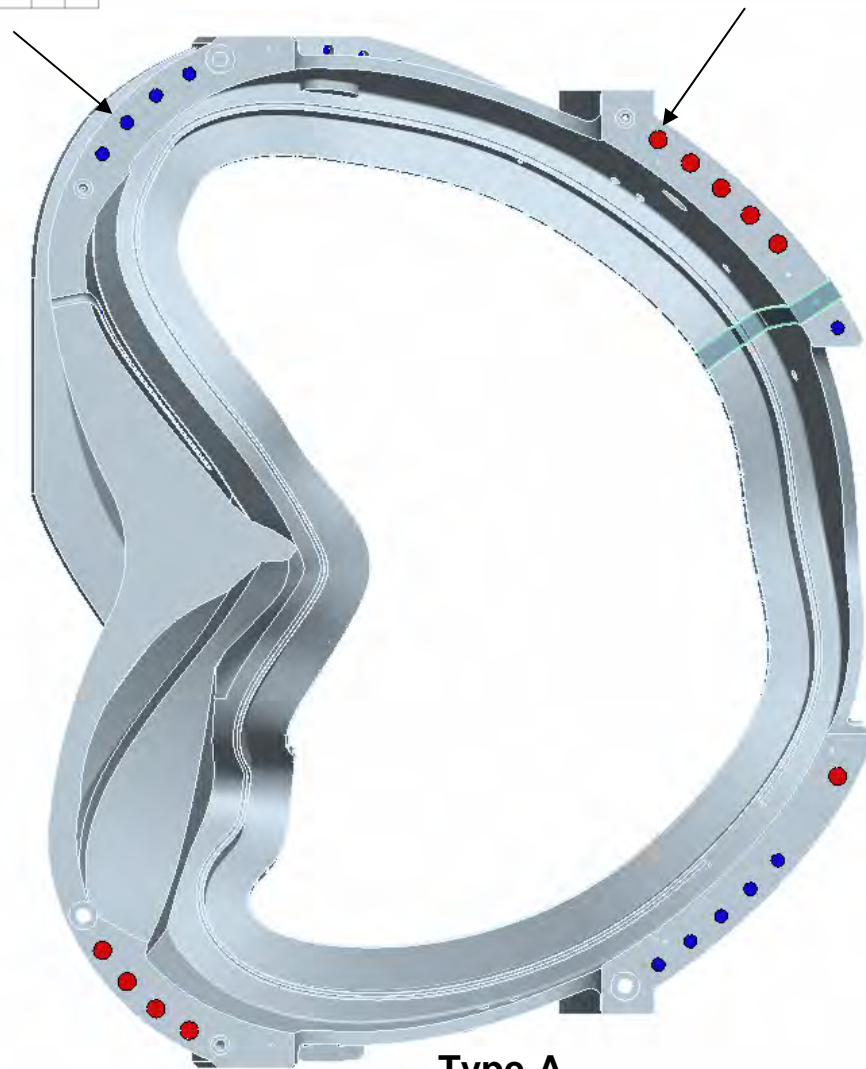
\varnothing	.06	M	A	D
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$\varnothing 1.885 \pm .003$ THRU
 $\sqsupset \varnothing 3.00$ SPOTFACE BACKSIDE
 MINIMUM TO CLEAN UP

\varnothing	.06	M	A	D
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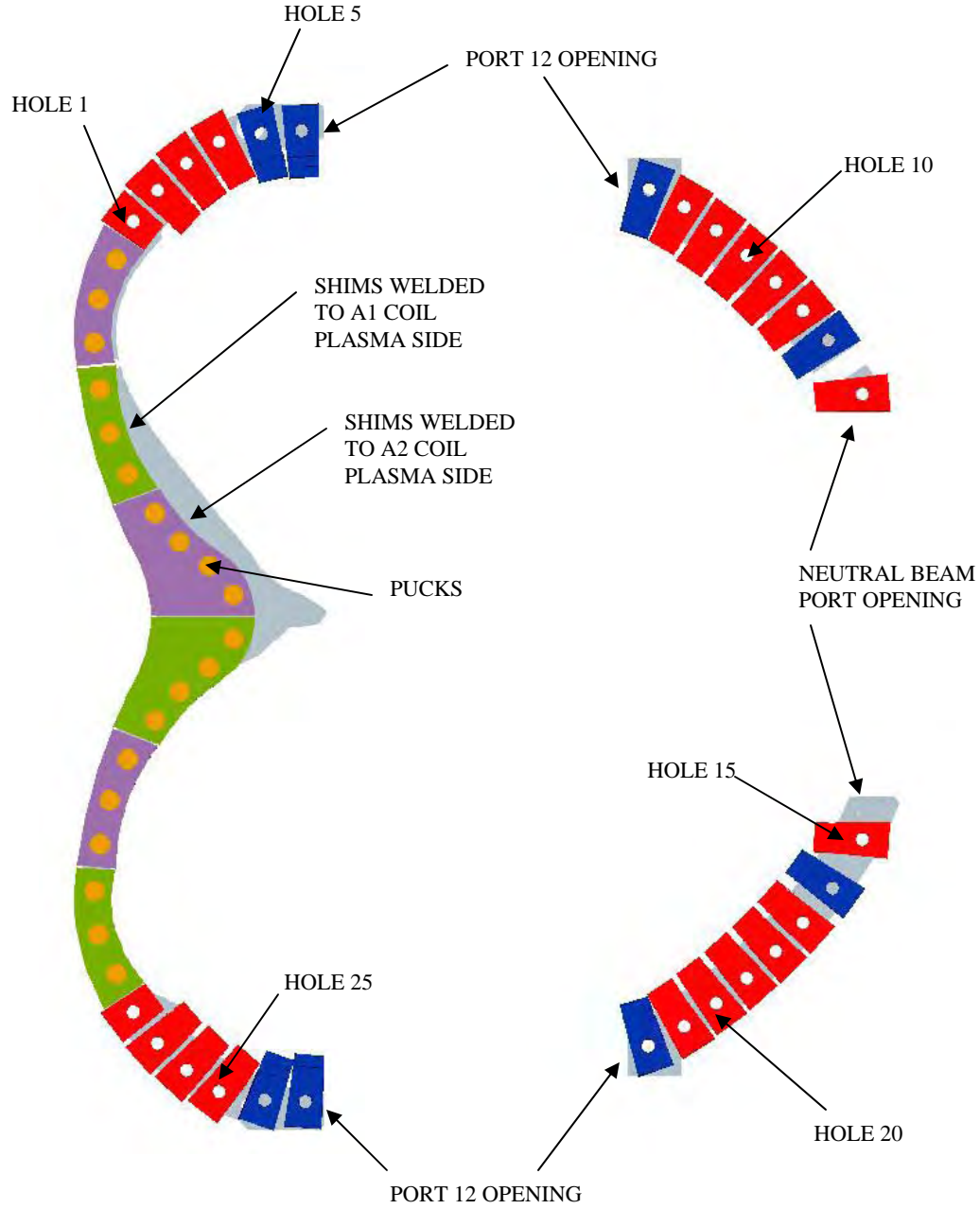
Type-A



Type-A

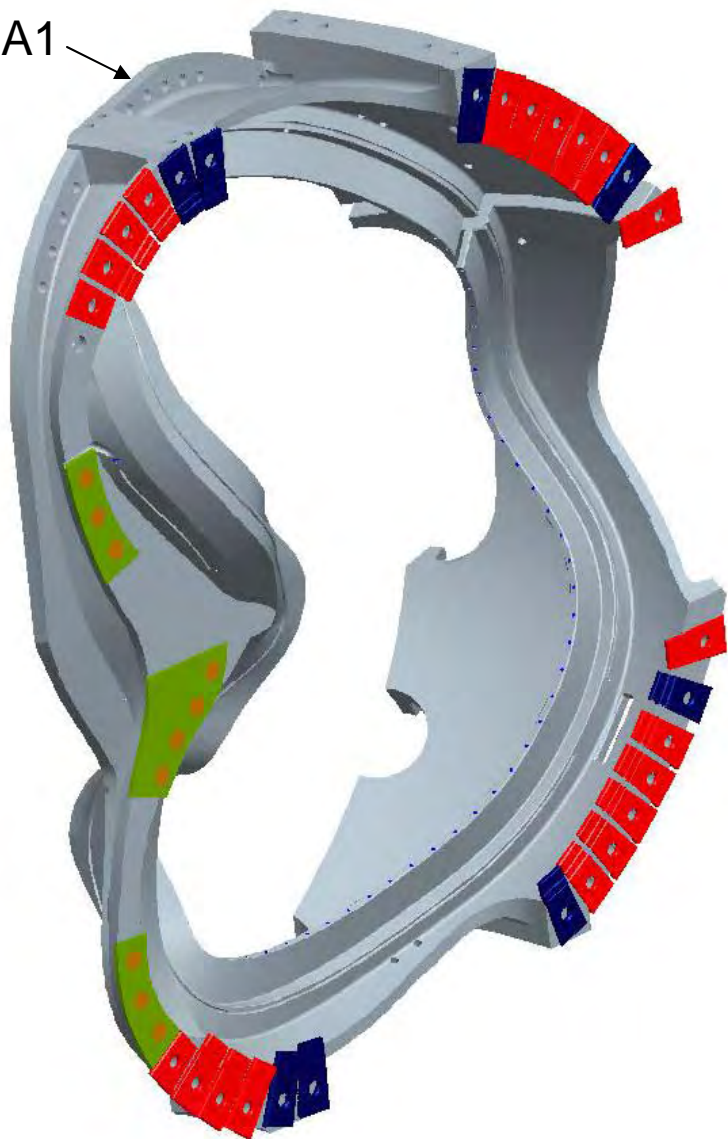
A-A FLANGE

AA Hole #	Shim Length Hole to Bottom	No Bolt Shim
1	2.75	
2	5.00	
3	5.00	
4	5.00	
5		5.00
6		5.00
7		5.00
8	5.00	
9	5.00	
10	5.00	
11	5.00	
12	5.00	
13		5.00
14	5.00	
15	5.00	
16		5.00
17	5.00	
18	5.00	
19	5.00	
20	5.00	
21	5.00	
22		5.00
23		5.00
24		5.00
25	5.00	
26	5.00	
27	5.00	
28	2.75	



AA Inboard Welded Shims

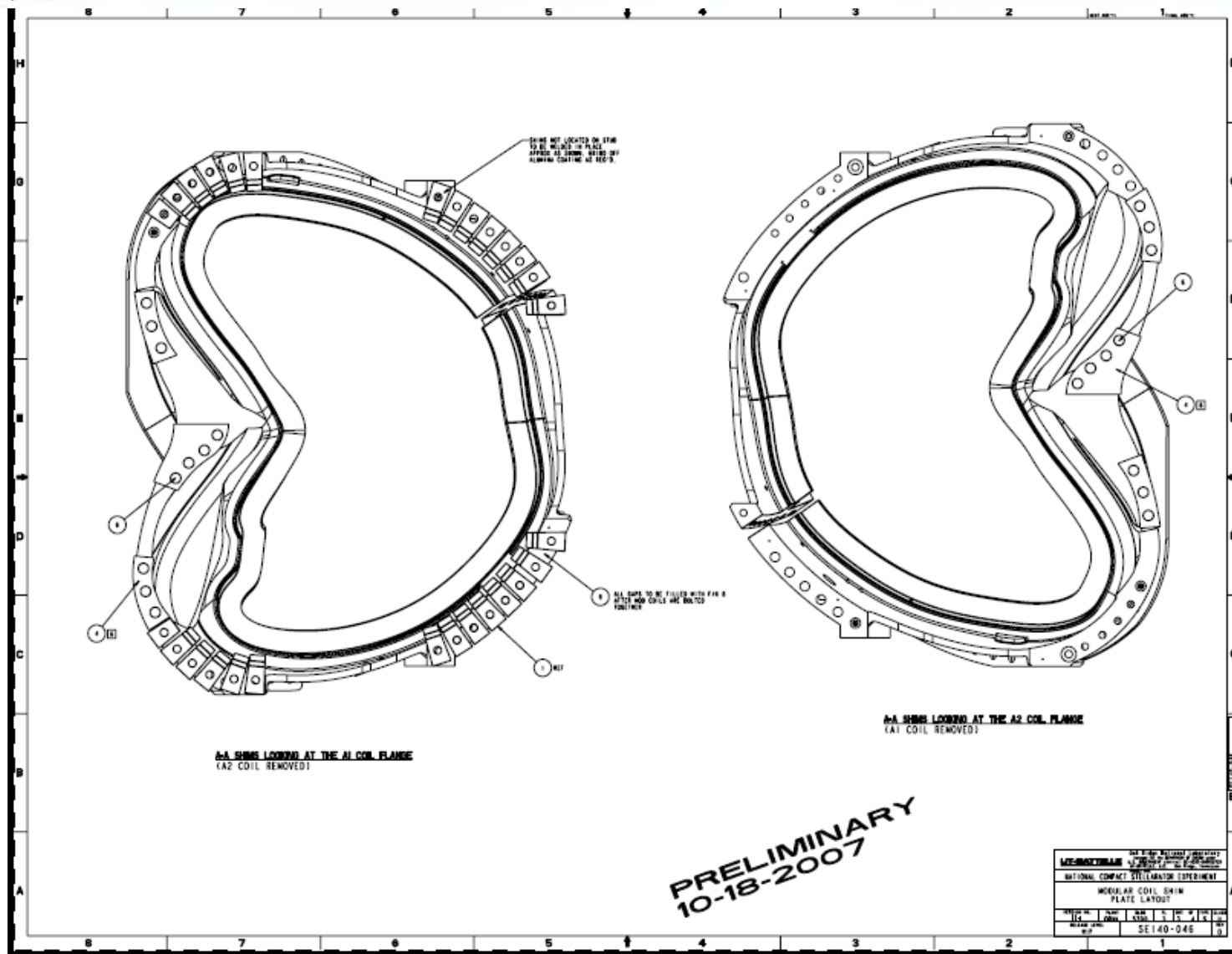
Type-A1



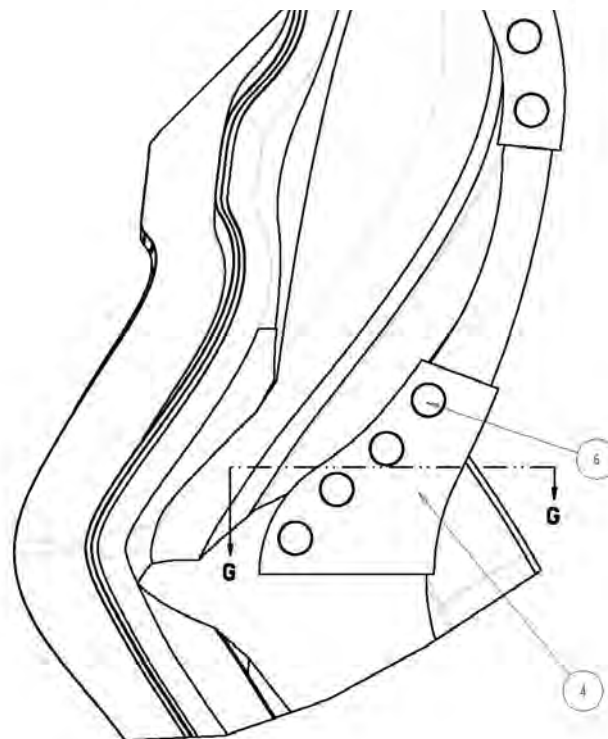
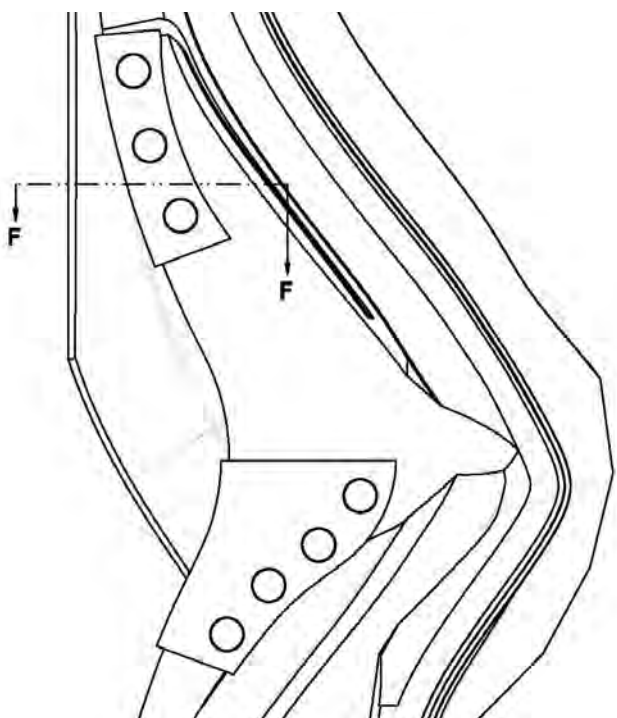
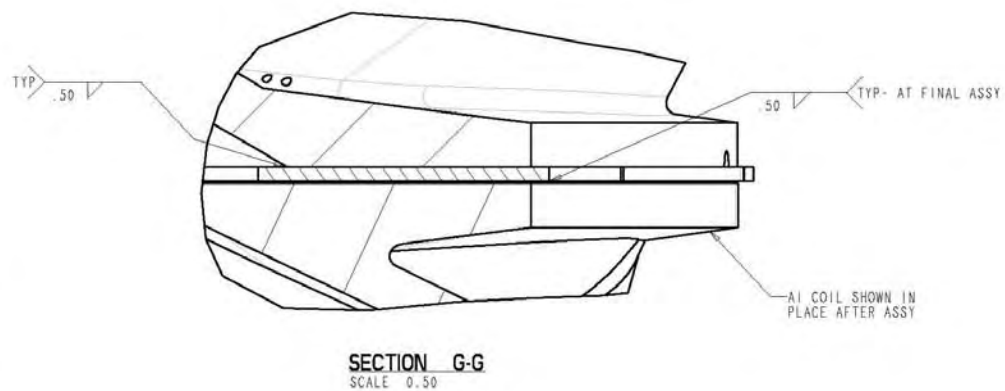
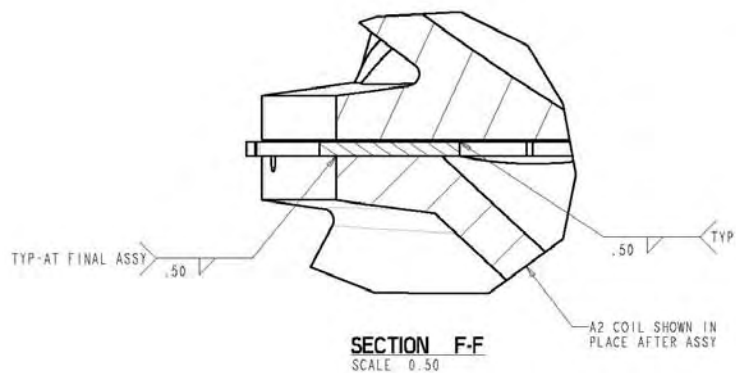
Type-A2



AA Inboard Welded Shims



AA Inboard Welded Shims

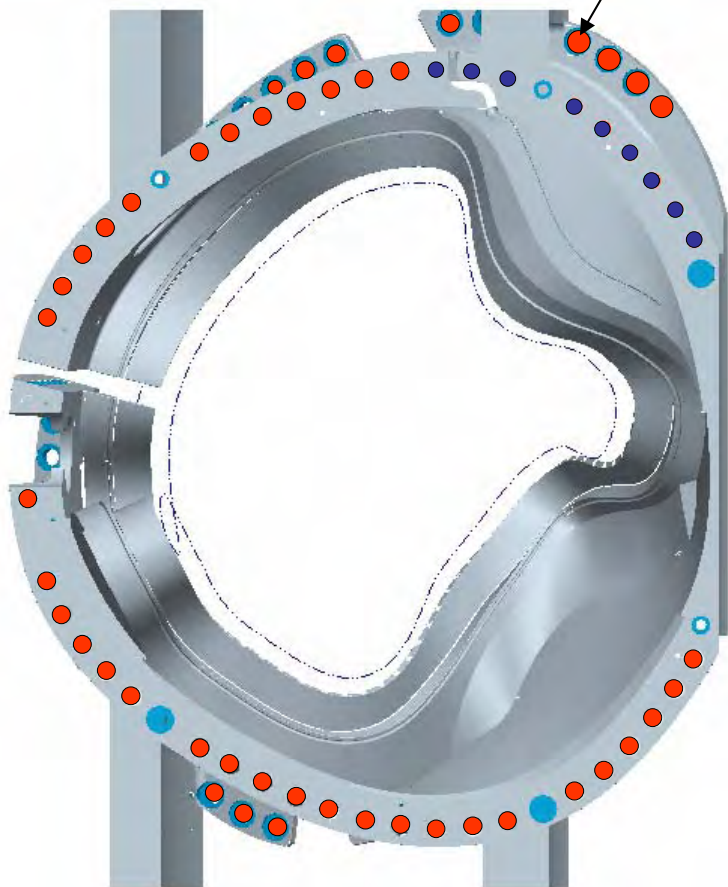


For Reference only Interface C-C

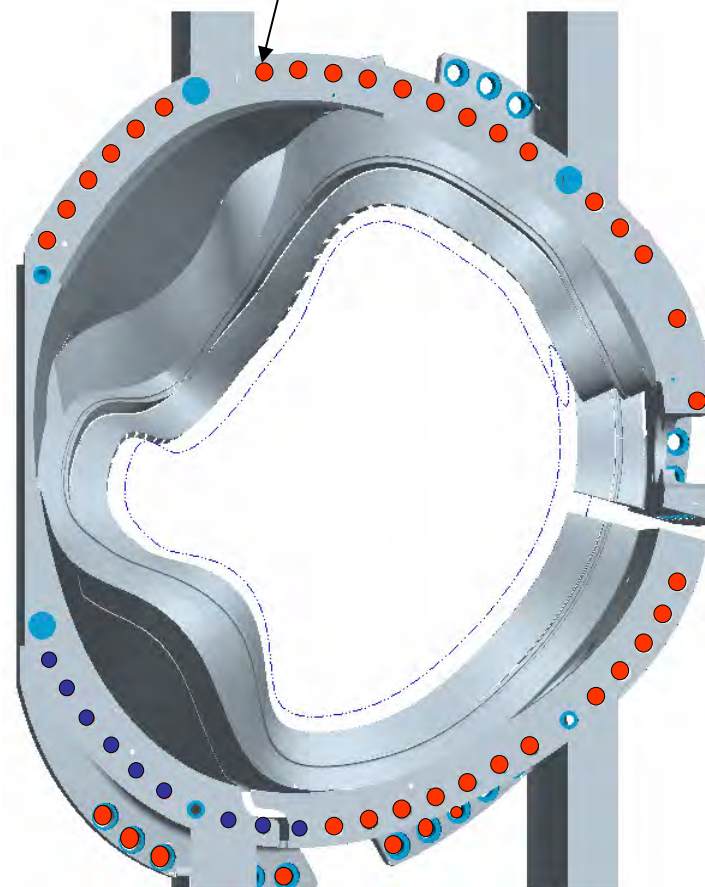
- 30 through holes
- 14 tapped holes

$\varnothing 1.375-6UNC \times 1.5 \text{ MIN}$
FOR FLANGE THK > 1.5
 $\varnothing .06 \text{ M A D}$

$\varnothing 3.00$ SPOTFACE BACKSIDE
MINIMUM TO CLEAN UP
 $\varnothing .06 \text{ M A D}$



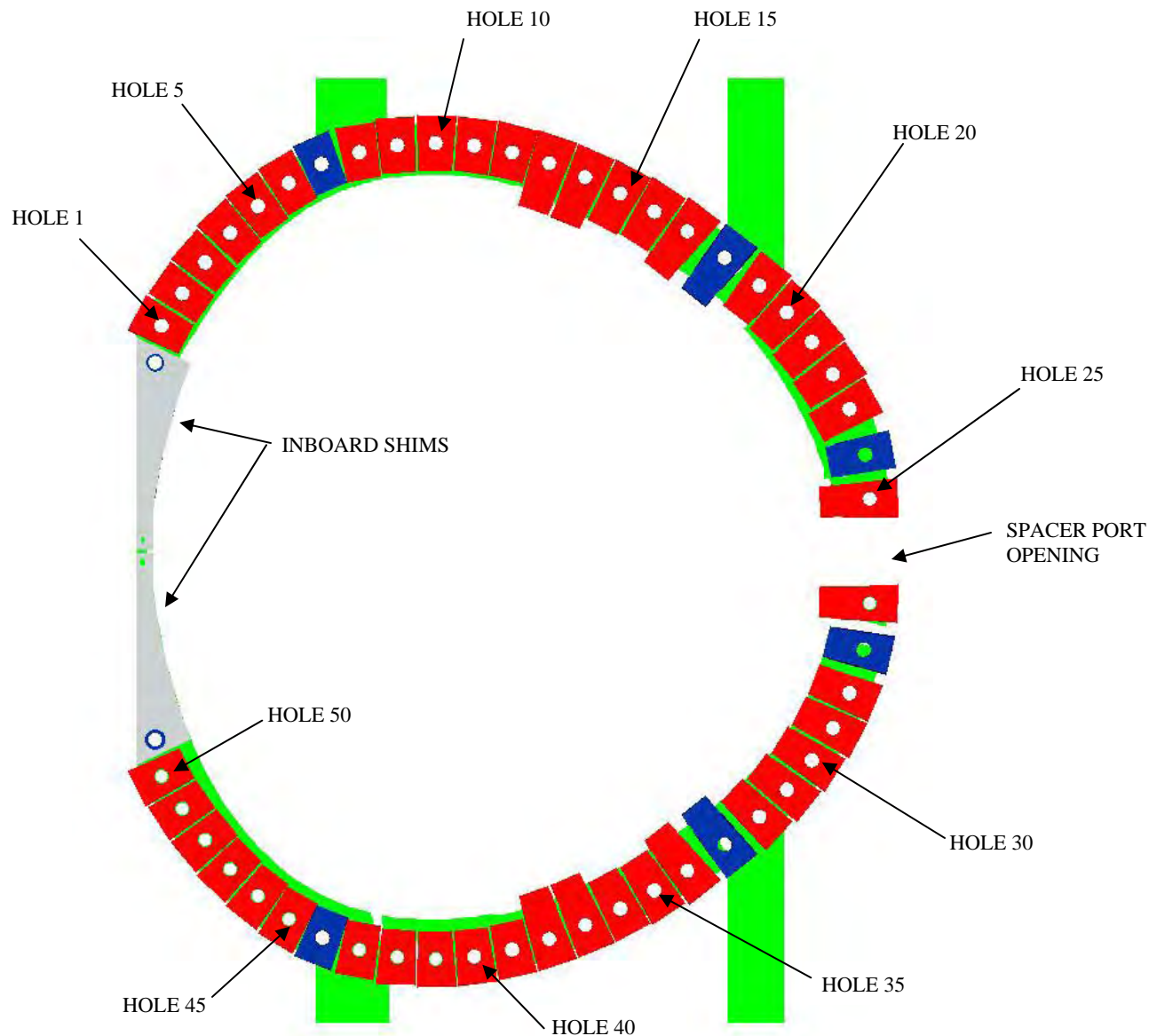
Type-C



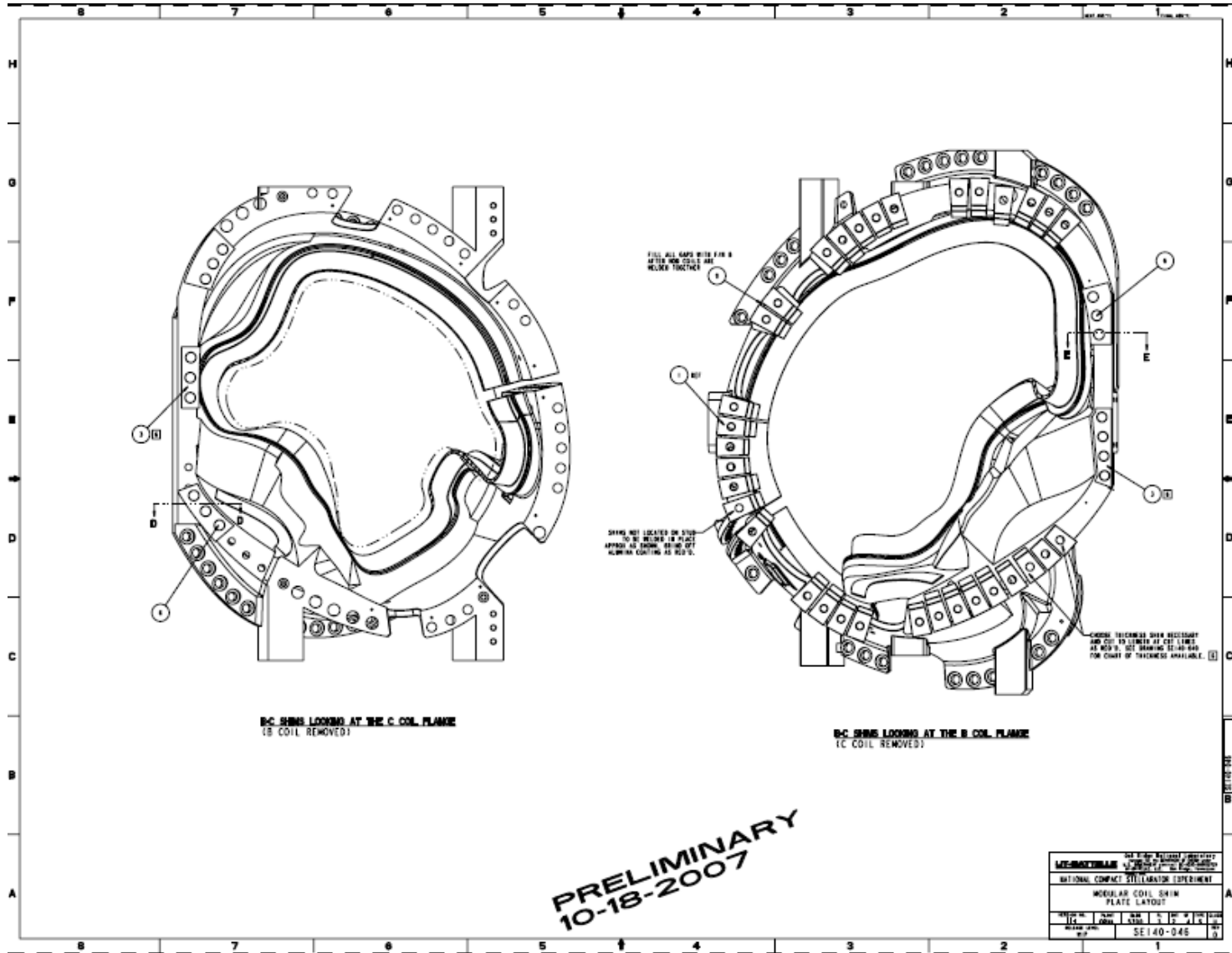
Type-C

C-C FLANGE

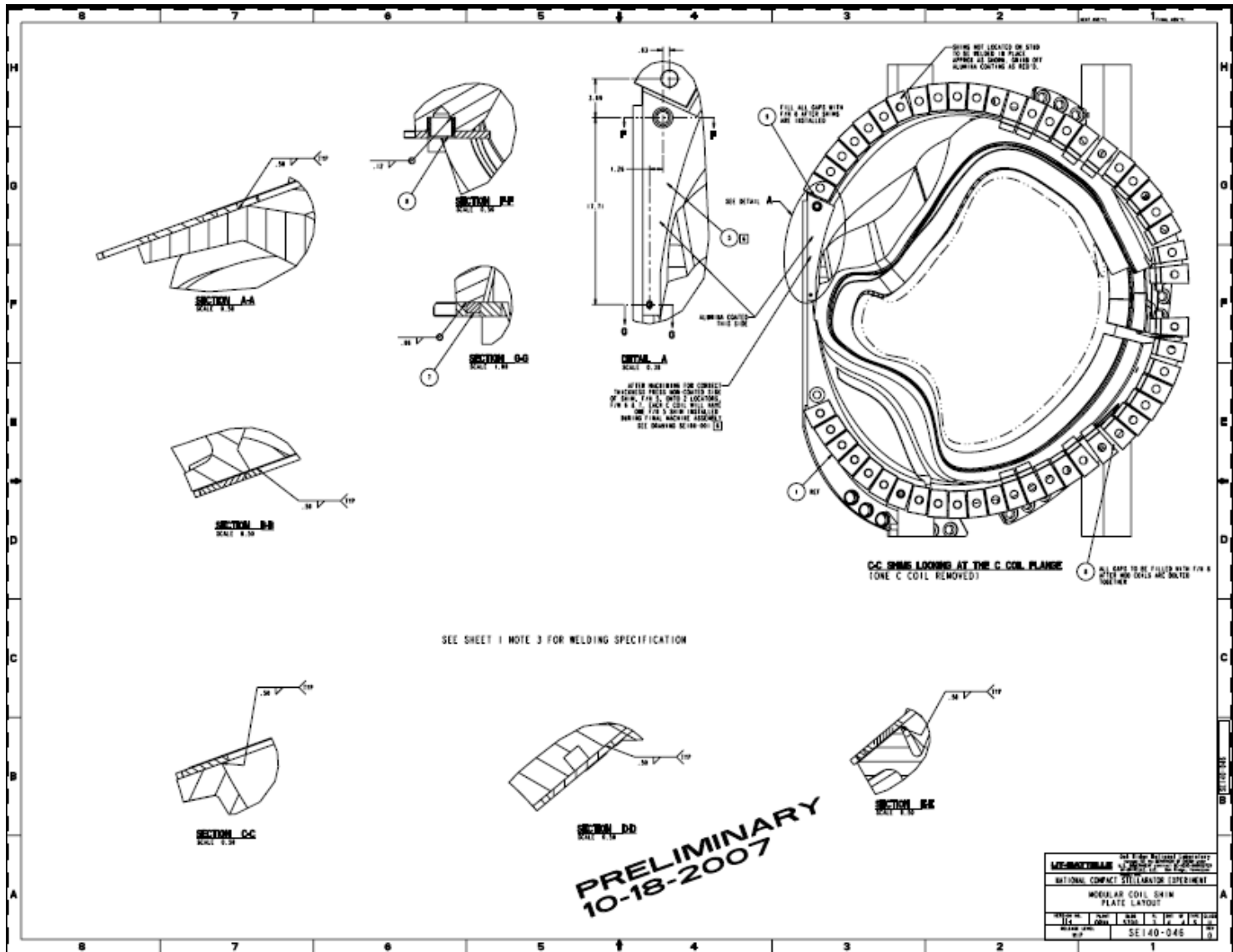
CC Hole #	Shim Length Hole to Bottom	No Bolt Shim
1	2.75	
2	2.75	
3	2.75	
4	2.75	
5	2.75	
6	2.75	
7		2.75
8	2.75	
9	2.75	
10	2.75	
11	2.75	
12	2.75	
13	5.00	
14	5.00	
15	3.75	
16	3.75	
17	5.00	
18		5.00
19	3.75	
20	3.75	
21	3.75	
22	3.75	
23	3.75	
24		3.75
25	5.00	
26	5.00	
27		3.75
28	3.75	
29	3.75	
30	3.75	
31	3.75	
32	3.75	
33		5.00
34	5.00	
35	3.75	
36	3.75	
37	5.00	
38	5.00	
39	2.75	
40	2.75	
41	2.75	
42	2.75	
43	2.75	
44		2.75
45	2.75	
46	2.75	
47	2.75	
48	2.75	
49	2.75	
50	2.75	



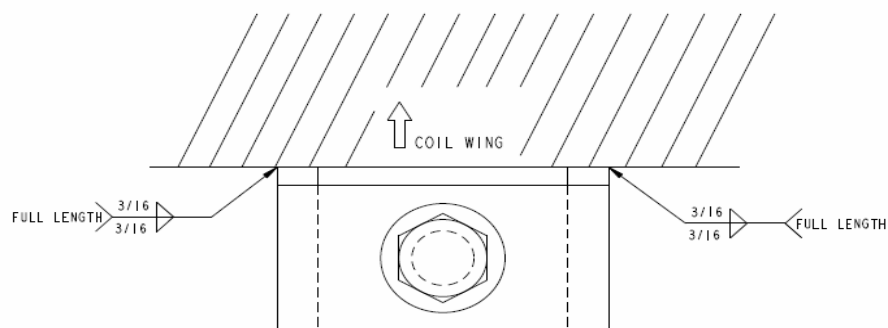
For Reference only Interface C-C



For Reference only Interface C-C

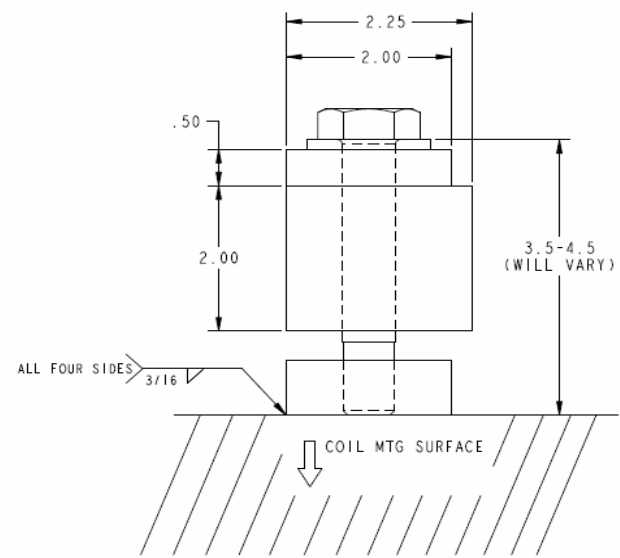
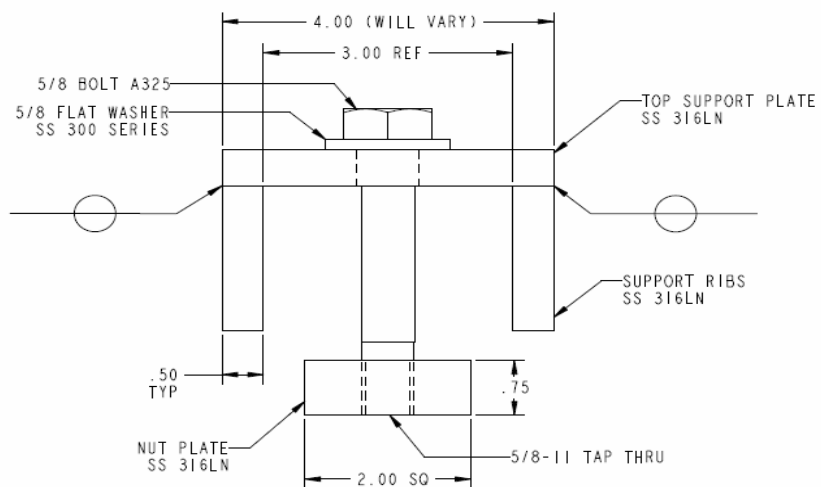
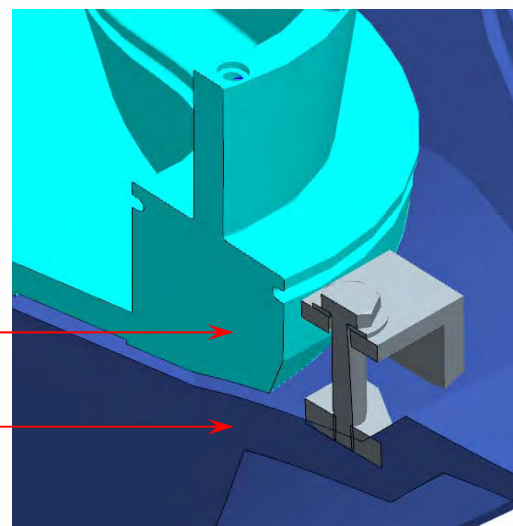


Wing support design



COIL WING

COIL MTG SURFACE



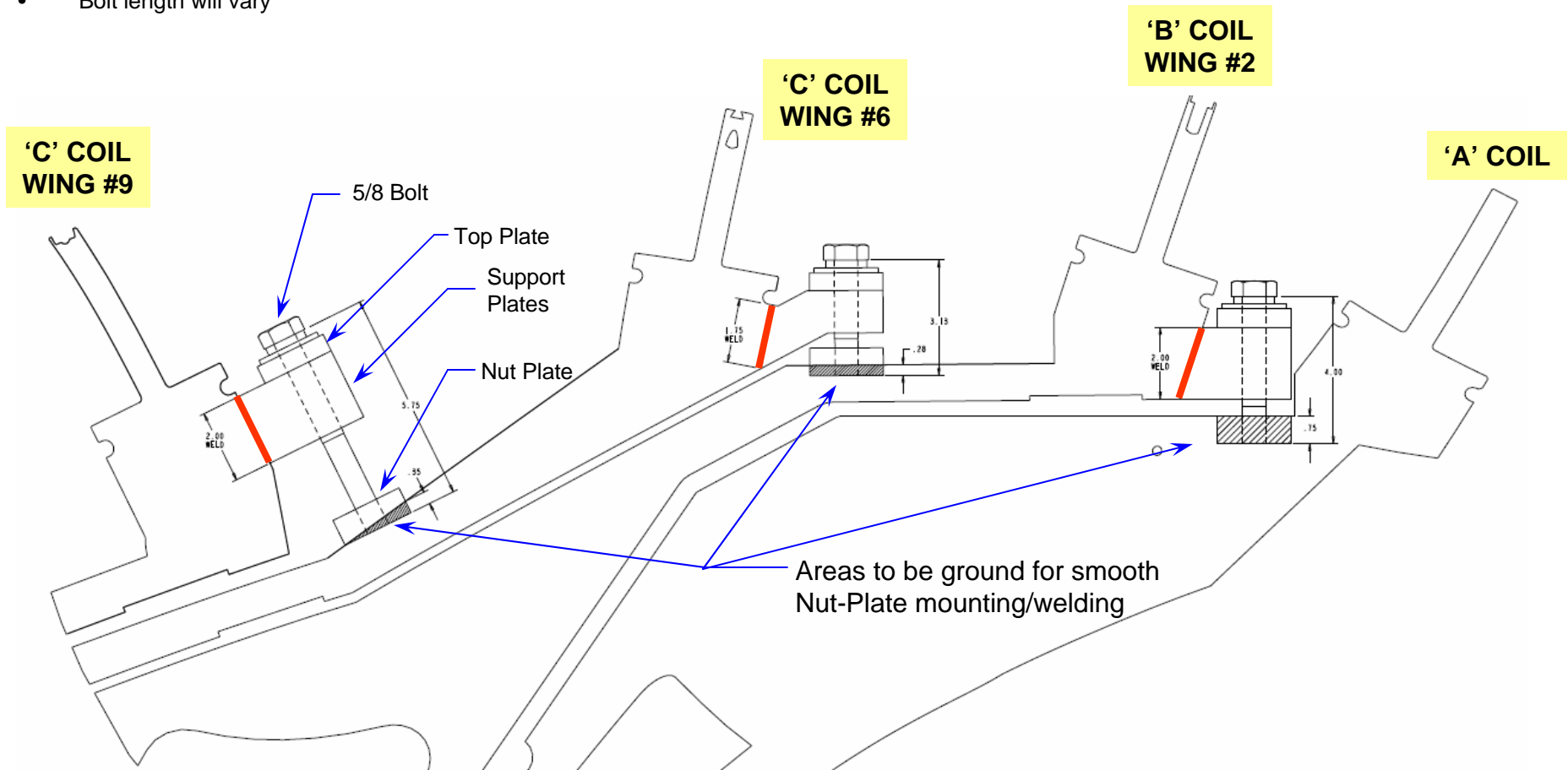
CONCEPT USING WELDED CHAIR WITH NUT-PLATE

WINGS # 2, 6 and 9



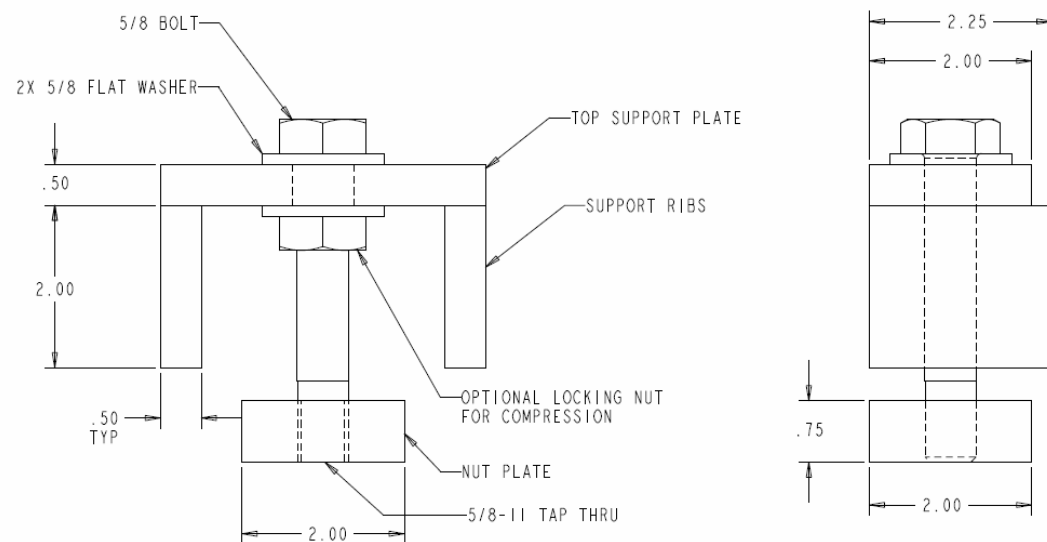
NOTES:

- Setup will vary for each wing and each chair weld on each wing
- Support plates can be fabricated to be ground to shape as necessary to match during installation
- This example shows only 3 specific areas – for depiction only
- Bolt length will vary



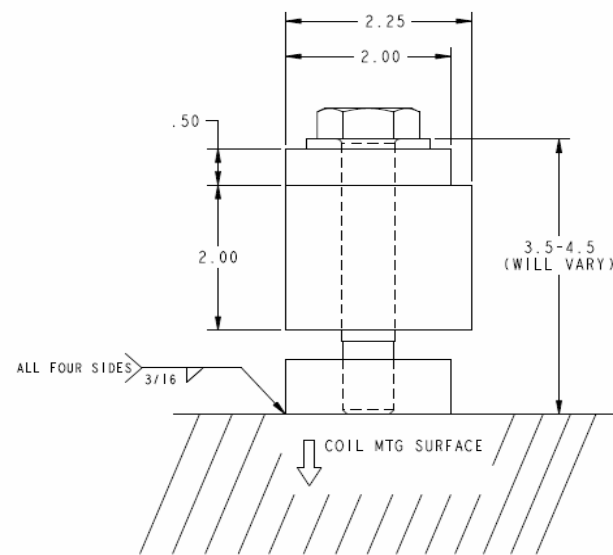
Bolt Tensile Stress Analysis

Total Bolt Load	10,000	
Nominal Bolt Diameter	0.625	in
Number of Threads per inch	11	threads / in
Bolt Tensile Stress Area	0.2260022	in ²
Bolt Tensile Stress	44,247	lb / in ²
Allowable Tensile Stress	45,000	lb / in ²



Nut Plate Weld Analysis

Total Load	10,000	lbs
Weld Length	8	in
Weld Size	0.1875	in
Effective Weld Throat	0.1326	in
Nut Plate Shear Stress	9,428	lb / in ²
Allowable Shear Stress	14,800	lb / in ²



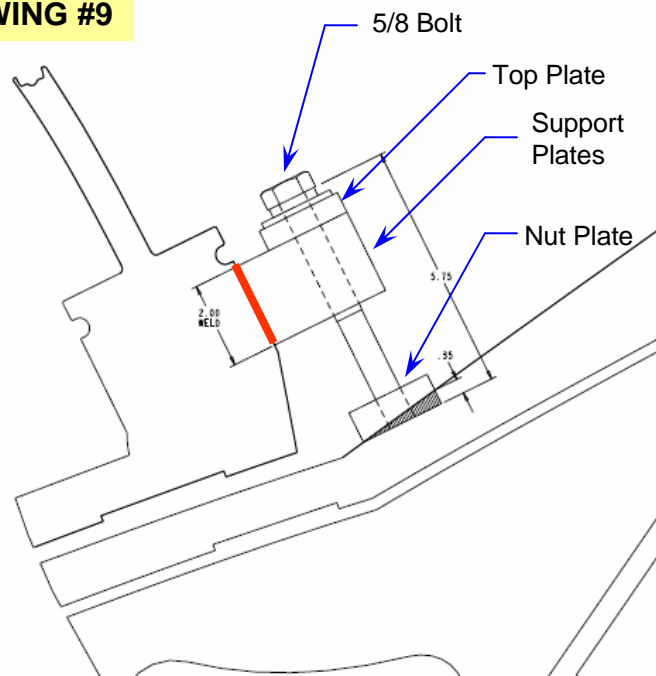
Wing support design



Chair Weld Analysis - 2" Straight - Shear Stress

Total Load	10,000	lbs
Weld Length	8	in
Weld Size	0.1875	in
Effective Weld Throat	0.1326	in
Chair Weld Shear Stress	9,428	lb / in ²

**'C' COIL
WING #9**



Bending Stress

Weld Size	0.1875	in
Cantilever	1.2500	in
Total Load	5,000	lbs
Moment	6,250	lb * in
Weld Length	2.5	in
Distance between welds	0.5	in

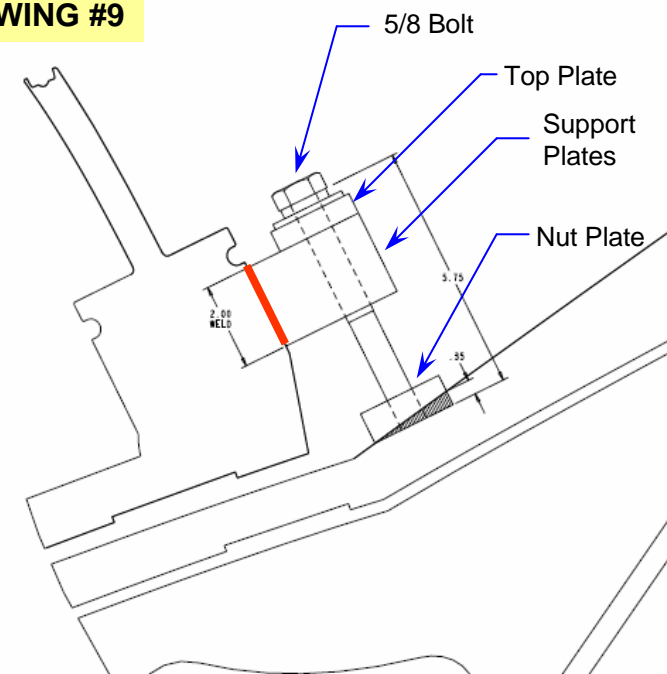
Unit Second Moment of Area 2.6041667

Second Moment of Area 0.3452148

Bending Stress **22,631**

lb / in²

**'C' COIL
WING #9**



- Modular coil asm design basis is defined by 5 analysis reports:

HM Fan, Nonlinear Analysis of Coil and Shell Structure, NCSX-CALC-14-001, APPROVED
HM Fan, Analysis of Integrated Structure, NCSX-CALC-14-003, APPROVED
K Freudenberg, Modular Coil Thermal Analysis, NCSX-CALC-14-002, DRAFT
K Freudenberg, Nonlinear Modular Coil Analysis, NCSX-CALC-14-004, DRAFT
D Williamson, Modular Coil Failure Modes Analysis, NCSX-FMEA-14-002, DRAFT

- Additional analysis reports are planned before Design Closeout:



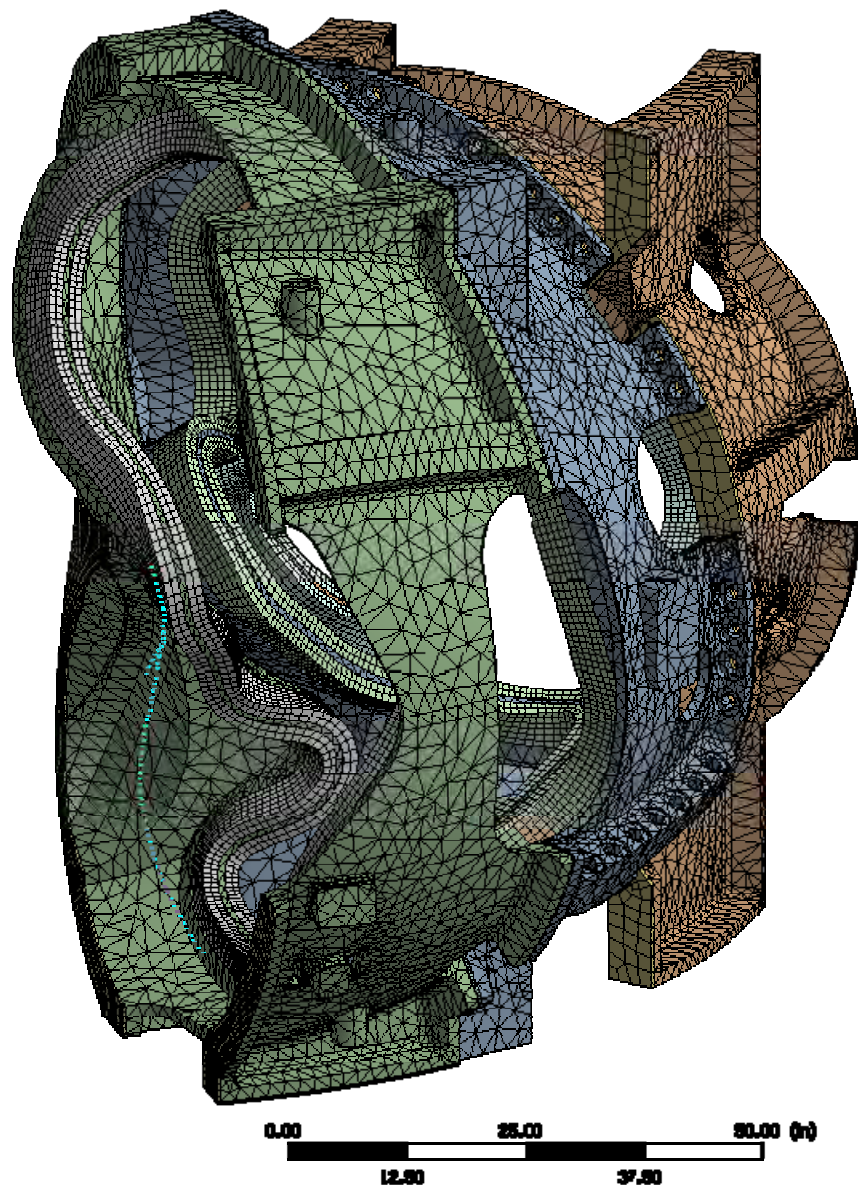
K Freudenberg, Outboard Bolted Joint Analysis, NCSX-CALC-14-006, DRAFT
K Freudenberg, Modular Coil Inboard Shims Analysis, IN PROGRESS
D Williamson, Modular Coil Leads Structural Analysis, PLANNED

Quick weld allowable calculation

- $S_m = 2/3 S_y$ at temp or $1/3 S_{ult}$ for all materials
- $S_y = 93.2$ ksi for Stelalloy but weld since S_{ult} is 157.5 -> $S_m = 52.5$ based on weld wire.
- Knockdown factor of .6 applied for visual inspected welds. → **31.5 ksi**. Which is our max stress intensity we can incur statically.

- Shim and pucks will be constructed of 316 LN stainless steel.
 - $S_y = 131$ ksi at 77K *
 - $S_{ult} = 205$ ksi at 77K *
 - ► **$S_m = 68$ ksi (1/3 S_{ult})**

New Global Model for welds



- Weld Elements are placed in the model on the each flange of interest in turn. (AA, AB, BC)
- Bolt holes and bolts taken out of all flanges except the one being studied (bolts may be added latter).
- This method locks up the outboard side of the global model with bonded contact but has the inboard compression pucks run with frictionless sliding (Keyopt 12 = 0). The weld takes all of the shear from the interface.
- Material Properties of weld match that of shim and castings.
- The analysis has only examined the 2T high beta case. (supposedly the worst EM load case.)
- A detailed sub-model of the in-board leg, including proper segmentation and filet weld shape may then be constructed on flanges where stress is deemed an issue.

Quick check FEA models

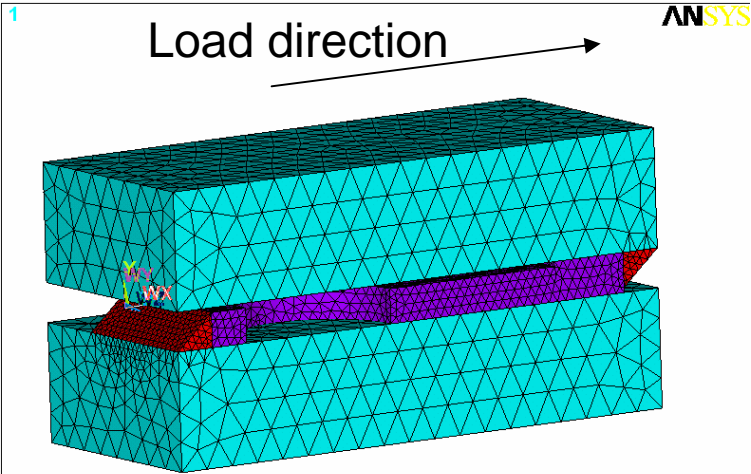


Geometry

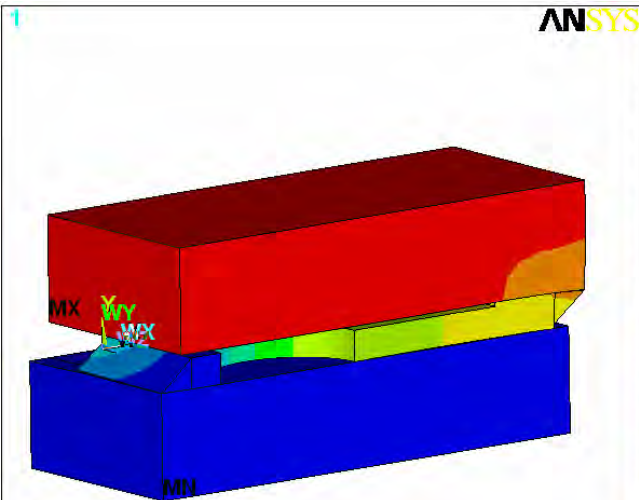
- gap between shims 0.25 in
- shim width (W) 4 in
- shear shim thickness (t) 0.438 in
- compressive puck thickness (pt) 0.5 in
- compressive puck diameter (pd) 2 in
- Hole diameter(d) 2.1 in
- Shim Length (L) 6 in
- weld size 0.309 in

0.25 in
4 in
0.438 in
0.5 in
2 in
2.1 in
6 in
0.309 in

Applied average shear force = 4.2 Kips/in



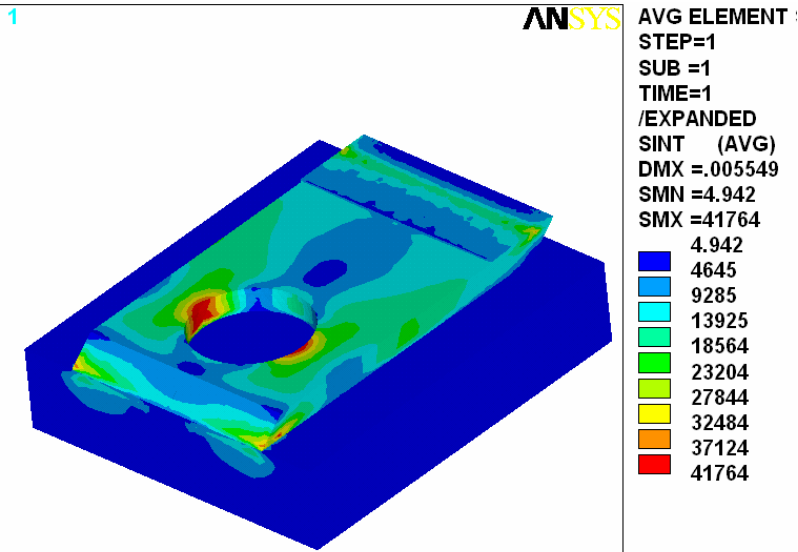
simple weld analysis



NODAL SOLUTION
 STEP=1
 SUB =1
 TIME=1
 USUM (AVG)
 RSYS=0
 PowerGraphics
 EFACET=1
 AVRES=Mat
 DMX =.005549
 SMX =.005549
 0
 .617E-03
 .001233
 .00185
 .002466
 .003083
 .003699
 .004316
 .004932
 .005549

simple weld analysis

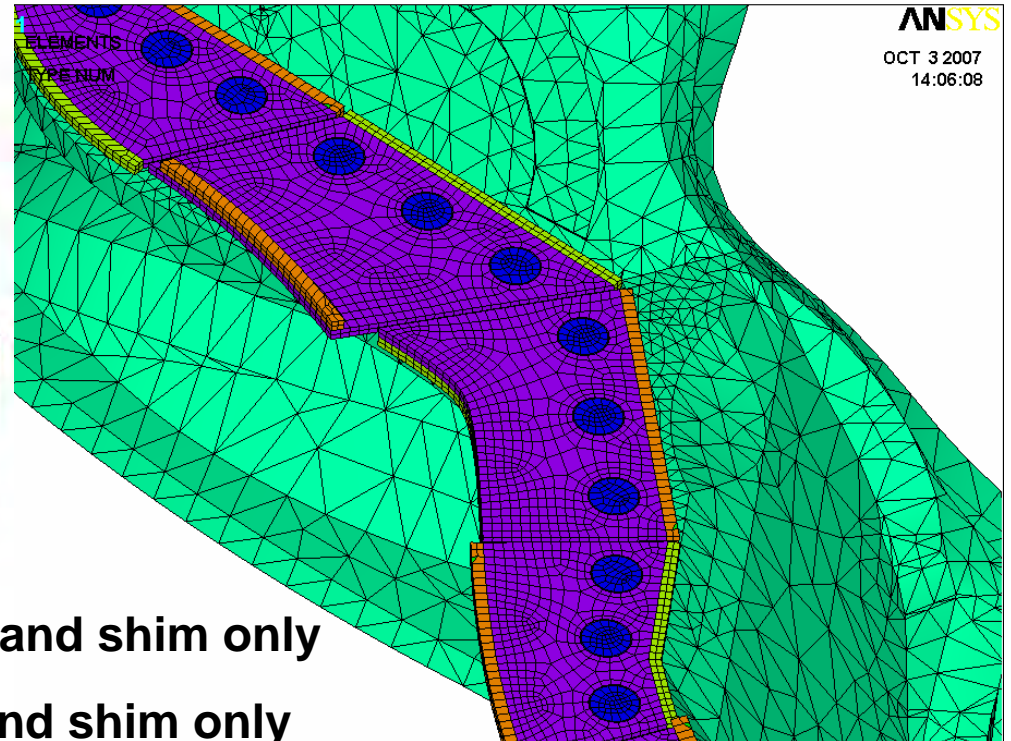
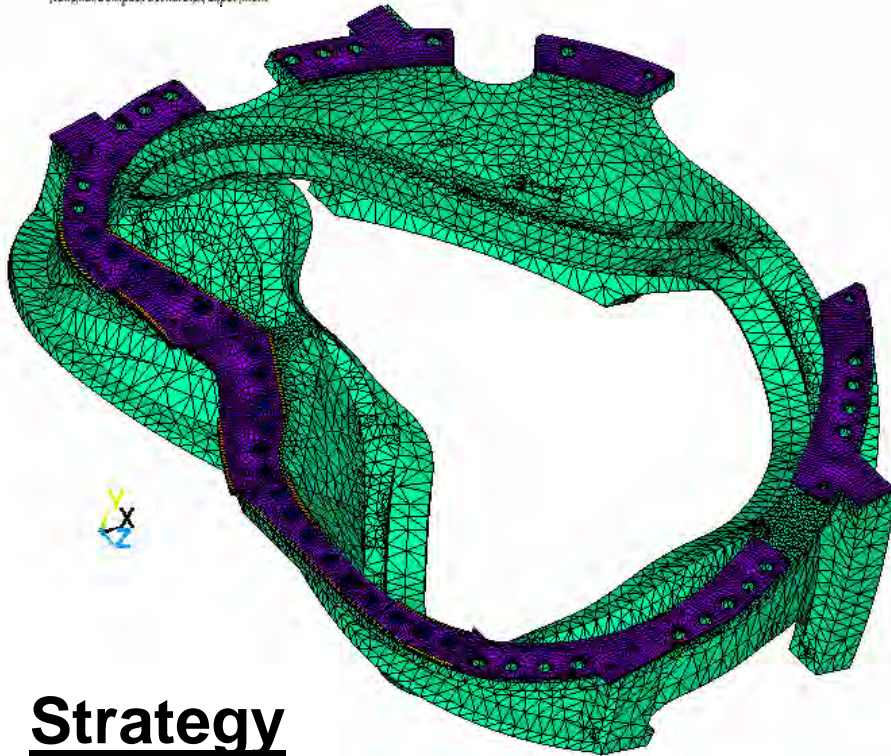
inches



stress Intensity

psi

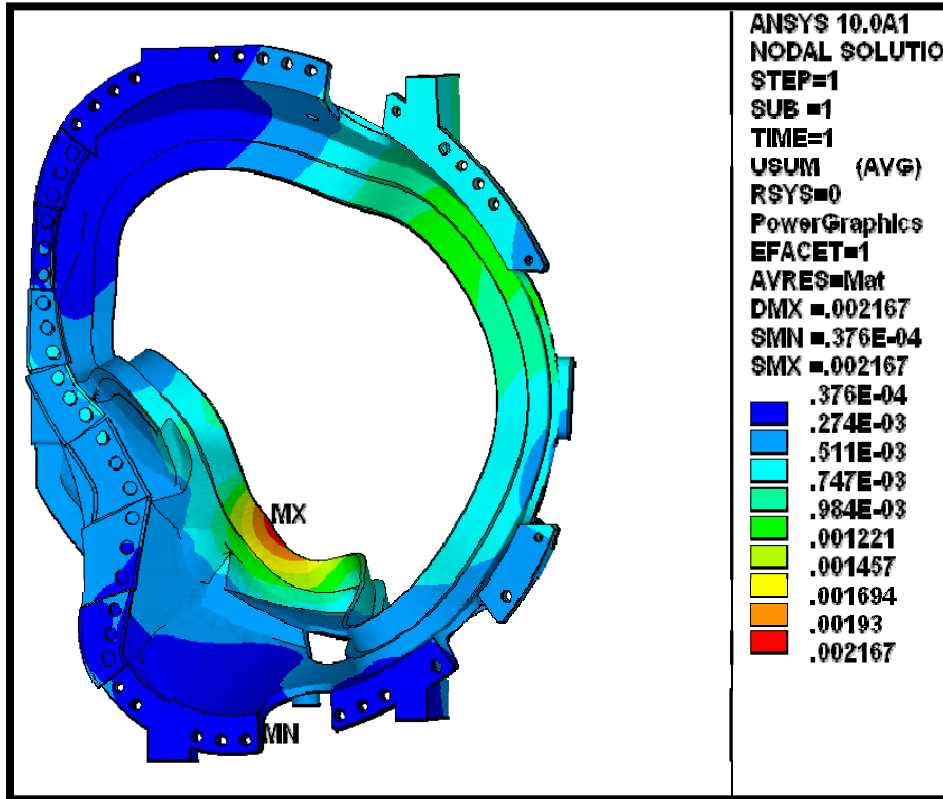
Global Model mesh of weld interface (AB)



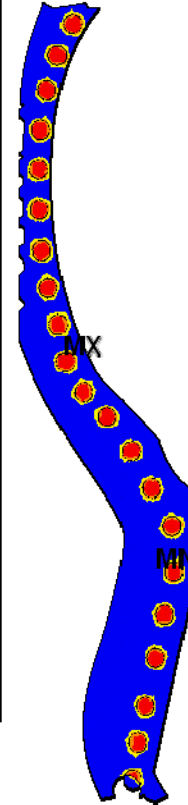
Strategy

- Orange welds connect to B casting and shim only
- Green welds connect to A casting and shim only
- Blue pucks carry compressive loads via frictionless interface on puck to B flange (blue pucks are glued to the A flange)
- Pucks are not connected to the shim.
- Shim in weld area will only connect to the A flange using frictionless contact and not be connected to the B flange at all (no compression carried by shim)

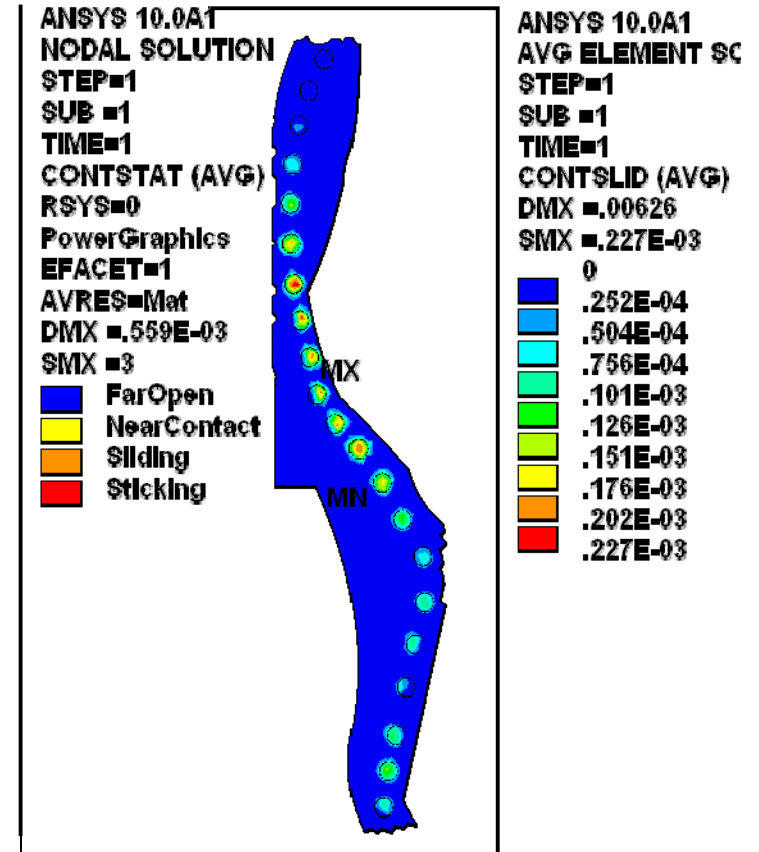
AB Deformation and puck check



Global Deformation. (meters)



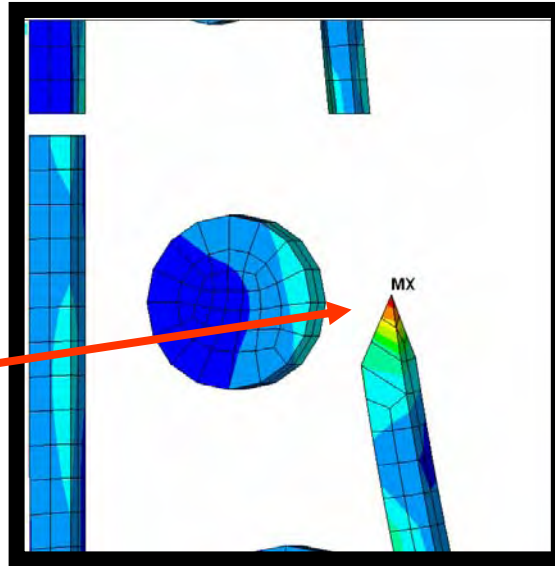
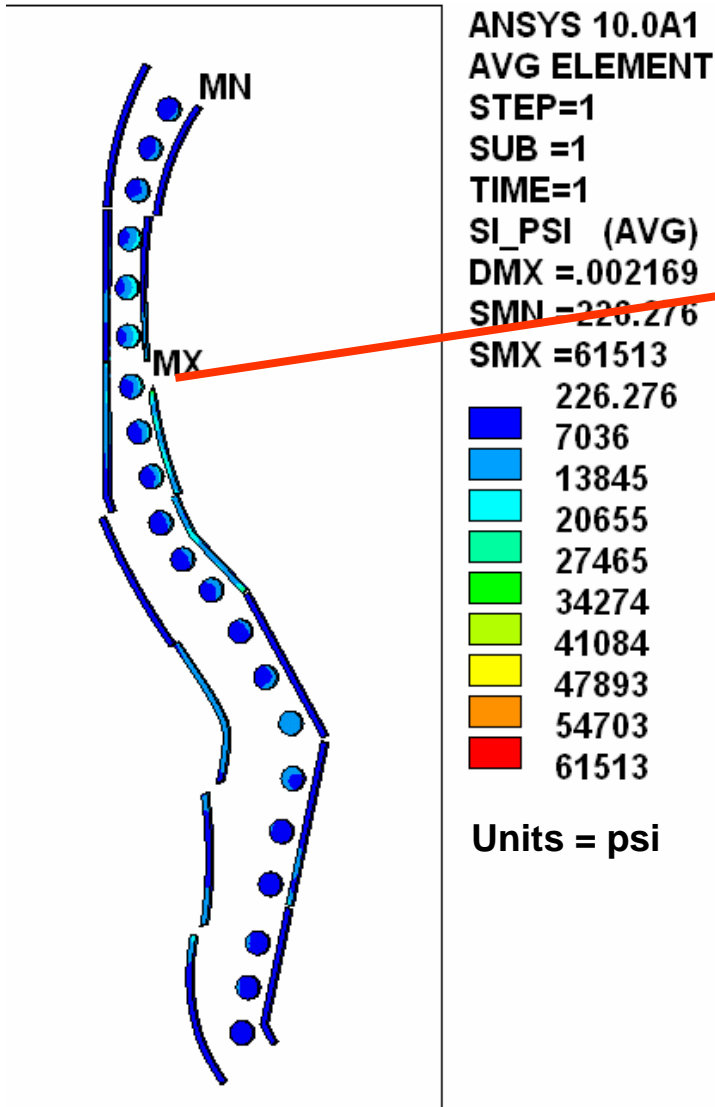
Pucks are stuck on B flange



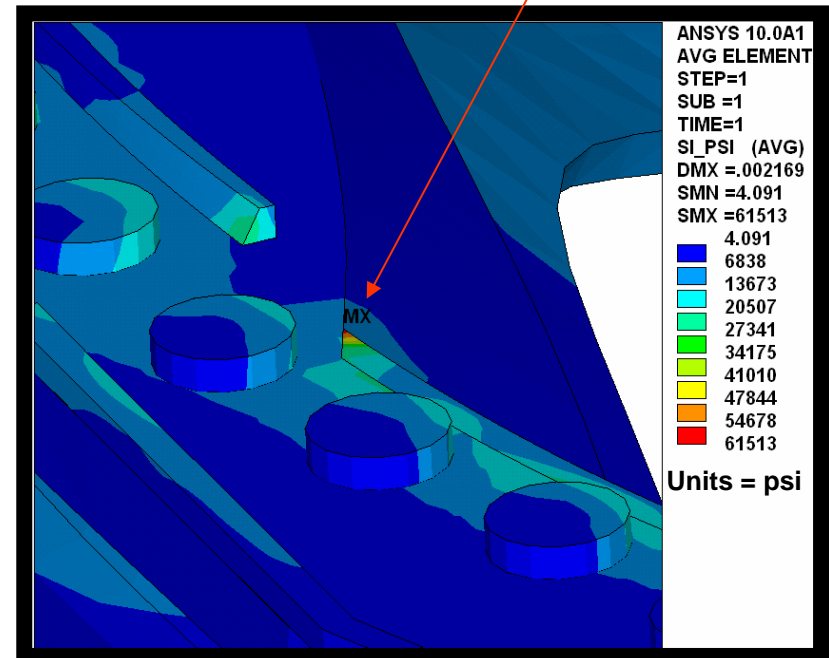
Pucks slide on A flange (0.227 mm)

The pucks are not connected to shim and the shim is not connected to either flange in the welded region.

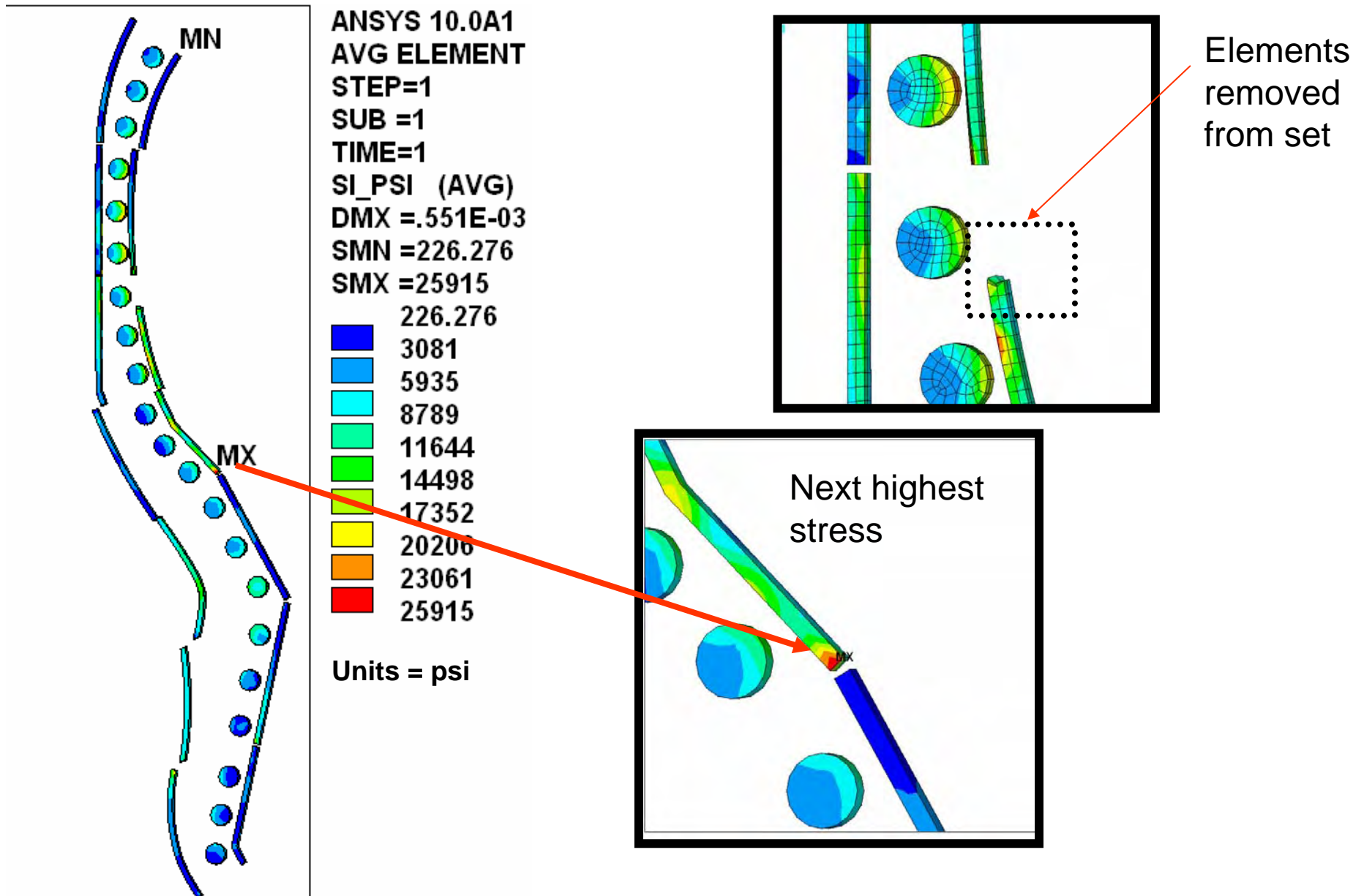
Stress Intensity on welds and pucks



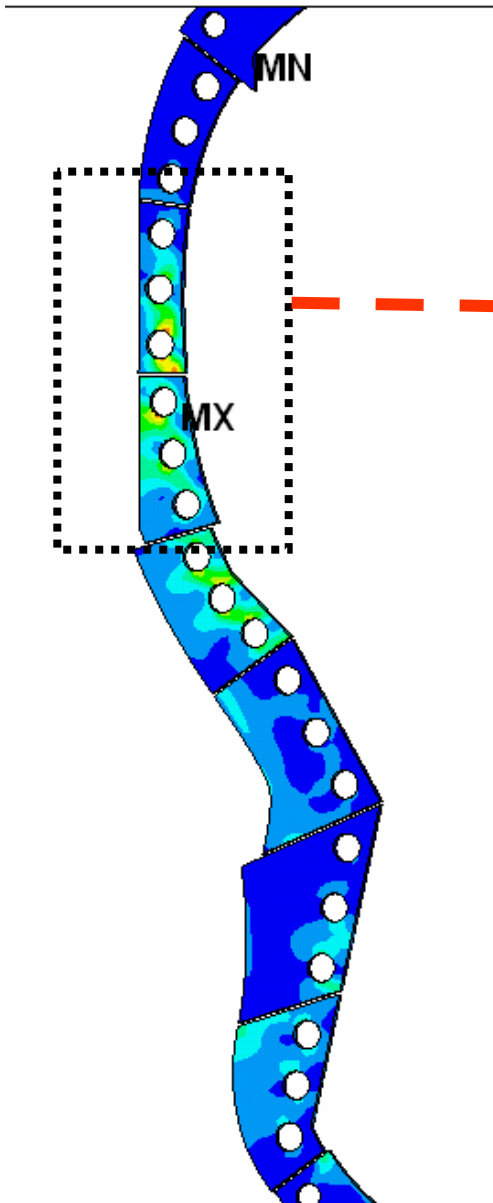
As with previous weld incarnations, a very peaky weld stress occurs on the corner edge of a weld (stress discontinuity, multiple corners)



Weld and Puck Stress- (After removal of stress discontinuity elements)



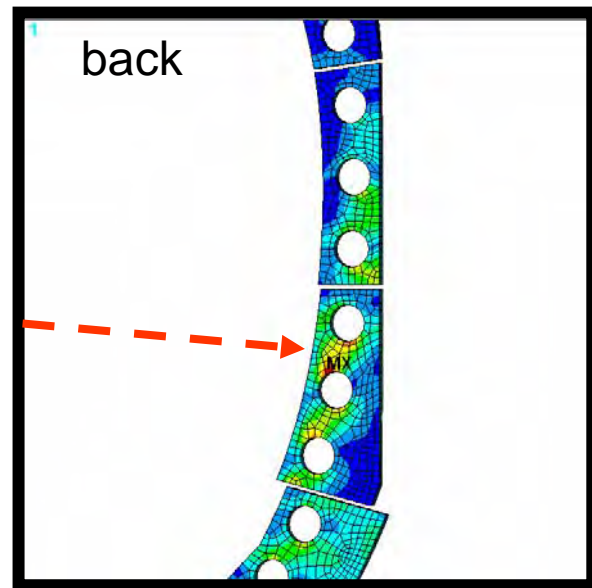
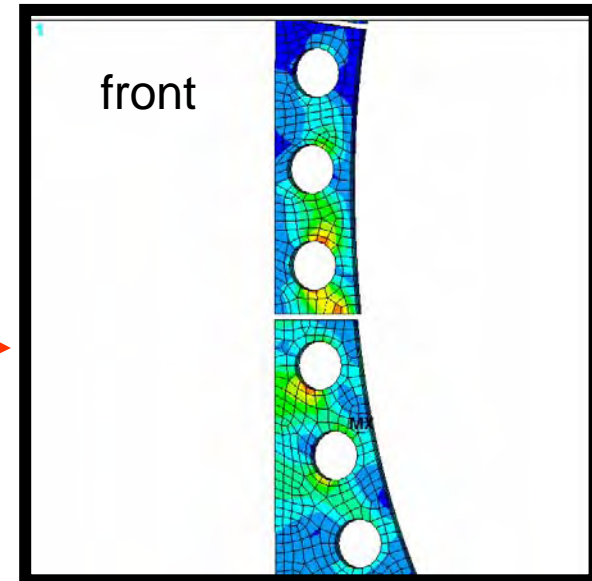
Stress in Shims



ANSYS 10.0A1
AVG ELEMENT
STEP=1
SUB =1
TIME=1
SI_PSI (AVG)
DMX =.675E-03
SMN =10.504
SMX =42010

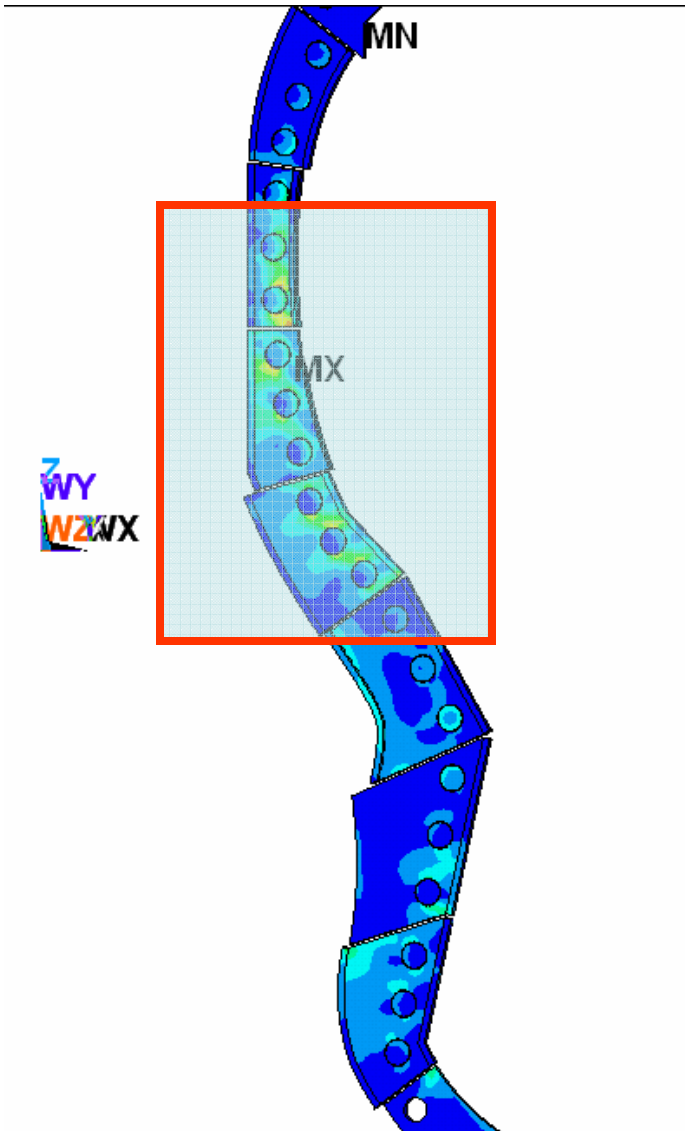


Units = psi



Max stress
occurs on
back of plate

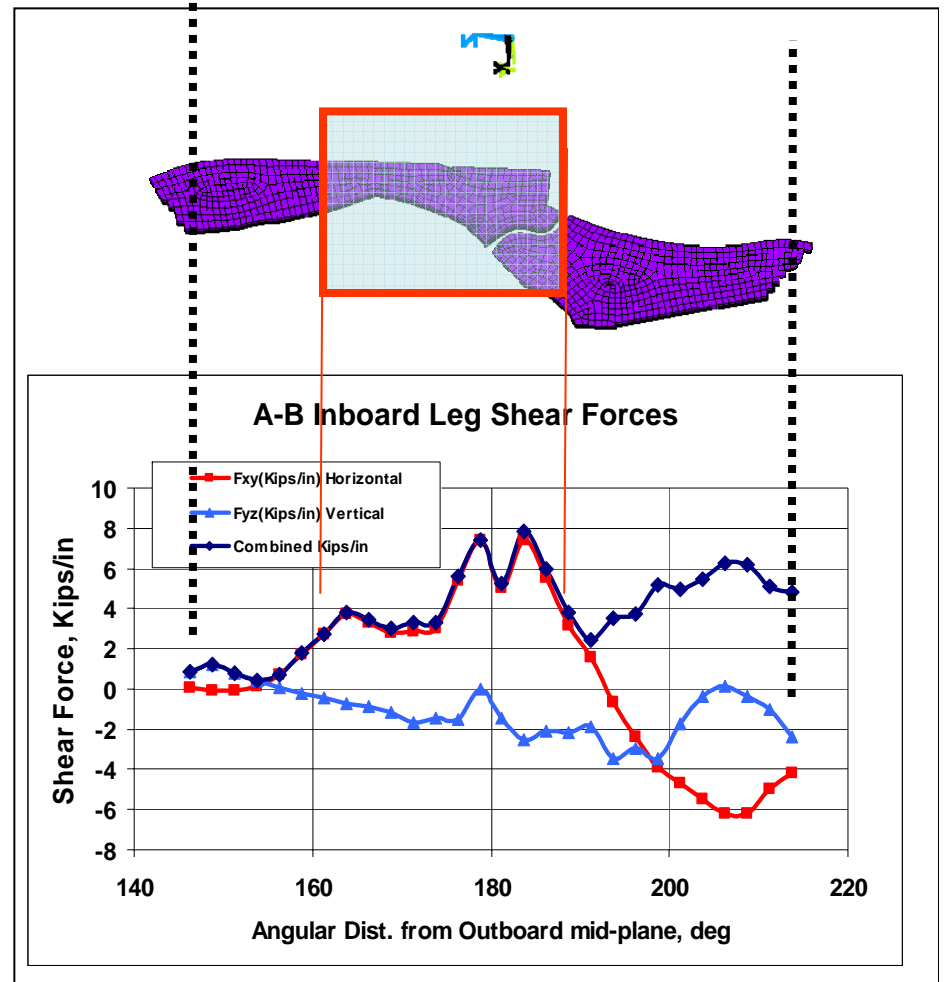
How does compare to previous results?



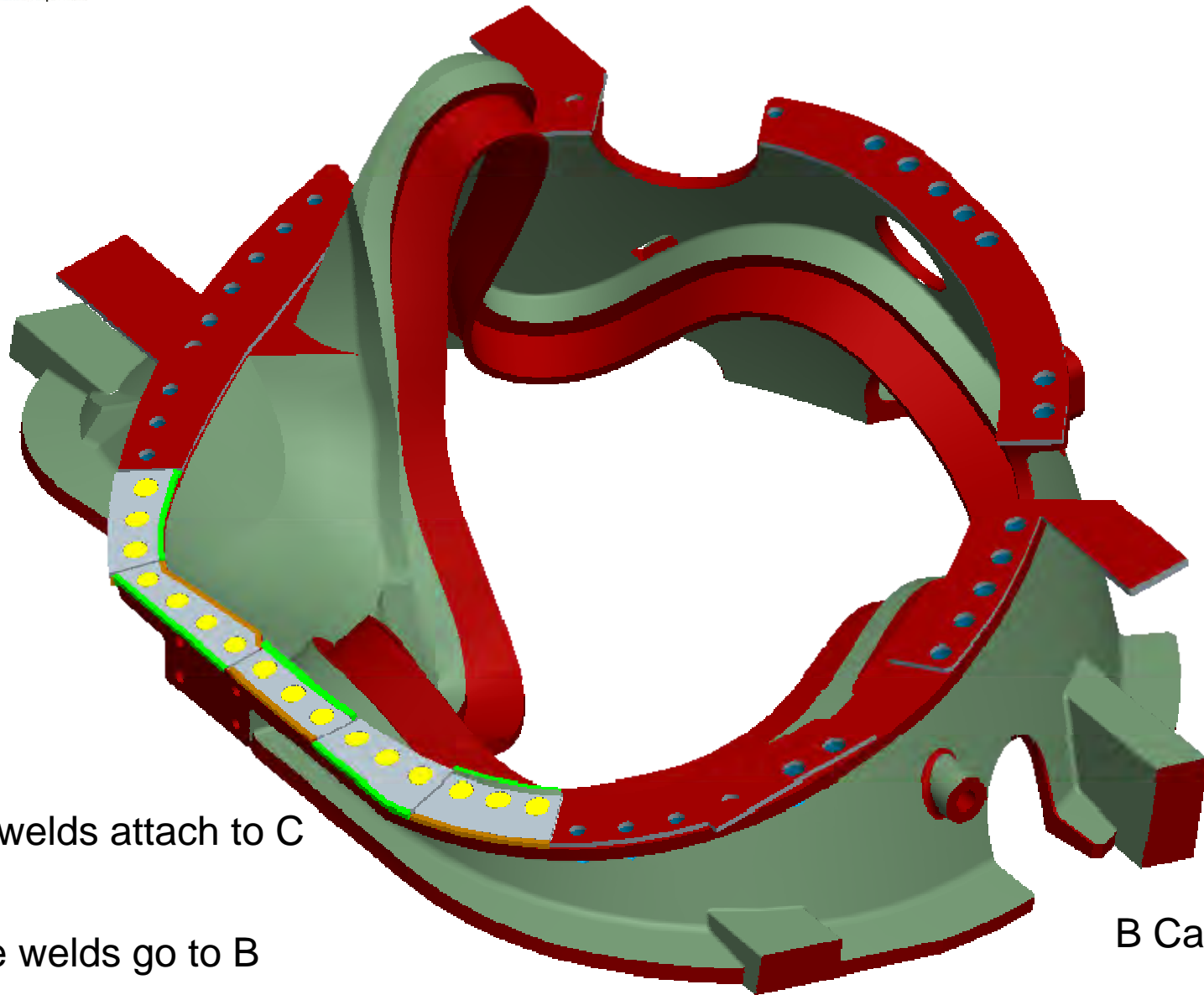
ANSYS 10.0A1
AVG ELEMENT
STEP=1
SUB =1
TIME=1
SI_PSI (AVG)
DMX =.675E-03
SMN =10.504
SMX =42010

10.504
4677
9344
14010
18677
23344
28010
32677
37344
42010

Peak Area of horizontal shear shown boxed in red



BC Connection

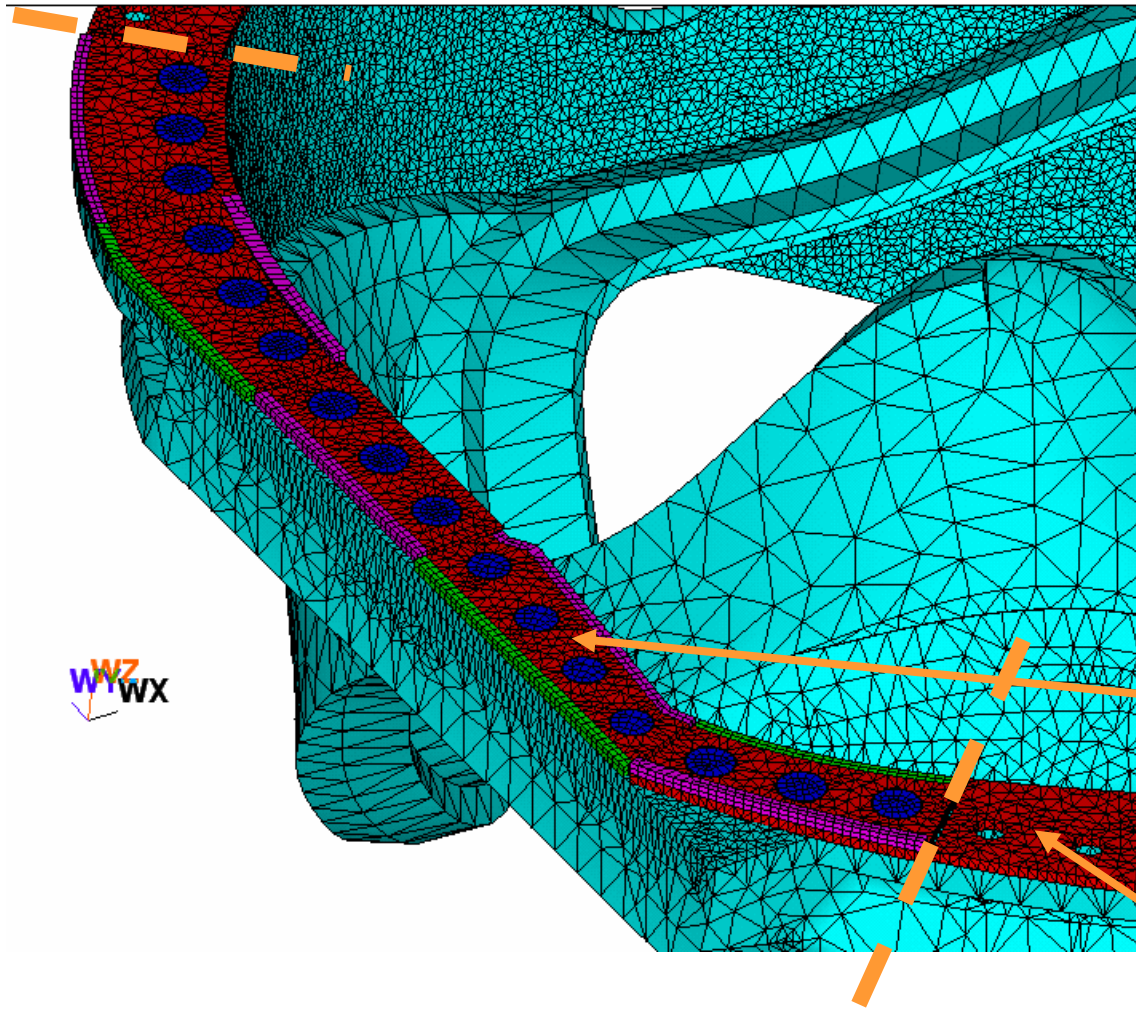


Green welds attach to C casting

Orange welds go to B casting

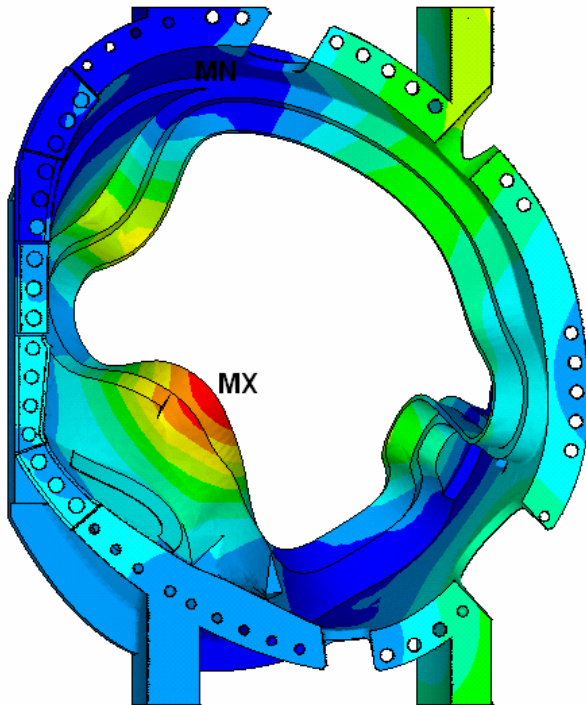
B Casting

BC Analysis (same procedure as AB)



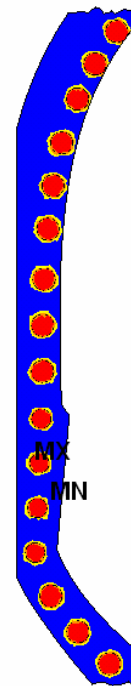
- Green welds attach to the C casting and shim only
- Pink welds attach to the B casting and shim
- Blue pucks connect are bonded to the C flange and slide on the B flange.
- Blue pucks do not interact with the red shim.
- The red shim does not interact with the casting surfaces in the region of the weld.
- The red shim is bonded to the flanges (bolt/friction) approx on the outboard areas away from the weld.

BC Deformation and puck check



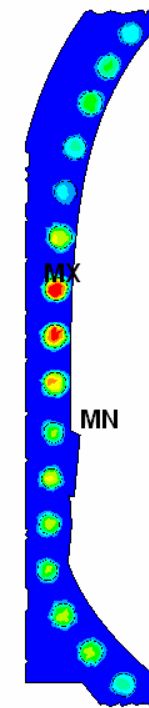
ANSYS 10.0A1
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
USUM (AVG)
RSYS=0
PowerGraphics
EFACET=1
AVRES=Mat
DMX =.001609
SMN =.913E-05
SMX =.001609

Blue	.913E-05
Light Blue	.187E-03
Light Cyan	.365E-03
Cyan	.543E-03
Green	.720E-03
Light Green	.898E-03
Yellow	.001076
Orange	.001254
Red-Orange	.001432
Red	.001609



ANSYS 10.0A1
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
CONTSTAT (AVG)
RSYS=0
PowerGraphics
EFACET=1
AVRES=Mat
DMX =.472E-03
SMX =3

Blue	FarOpen
Yellow	NearContact
Orange	Sliding
Red	Sticking



ANSYS 10.0A1
AVG ELEMENT SOLUTION
STEP=1
SUB =1
TIME=1
CONTSOLID (AVG)
DMX =.467E-03
SMX =.217E-03

Blue	0
Light Blue	.241E-04
Light Cyan	.482E-04
Cyan	.723E-04
Green	.964E-04
Light Green	.121E-03
Yellow	.145E-03
Orange	.169E-03
Red-Orange	.193E-03
Red	.217E-03

Pucks are stuck
on C flange

Pucks slide on B
flange (0.217 mm)

BC Stress Intensities

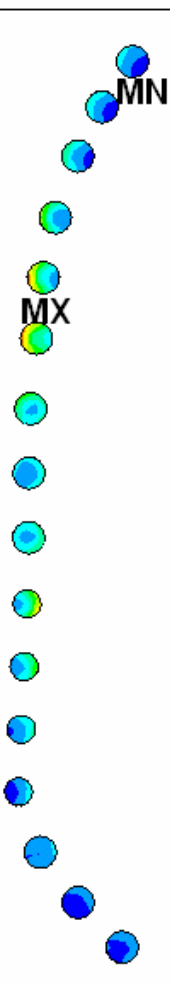


Welds

ANSYS 10.0A1
AVG ELEMENT
STEP=1
SUB =1
TIME=1
SI_PSI (AVG)
DMX =.466E-03
SMN =327.733
SMX =27259

327.733
3320
6312
9305
12297
15290
18282
21274
24267
27259

Units = psi

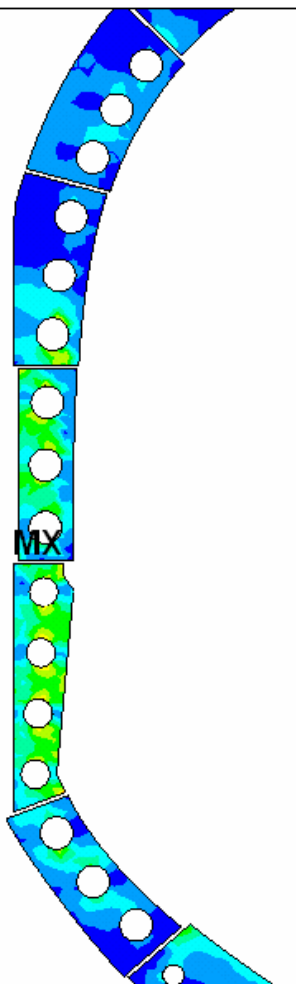


Pucks

ANSYS 10.0A1
AVG ELEMENT
STEP=1
SUB =1
TIME=1
SI_PSI (AVG)
DMX =.469E-03
SMN =551.793
SMX =36747

551.793
4573
8595
12617
16639
20660
24682
28704
32725
36747

Units = psi



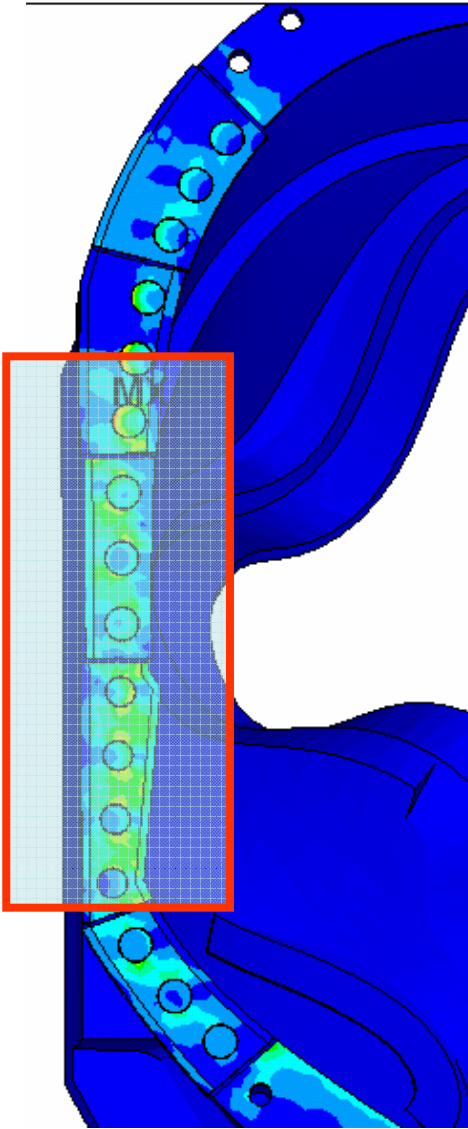
Shims

ANSYS 10.0A1
AVG ELEMENT
STEP=1
SUB =1
TIME=1
SI_PSI (AVG)
DMX =.001105
SMN =4.412
SMX =35781

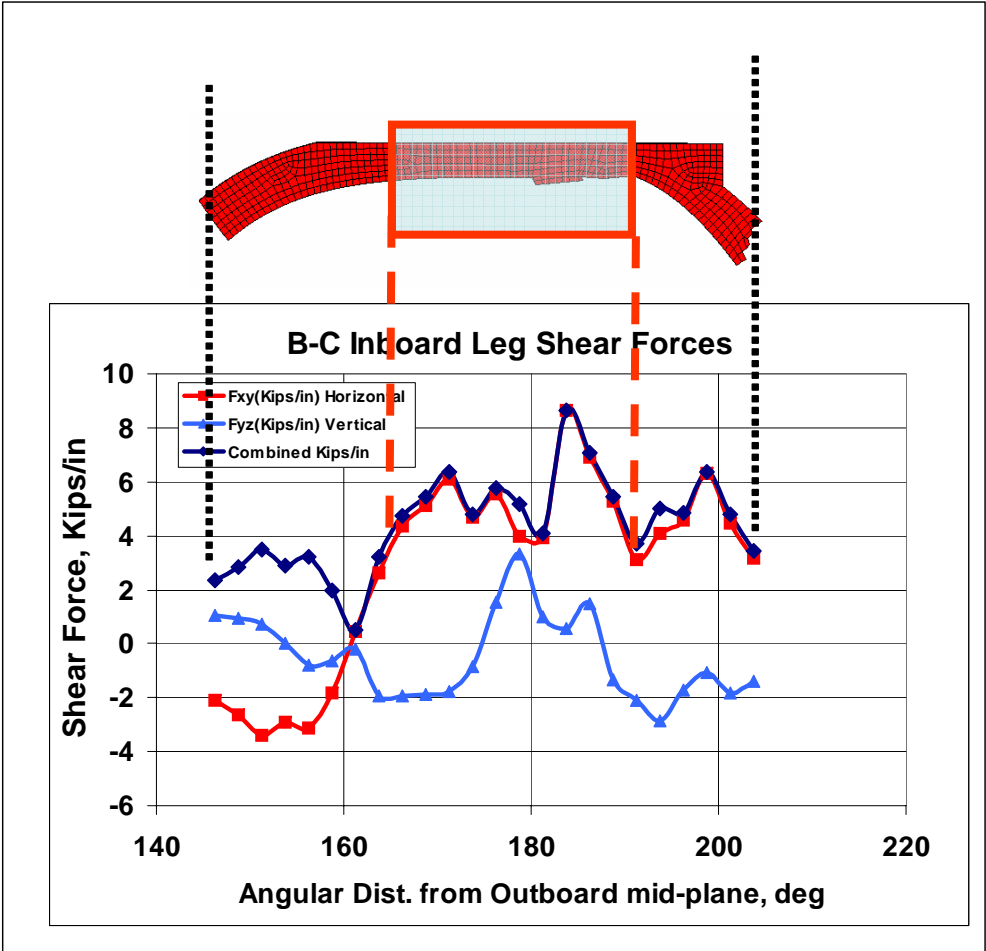
4.412
3980
7955
11930
15905
19880
23856
27831
31806
35781

Units = psi

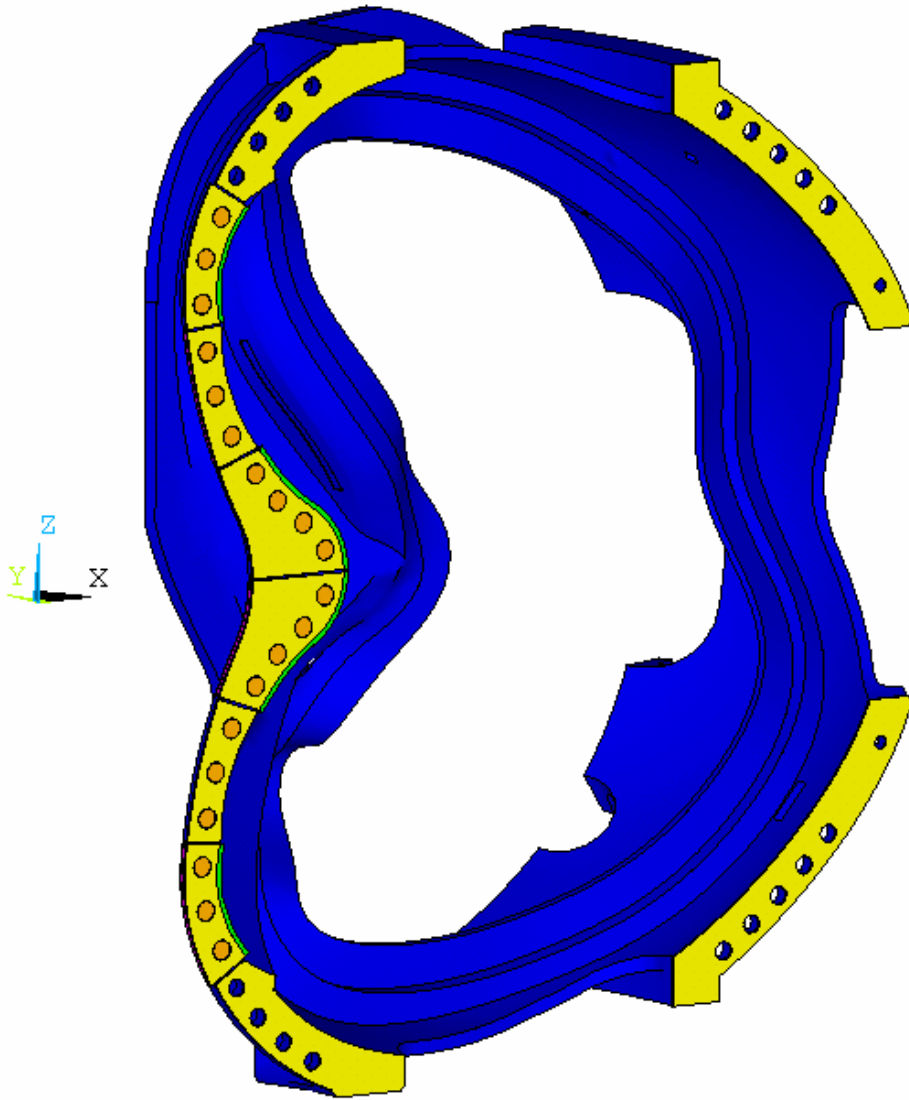
Stress Check



ANSYS 10.0A1
 AVG ELEMENT
 STEP=1
 SUB =1
 TIME=1
 SI_PSI (AVG)
 DMX =.001609
 SMN =4.412
 SMX =36747
 4.412
 4087
 8169
 12252
 16334
 20417
 24500
 28582
 32665
 36747



AA Model (special case)



Recall that the ends of the half period have anti-cyclic symmetry conditions applied to them.

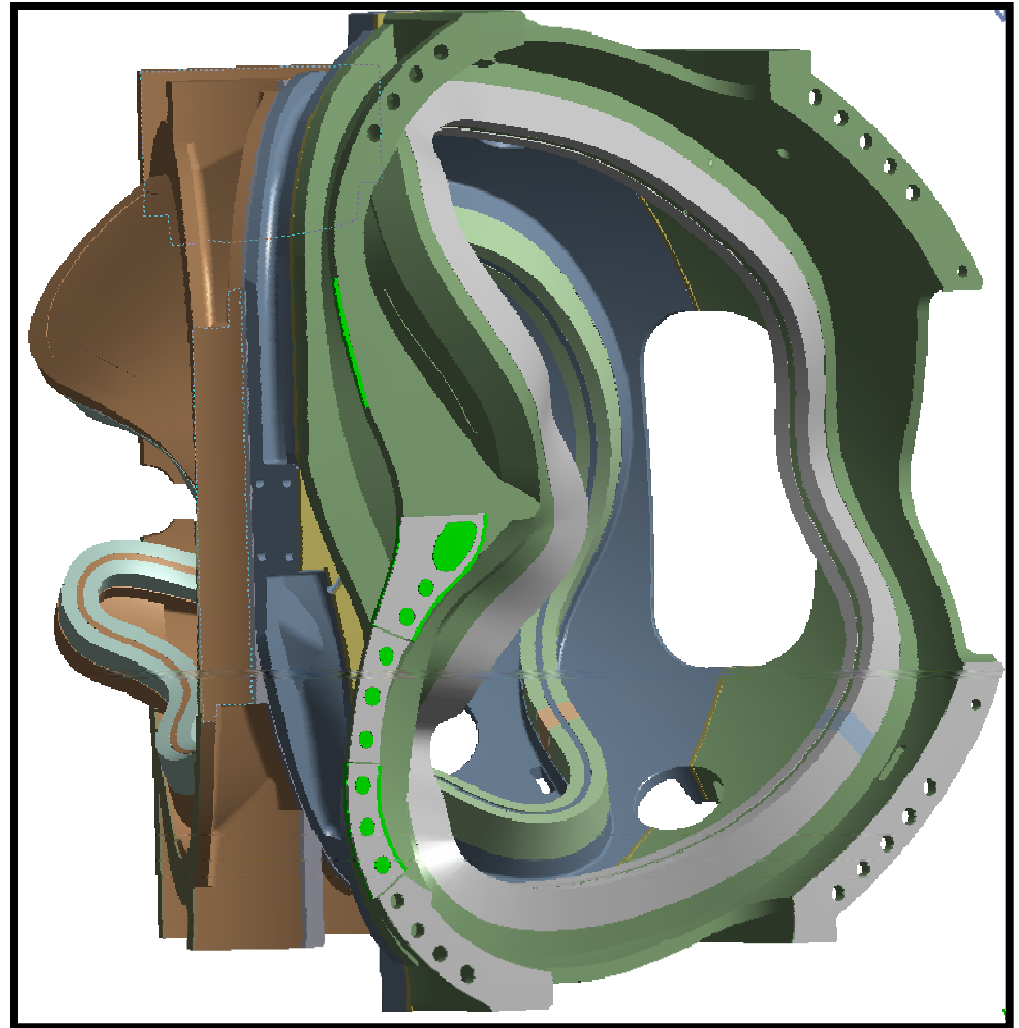
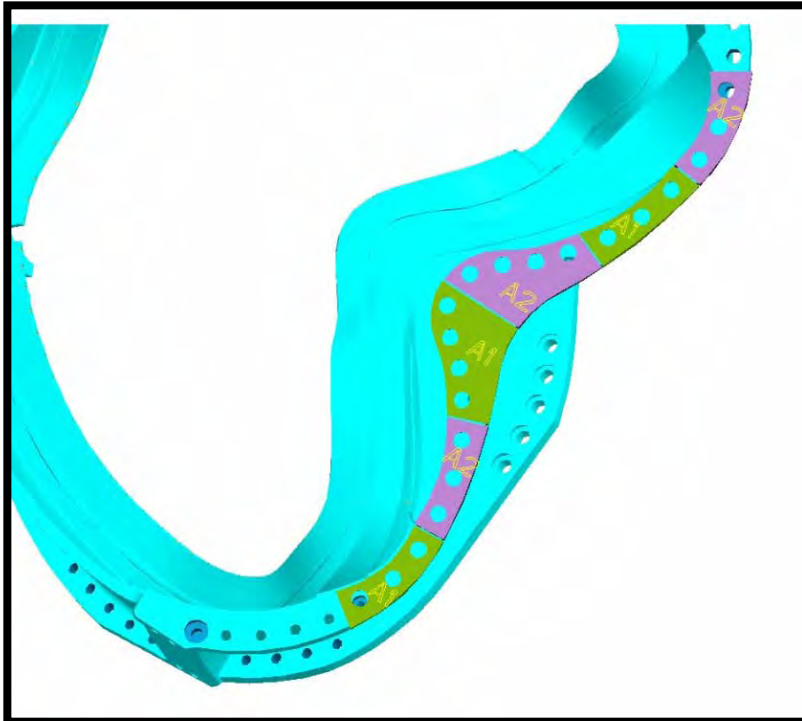
This requires constraint equations between nodes and the mesh must be identical above and below the midplane.

The pucks are reflected and are only connected in the y direction to their mating pair below the midplane. Thus, they are free to move in the plane of the interface and they still take the compressive load.

The welds are reflected and appropriate contact interfaces are made between the shim and flanges.

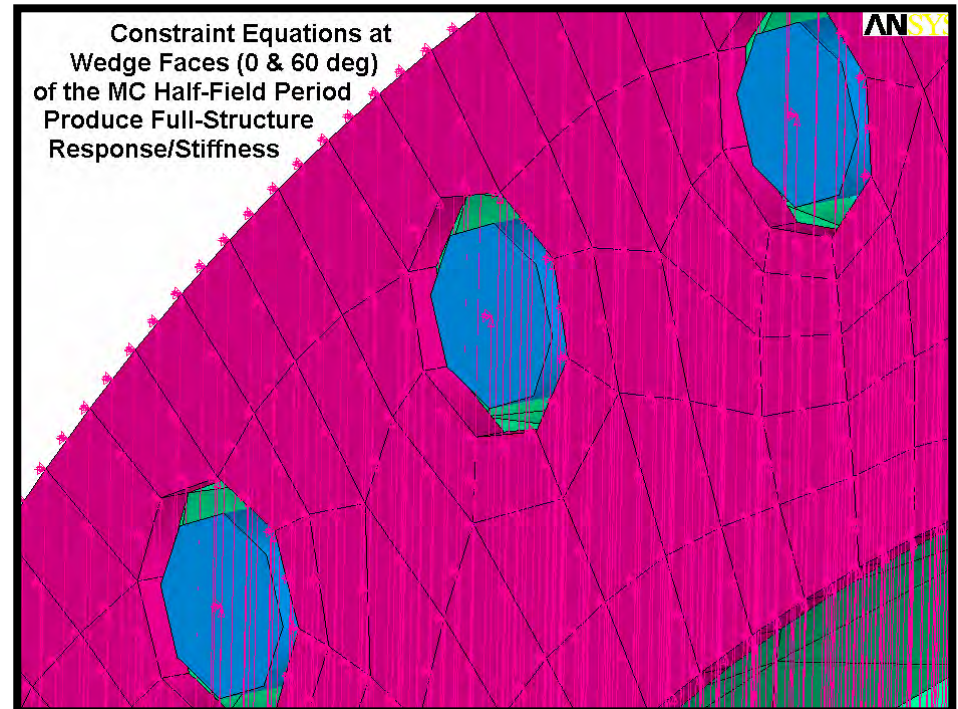
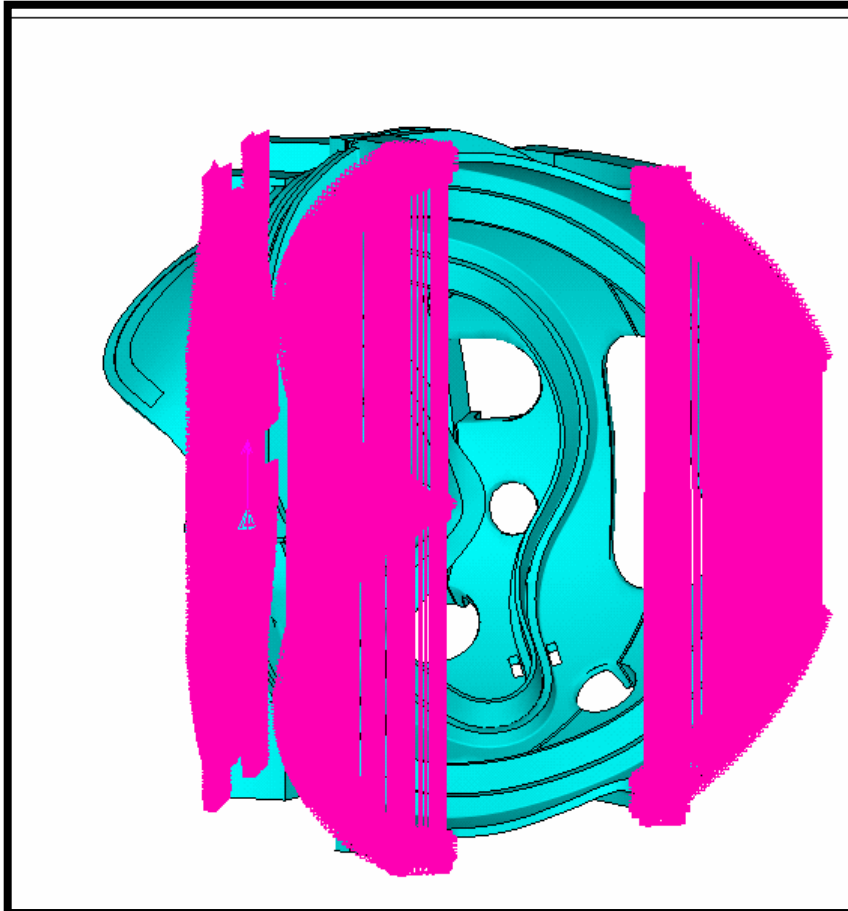
The shims are not connected to the AA flange in the region of the weld. (shear only)

AA shim layouts



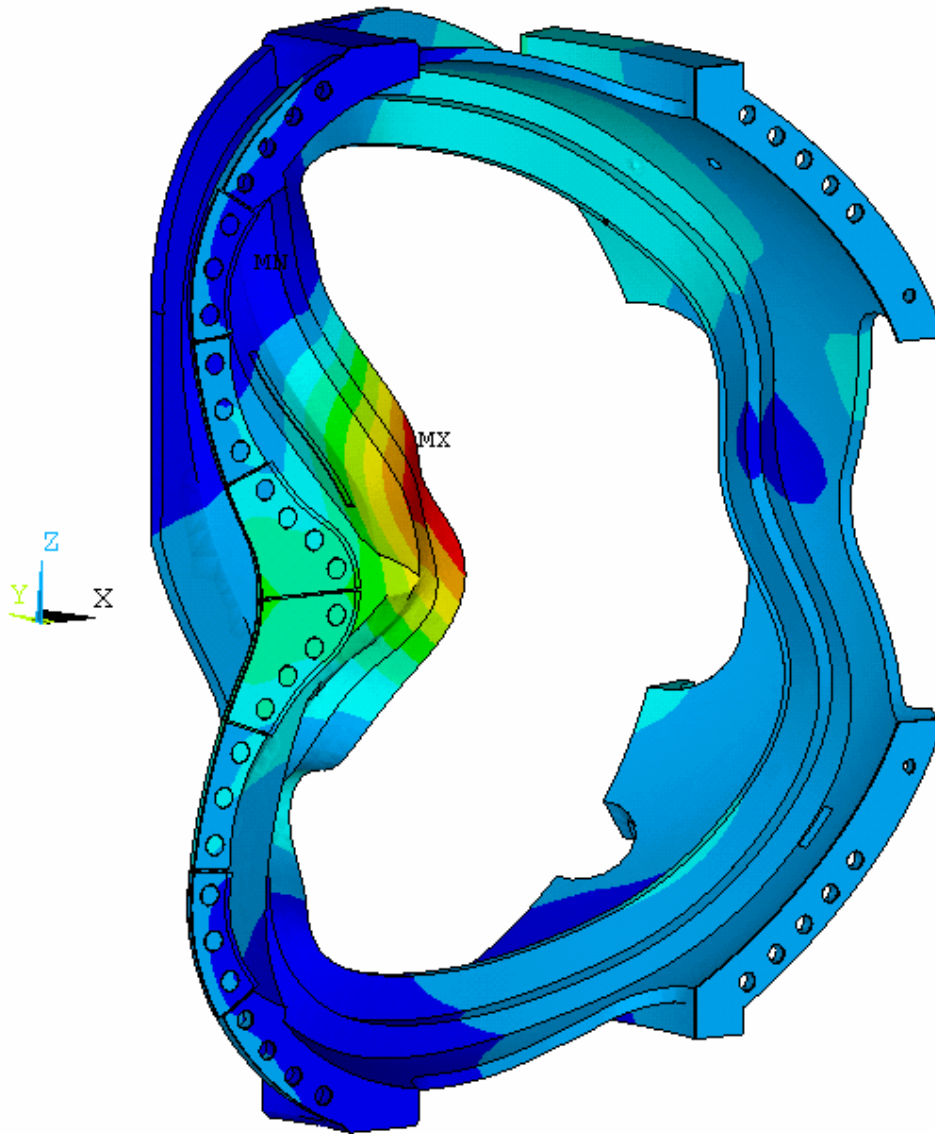
PROE model showing half shim before transfer to ANSYS and symmetry flipping of shim and welds.

Constraint equations



This model was used to model the bolt loads using tie elements, but procedure is the same for the welded models

Global Deflection results (meters)



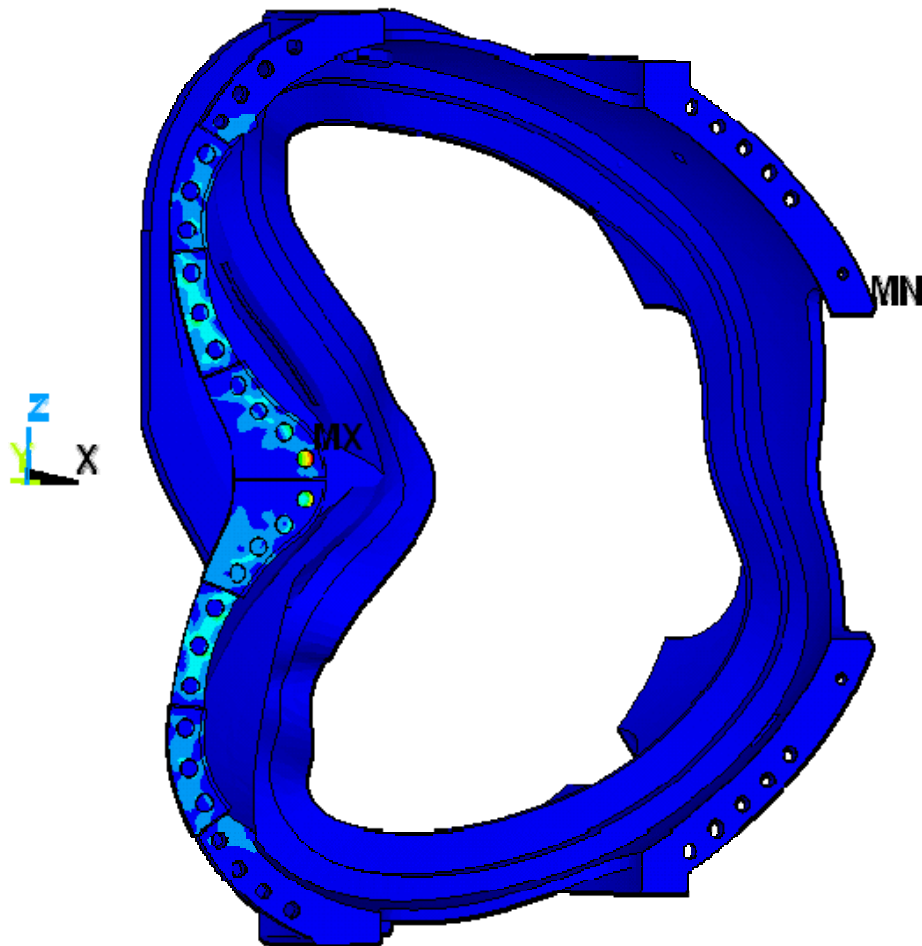
```

ANSYS 11.0
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
USUM      (AVG)
RSYS=0
PowerGraphics
EFACET=1
AVRES=Mat
DMX =.001872
SMN =.430E-04
SMX =.001872
.430E-04
.246E-03
.450E-03
.653E-03
.856E-03
.001059
.001263
.001466
.001669
.001872
    
```

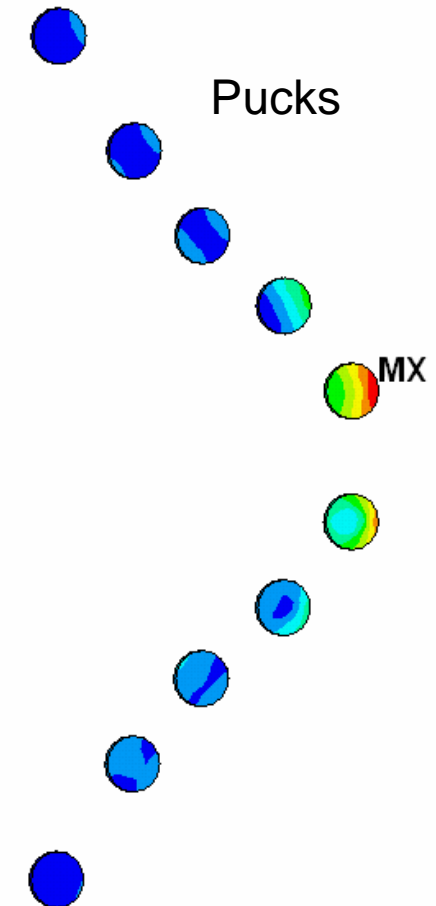
Units = meters

Stress Intensity (psi)

1

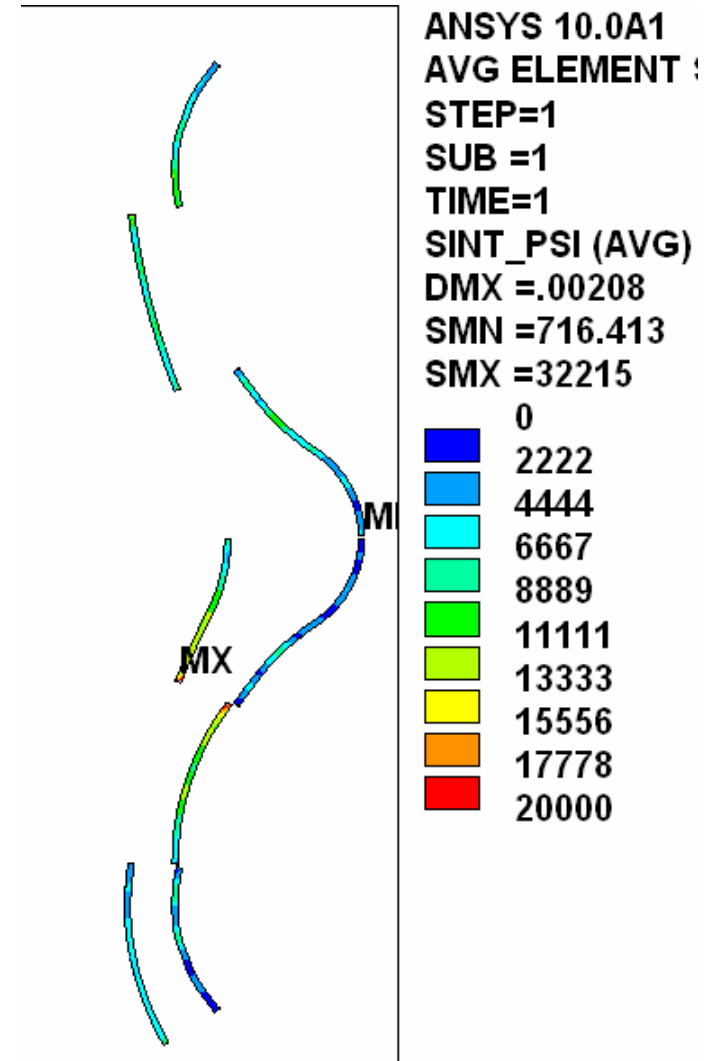
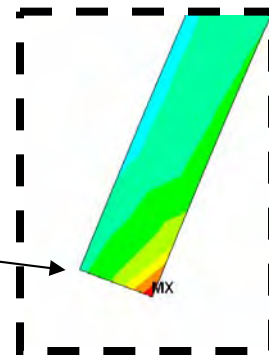
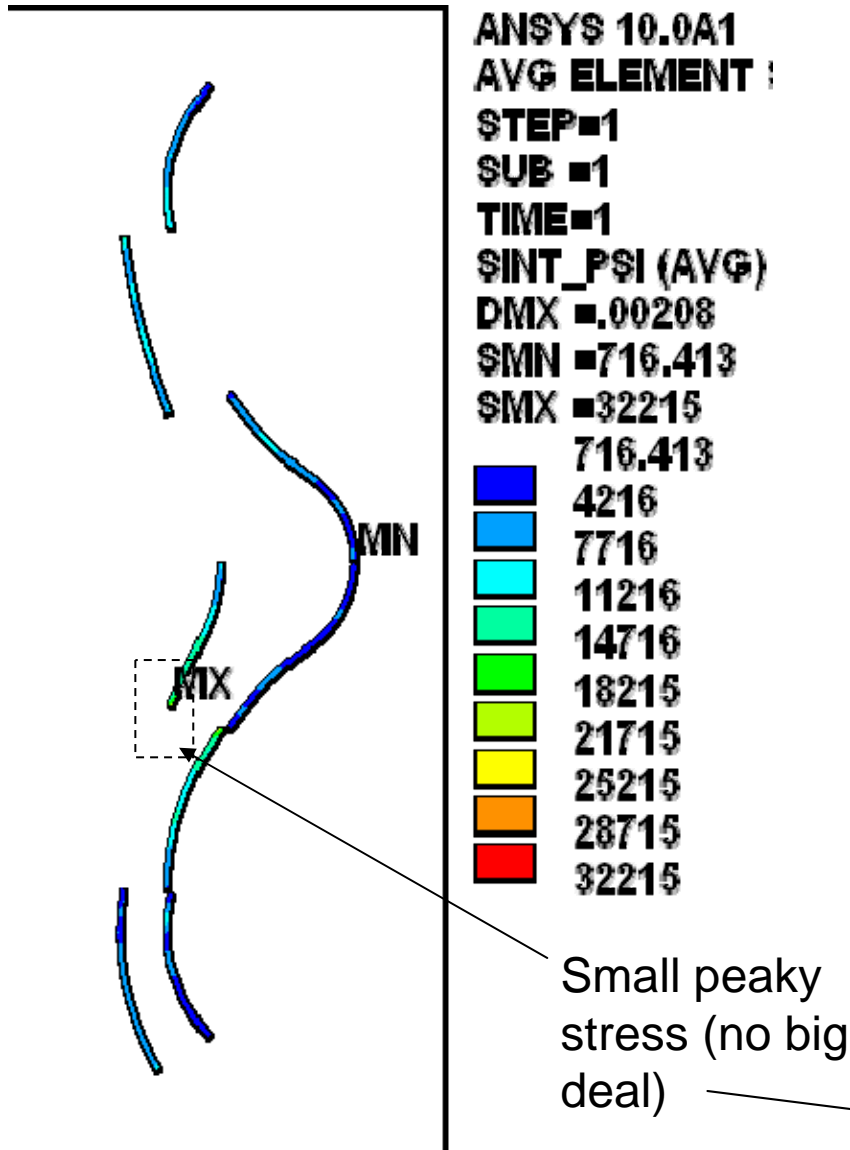


ANSYS 10.0A1
AVG ELEMENT
STEP=1
SUB =1
TIME=1
\$INT_PSI (AVG
DMX =.00208
\$MN =24.267
\$MX =56896



The first two pucks closest to the midplane are seeing the most compression.

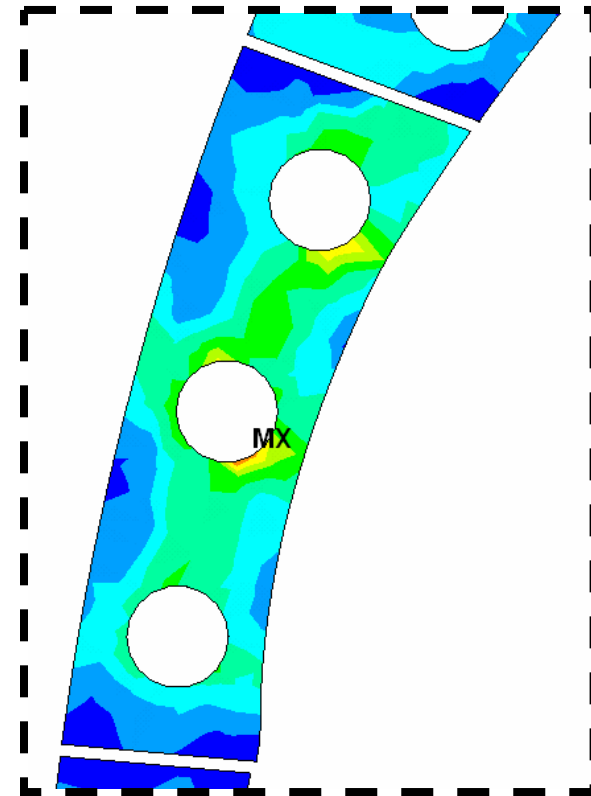
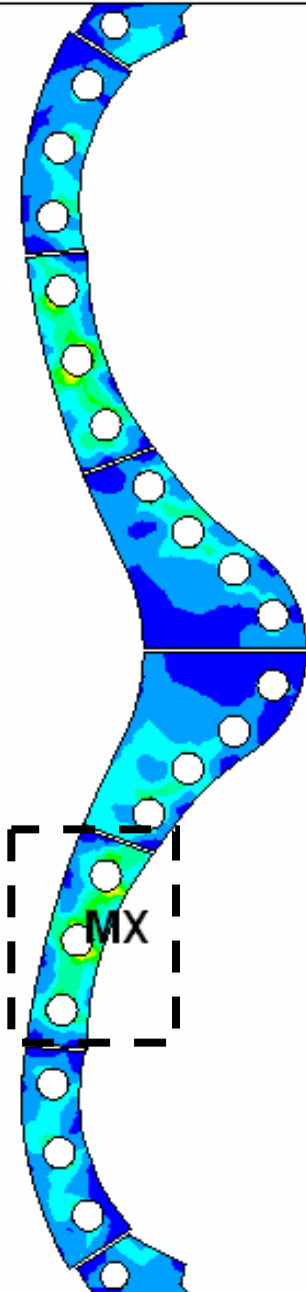
Weld Stresses (psi)



Stress Rescaled to max of 20 ksi. (very small end effects exceed this value.)

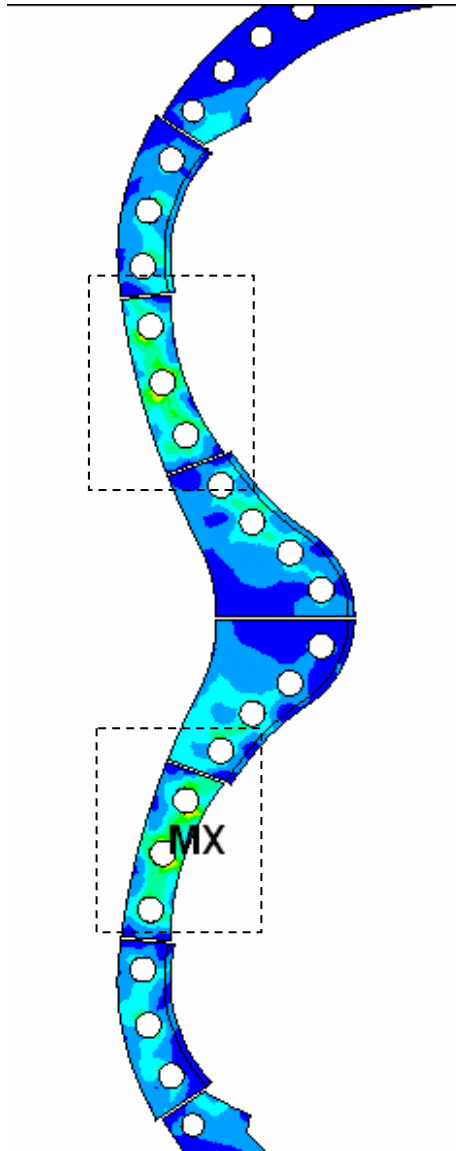
Shim Stresses (psi)

ANSYS 10.0A1
AVG ELEMENT
STEP=1
SUB =1
TIME=1
SINT_PSI (AVG)
DMX =.00208
SMN =24.267
SMX =37633



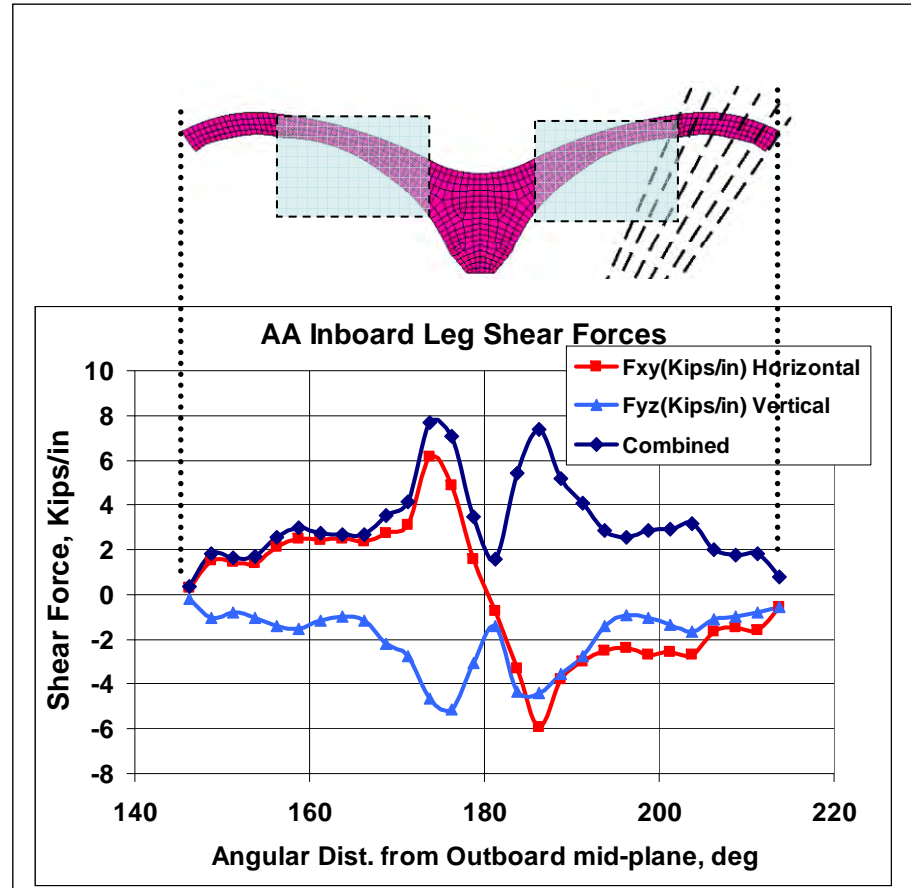
Small localized stress which can go to $1.5 \cdot S_m$ where S_m is 68 ksi for 316LN.

Shear check on AA

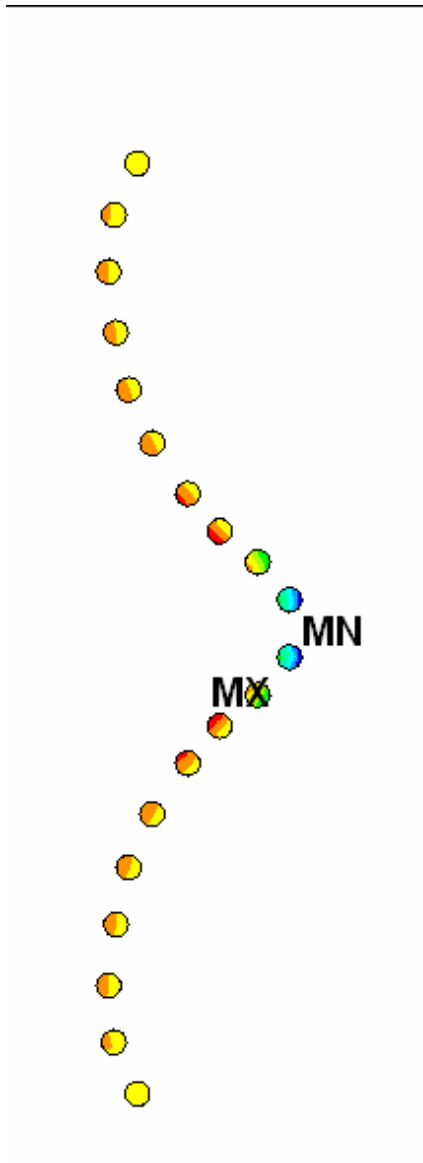


ANSYS 10.0A1
 AVG ELEMENT
 STEP=1
 SUB =1
 TIME=1
 SINT_PSI (AVG)
 DMX =.00208
 SMN =24.267
 SMX =37633

24.267
4203
8382
12560
16739
20918
25096
29275
33454
37633

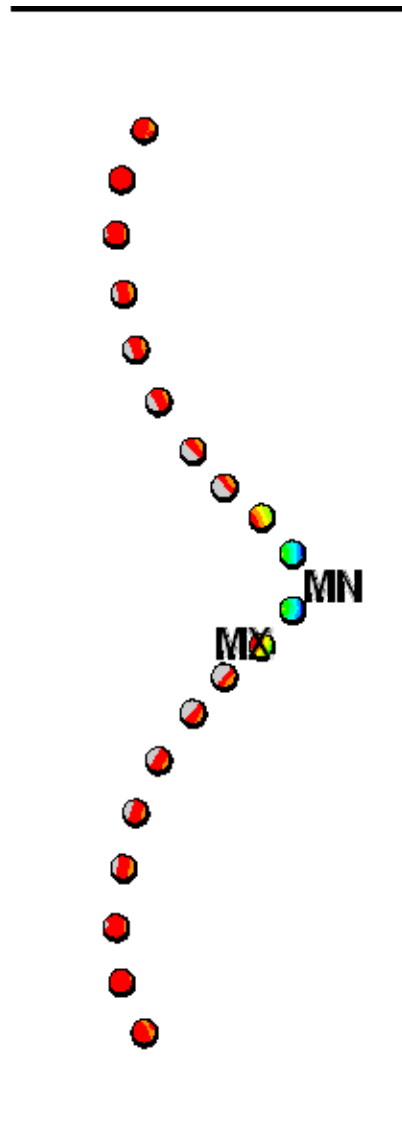


Normal Stress on Pucks (psi)



ANSYS 10.0A1
 AVG ELEMENT
 STEP=1
 SUB =1
 TIME=1
 SY_PSI (AVG)
 DMX =.00208
 SMN =-60179
 SMX =13263

Blue	-60179
Light Blue	-52019
Cyan	-43859
Light Green	-35698
Green	-27538
Yellow-Green	-19378
Yellow	-11217
Orange	-3057
Red-Orange	5103
Red	13263



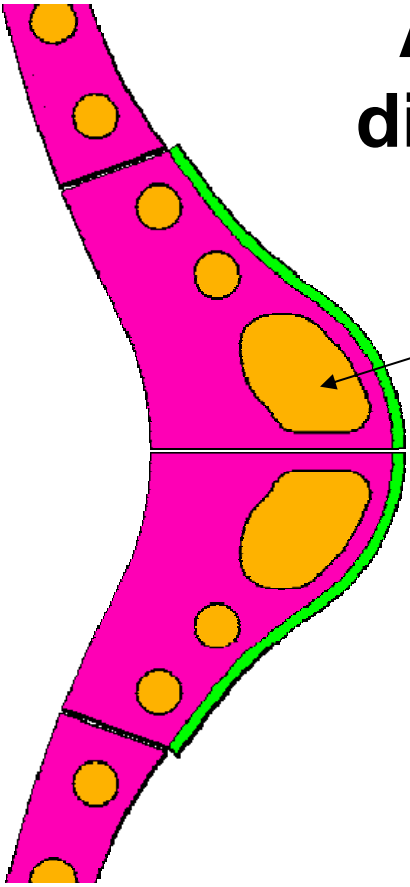
ANSYS 10.0A1
 AVG ELEMENT
 STEP=1
 SUB =1
 TIME=1
 SY_PSI (AVG)
 DMX =.00208
 SMN =-60179
 SMX =13263

Blue	-61000
Light Blue	-54222
Cyan	-47444
Light Green	-40667
Green	-33889
Yellow-Green	-27111
Yellow	-20333
Orange	-13556
Red-Orange	-6778
Red	0

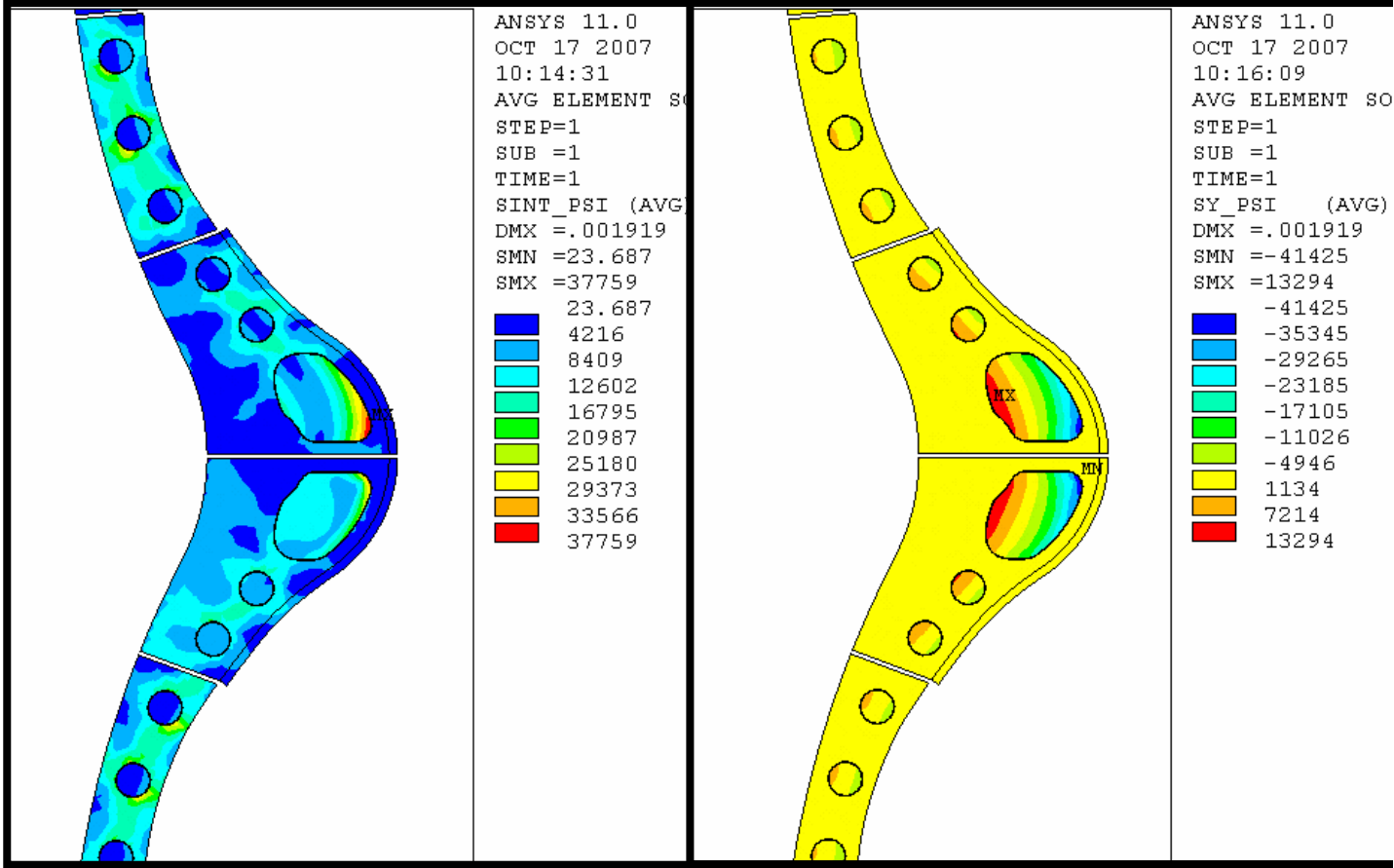
Entire puck set rescaled to min zero

Current Solution method allows the pucks on the AA interface to carry Tension!

Amoeba Plug – An attempt to distribute the compressive load.



Elongated compressive plug



Stress Intensity (psi)

Y stress (normal to flange) psi

Weld Analysis Summary Table

Flange	Peak weld stress Intensity (edges) (ksi)	Average weld stress Intensity eyeball range (ksi)	Peak Shim Stress Intensity (ksi)	Peak Puck Compressive stress (ksi)
AA	32	12-15	38	61
AB	61* (26)	15-18	42	24
BC	27	15-18	36	37

Preliminary Analysis Comments

- Peak weld stress of 61 ksi on AB is anomalous as there will be additional weld material immediately near it. It is extremely peaky and likely due to sharp geometric cornering.
- Average stresses across weld when not considering those regions are approximately 15-20 ksi.
- Global Fea Model is currently non-conservative in that it does not include weld segmentation of 4" in with 0.25" gaps. Currently, all welds are continuous. (Small effect anticipated as seen with previous sub-models of older NCSX designs)
- Global Fea Model is currently non-conservative in that it does not model the weld as a fillet weld and instead uses the easier to construct square type welds. MIG welds actually resemble more square shape than triangular anyway. (Small effect anticipated as seen with previous sub-models of older NCSX designs)
- AA weld analysis is conservative in that the pucks in the model currently carry tension. Peak compressive stress will be lower than shown.
- The global Model has shown that a 0.5" fillet welds (0.35" throat) are adequate as modeled to support the shear loads on the AA, AB and BC flanges.
- Quick check FEA model verifies a small effect on the smaller weld size [0.44" (0.31" throat)] and shape.

- Peer review of inboard welded shims conducted on May-18
- No chits submitted, but weld distortion identified as main risk
- Actions planned at that time:
 - Shim layout
 - Structural analysis of welds
 - Access for welding
 - Development of weld procedure
 - Material selection
 - Deformation control options

Plan to complete final design

Task	Description	Proposed	Scheduled
INTRF-040	Analysis of tensile loads (ORNL)	18-Oct-07	15-Aug-07
INTRF-064	PDR	18-Oct-07	18-Oct-07
INTRF-054	FDR prep AB/BC/AA inboard Interface	19-Oct-07	4-Sep-07
INTRF-055	AB/BC/AA inboard interface - FDR	22-Nov-07	4-Sep-07
1421-3138	Resolve issues, release assembly spec&drawings	23-Nov-07	11-Sep-07
* 1429-3069X	Inboard Shims Available for 1st 3 pack MC assy	11-Dec-07	

- * Note: Task 1429-3069X - We will initiate a risk release after the PDR to start fabrication of the inboard shims. Material for fabricating the shims is in the process of being ordered.

- Are the requirements defined? What is the proposed design?
Requirements are defined. Design is documented and drawings have been developed.
- What is the status of welding trials?
Welding trial samples are have been completed. The type C prototype longitudinal weld test is underway. Welding procedure and welder qualifications will be formalized prior to performing assembly welds.
- Is the analysis consistent with proposed design?
Analysis updated- AA, AB, BC welds are adequate
- Have prior design review chits been addressed?
Peer review identified development activities- underway
- Have all technical, cost, schedule, and safety risks been addressed?