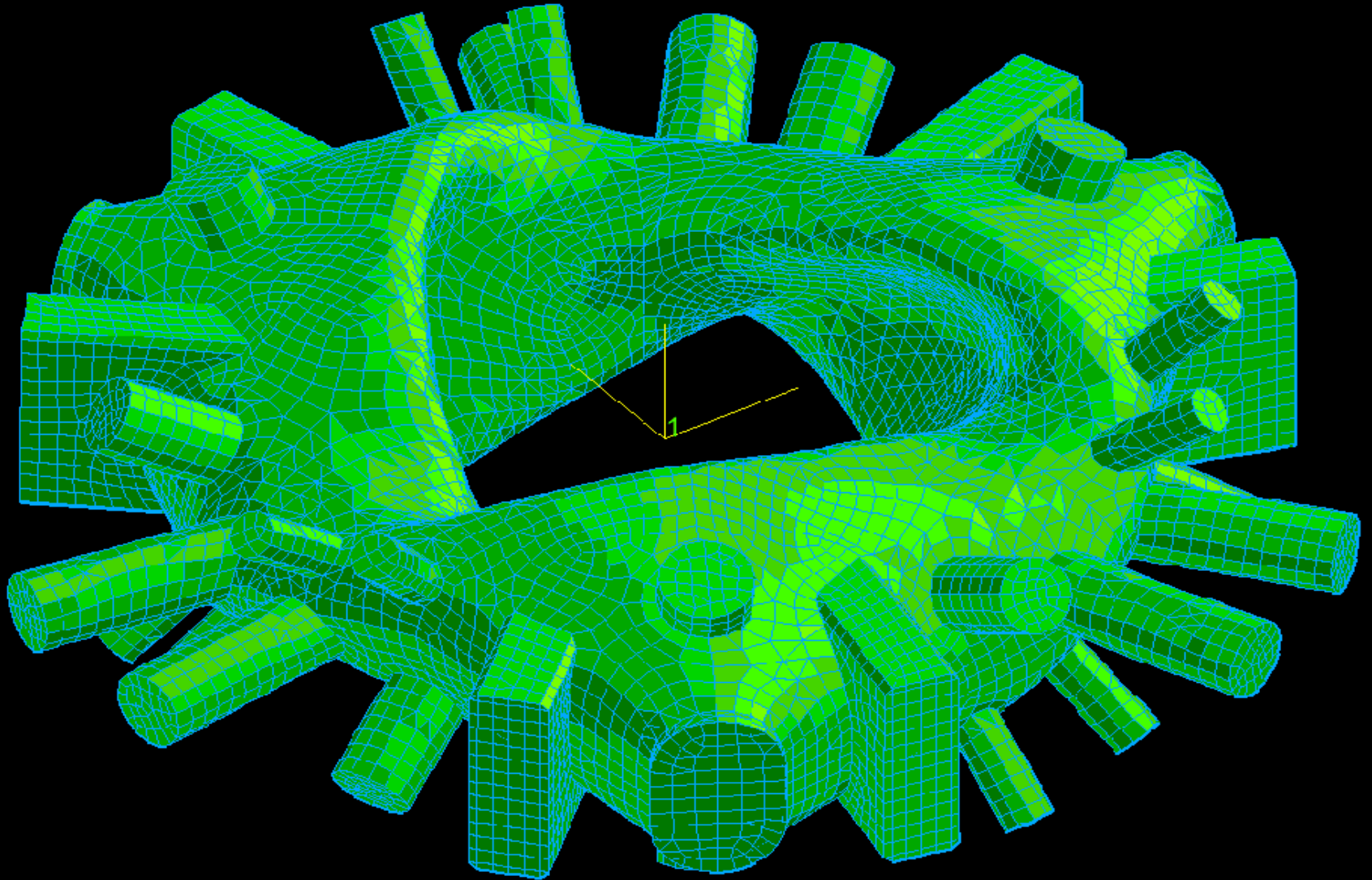


# NCSX-Vessel Buckling

LI386-1 atmosphere, free radial b.c.

Fred Dahlgren - 26 April 2001

# NCSX Vacuum Vessel - Nastran Shell Element Model



## FEA Model Details:

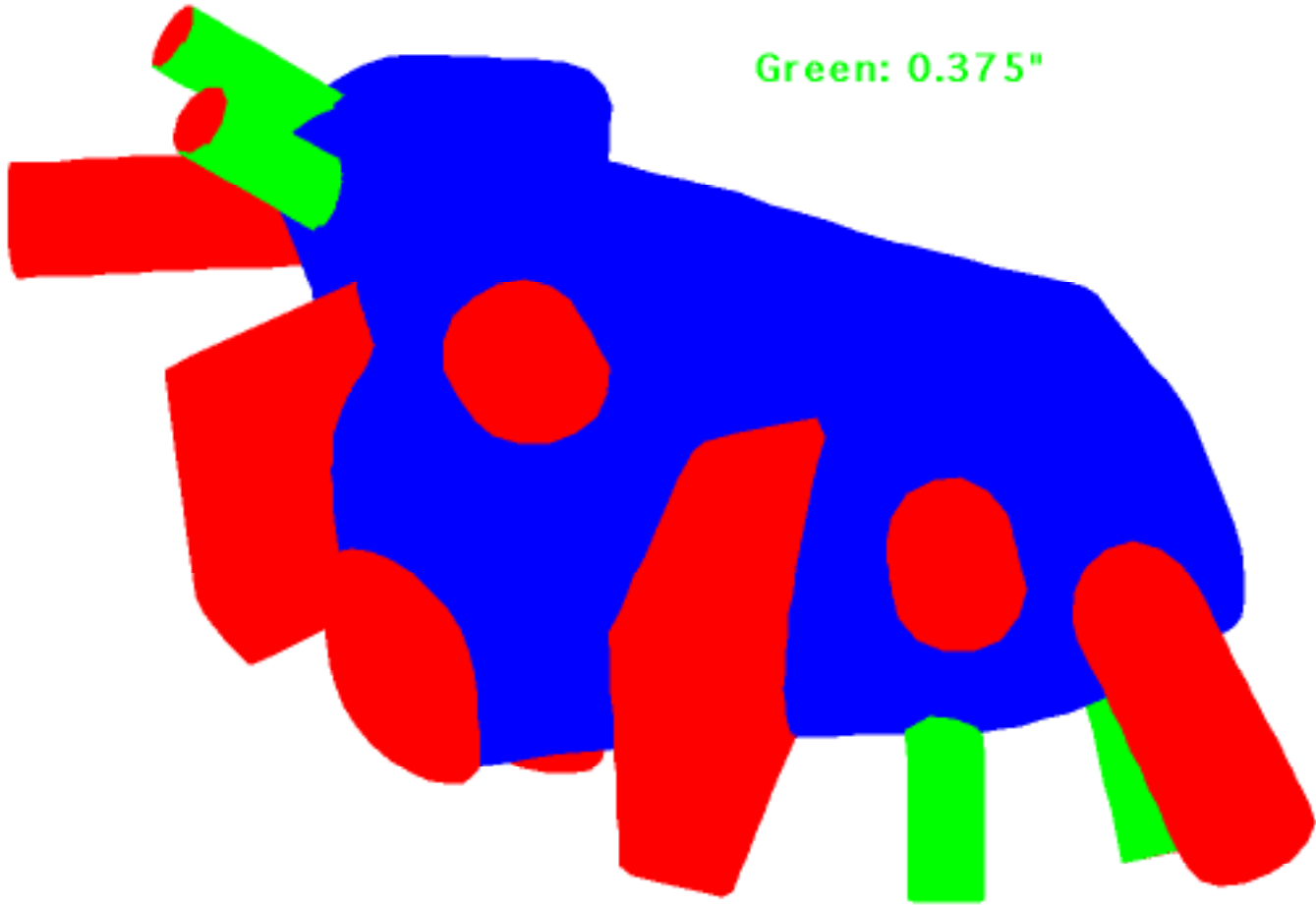
- |              |  |              |
|--------------|--|--------------|
| 1. Elements: | CQUAD4   | 9,876        |
|              | CTRIA3   | <u>4,104</u> |
|              | Total:   | 13,980       |
| 2. Nodes:    | GRIDS  | 11,928       |
|              | (DOF's)  | 71,568 (-6)  |
| 3. B.C.s:    | Vertical & Circumferential SPCs<br>@ NB Port Centerline<br>(6 DOFs, constraining R.B. modes) |              |
| 4.           | Full 360 degrees model – Static pressure load  |              |
| 5.           | Lanzcos Eigenvalue problem using differential<br>Stiffness derived from static analysis.     |              |

# Model A Thicknesses

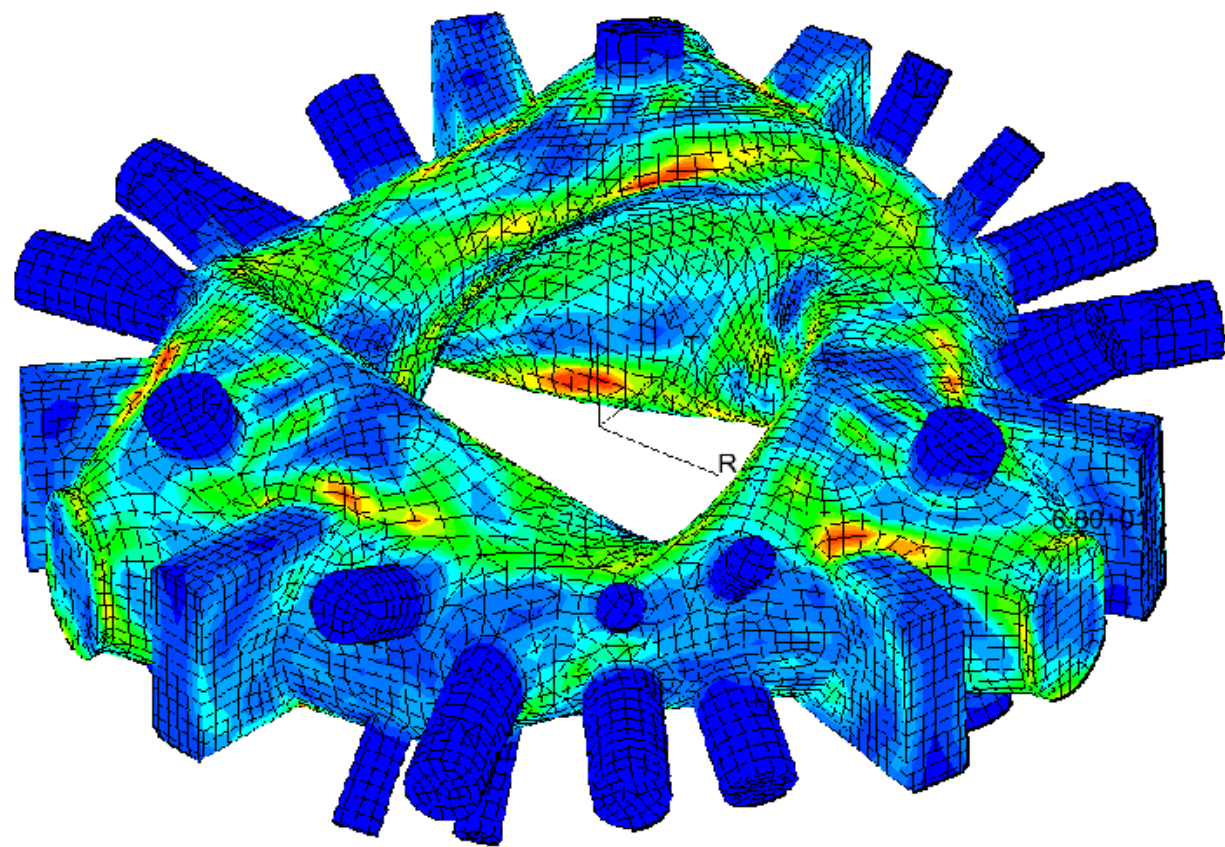
Red: 0.475"

Blue: 0.175"

Green: 0.375"



# Shell Thickness = 0.175



2.03+04

1.90+04

1.76+04

1.63+04

1.49+04

1.36+04

1.22+04

1.09+04

9.51+03

8.16+03

6.81+03

5.47+03

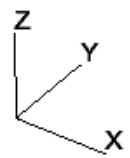
4.12+03

2.77+03

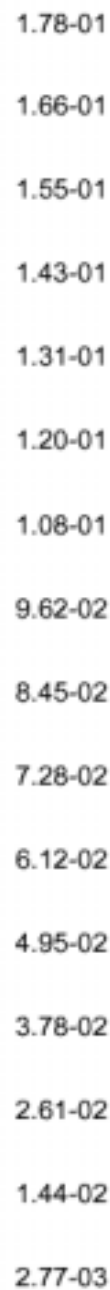
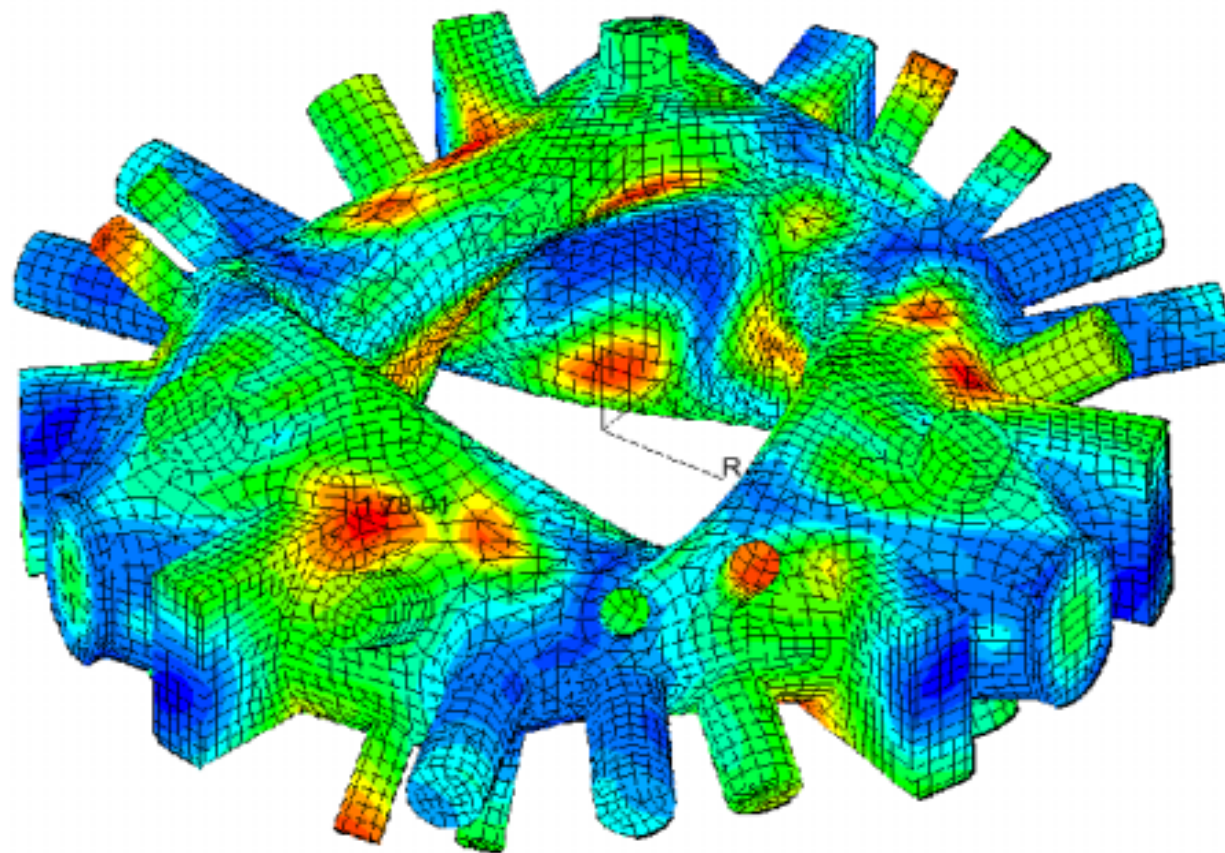
1.42+03

6.80+01

default\_Fringe :  
Max 2.03+04 @Nd 1329  
Min 6.80+01 @Nd 577

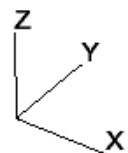
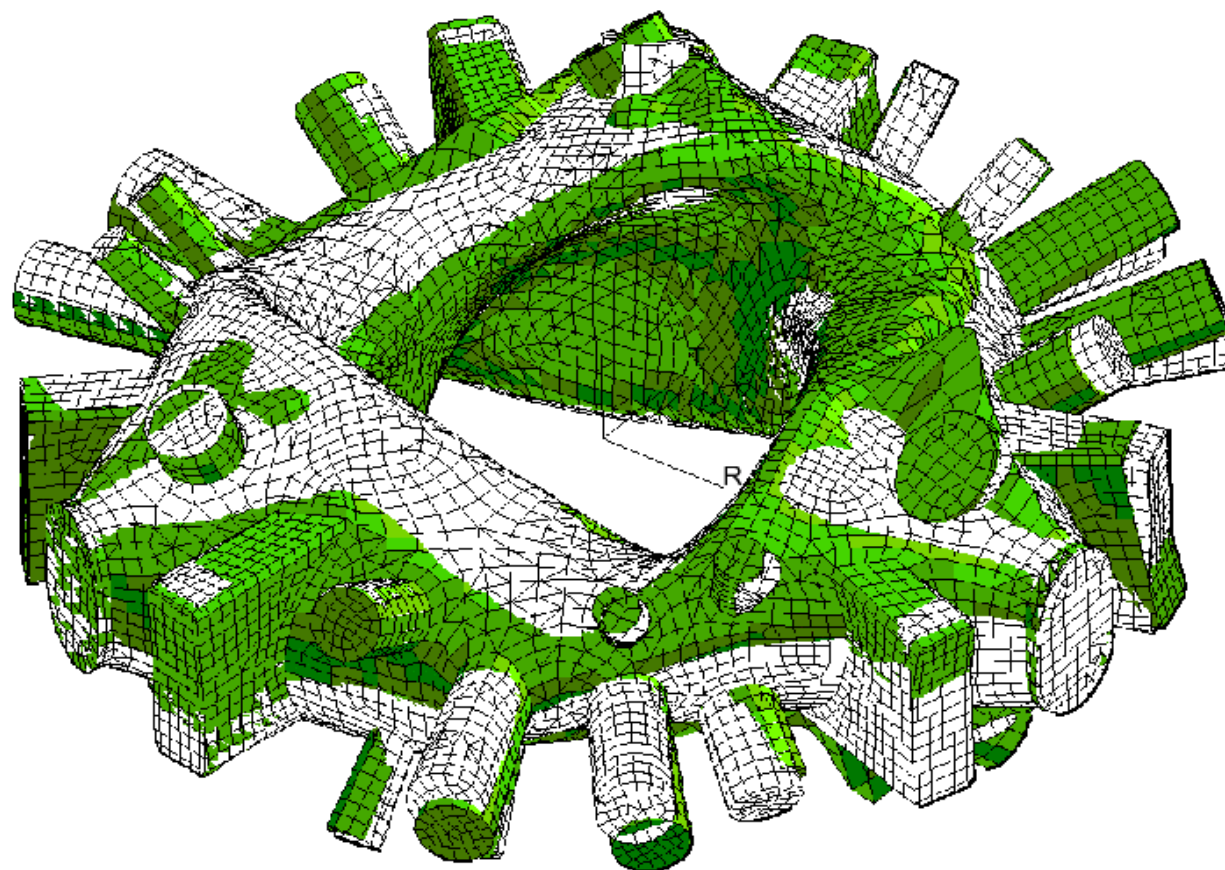


Shell Thickness = 0.175



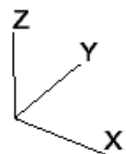
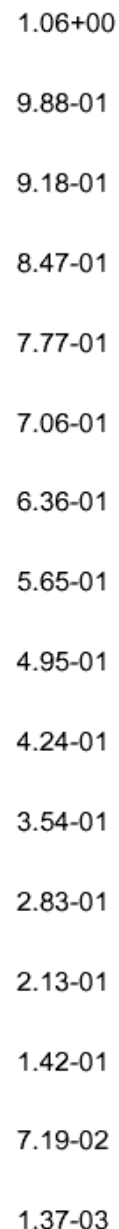
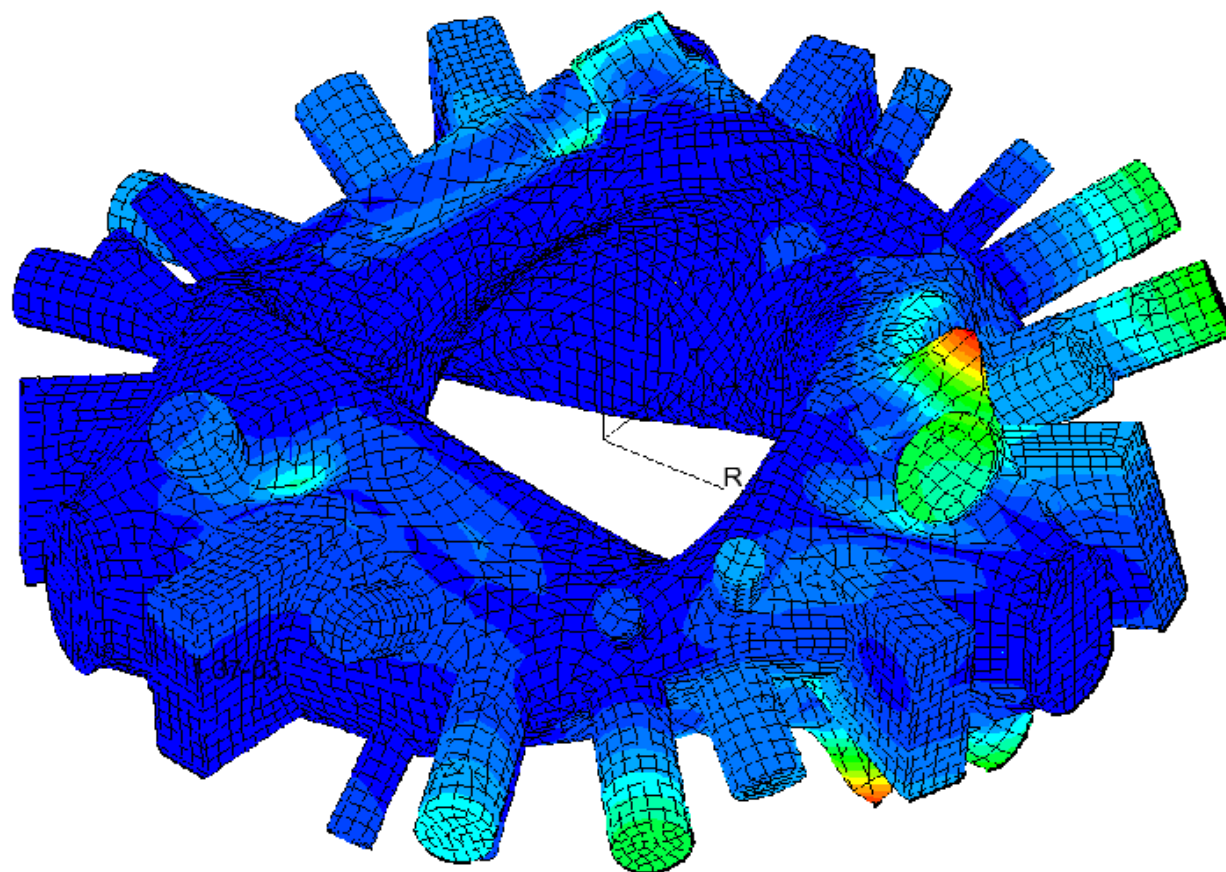
default\_Fringe :  
Max 1.78-01 @Nd 9514  
Min 2.77-03 @Nd 6538

# Shell Thickness = 0.175



## 1<sup>st</sup> Buckling Mode

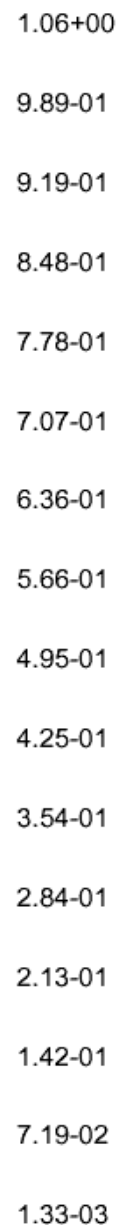
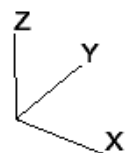
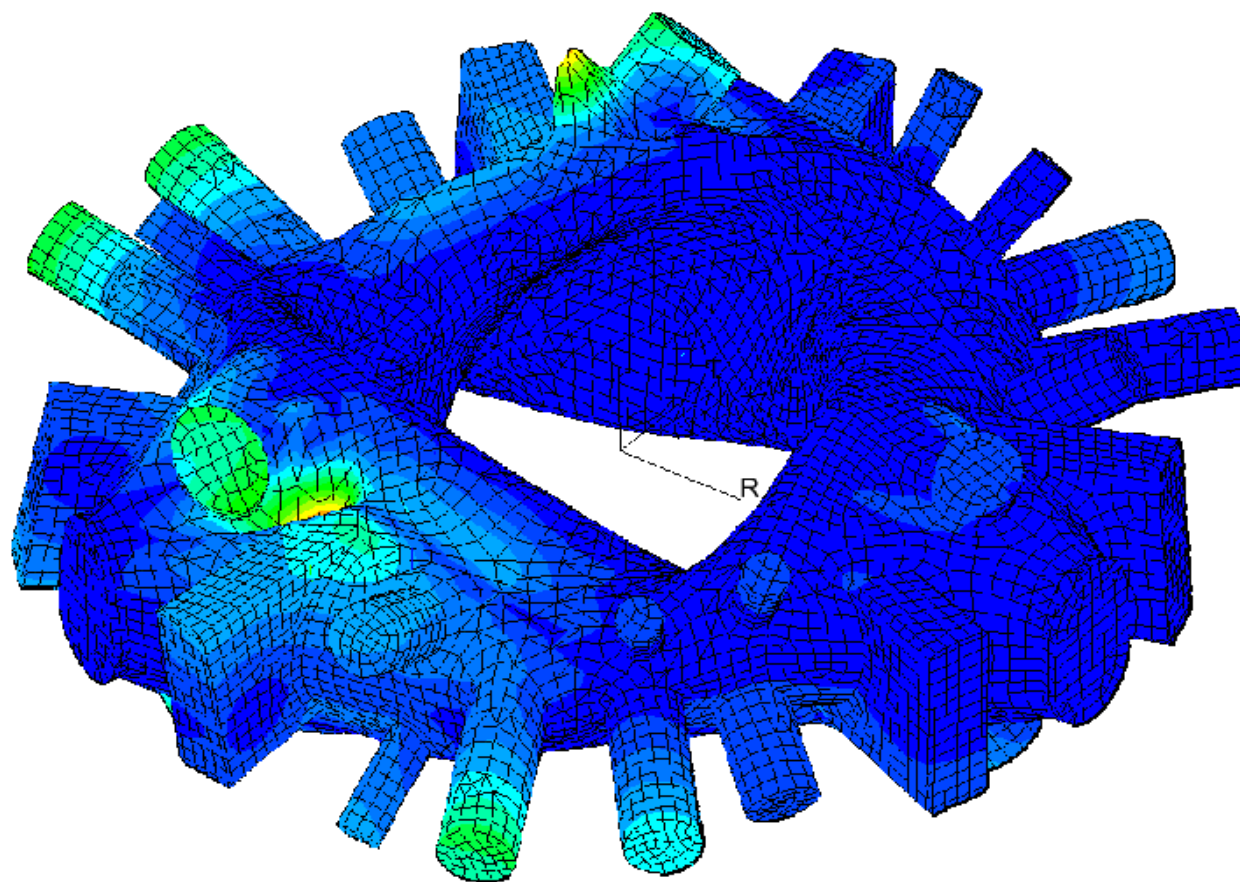
# Shell Thickness = 0.175



default\_Fringe :  
Max 1.06+00 @Nd 1769  
Min 1.37-03 @Nd 10143  
default\_Deformation :  
Max 1.06+00 @Nd 1769



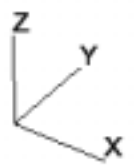
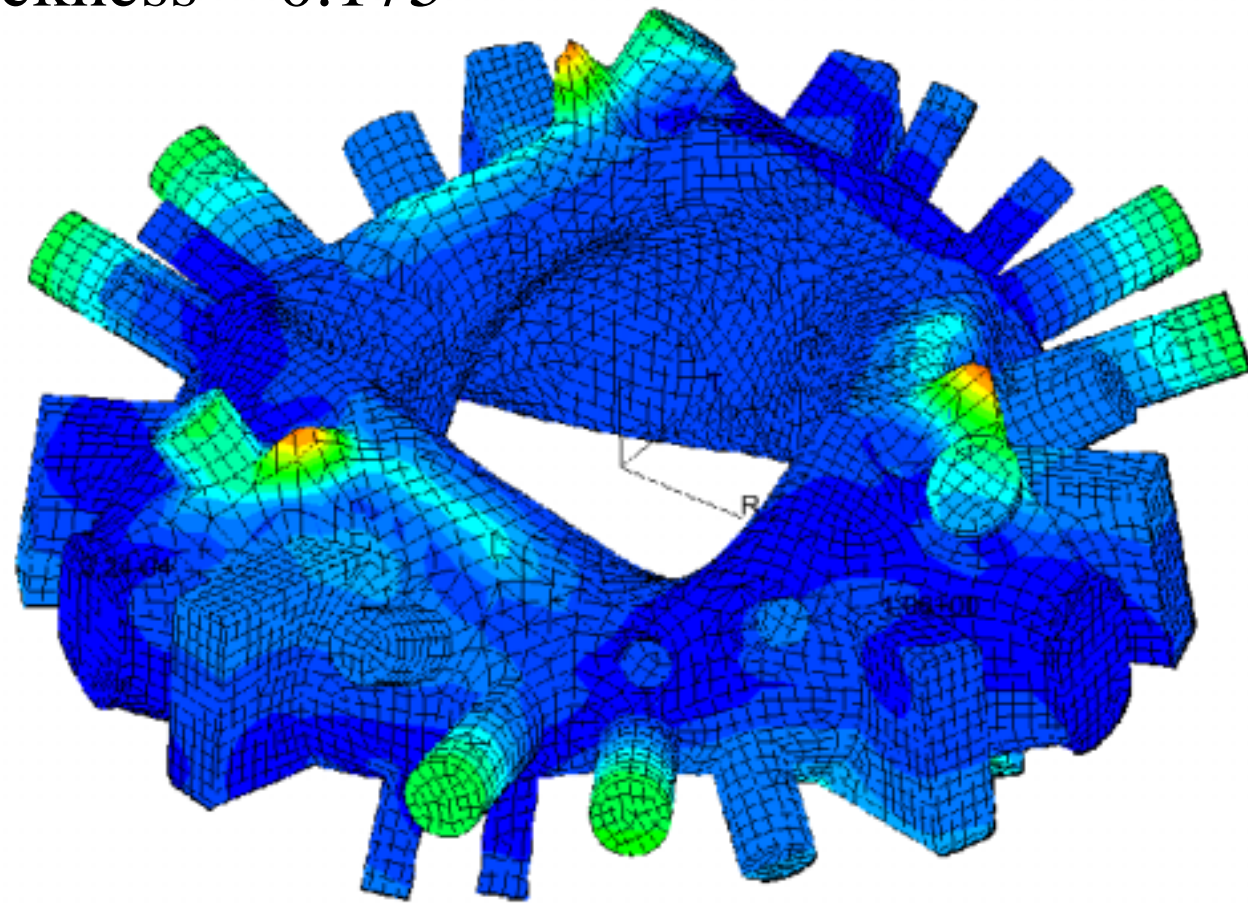
# Shell Thickness = 0.175



default\_Fringe :  
Max 1.06+00 @Nd 11727  
Min 1.33-03 @Nd 1531  
default\_Deformation :  
Max 1.06+00 @Nd 11727

Fringe: 1 ATMOSPHERE (LI386 MODEL), Mode 3 : Factor=2.9624: Eigenvectors, Translational-(NON-LAYERED) (MAG)  
Deform: 1 ATMOSPHERE (LI386 MODEL), Mode 3 : Factor=2.9624: Eigenvectors, Translational

Shell Thickness = 0.175



default\_Fringe :  
Max 1.06+00 @Nd 3814  
Min 3.24-04 @Nd 11983  
default\_Deformation :  
Max 1.06+00 @Nd 3814

1.05+00

9.82-01

9.12-01

8.42-01

7.72-01

7.03-01

6.33-01

5.63-01

4.93-01

4.23-01

3.53-01

2.83-01

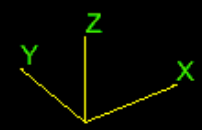
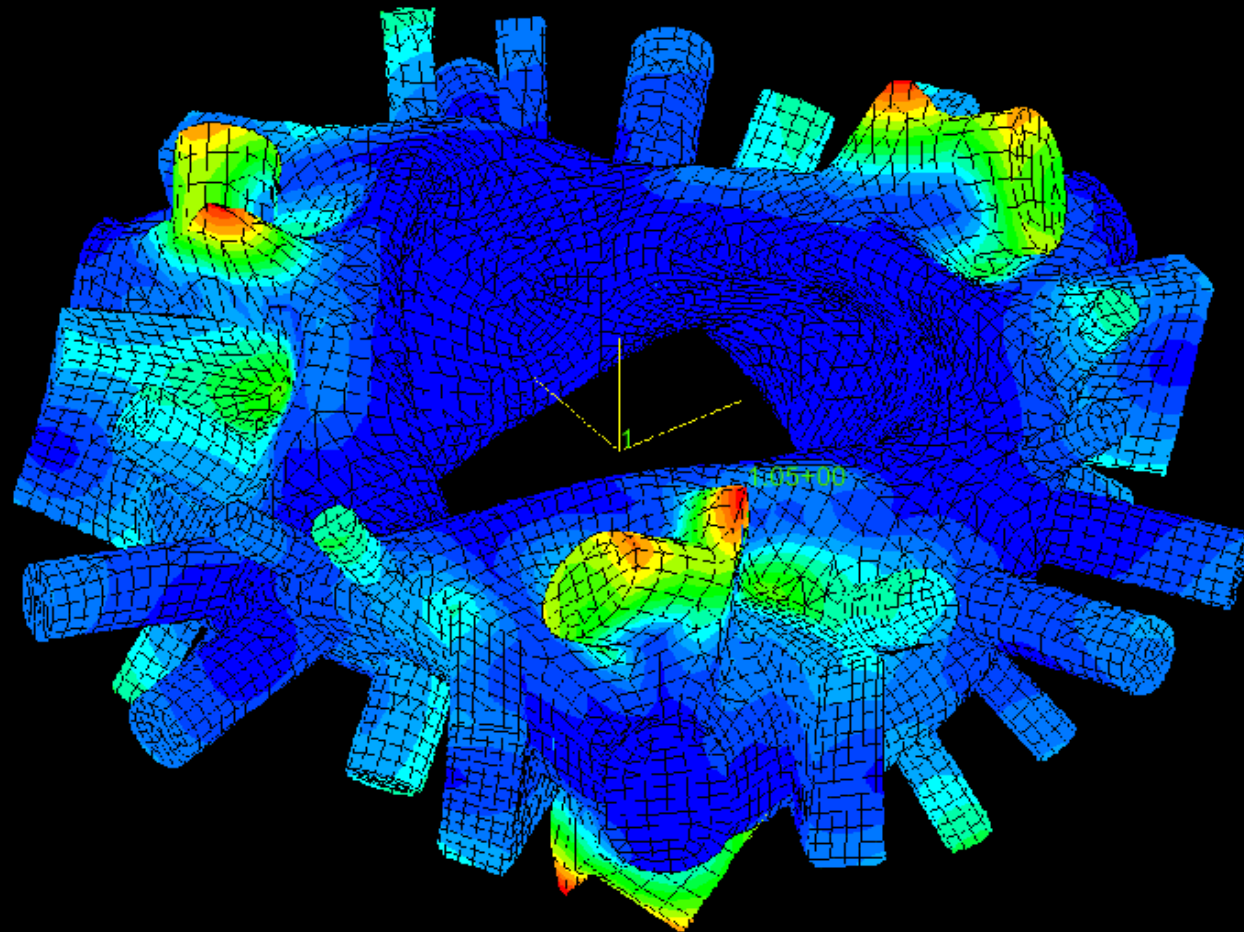
2.13-01

1.43-01

7.35-02

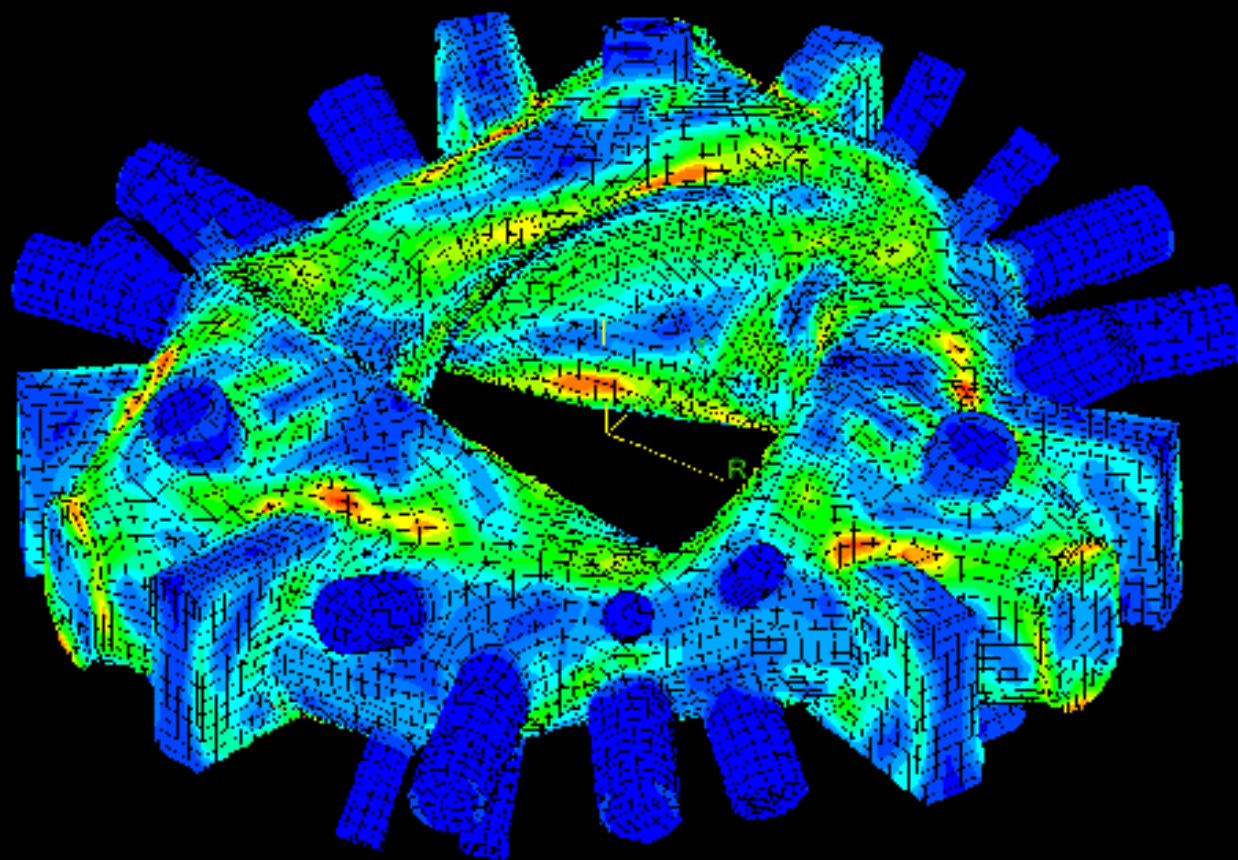
3.57-03

# Shell Thickness = 0.250



default\_Fringe :  
Max 1.05+00 @Nd 9776  
Min 3.57-03 @Nd 7467  
default\_Deformation :  
Max 1.05+00 @Nd 9776

Shell Thickness = 0.250



1.34+04

1.25+04

1.16+04

1.08+04

9.86+03

8.97+03

8.08+03

7.19+03

6.30+03

5.41+03

4.52+03

3.63+03

2.74+03

1.85+03

9.62+02

7.19+01

default\_Fringe :  
Max 1.34+04 @Nd 7850  
Min 7.19+01 @Nd 12364

## Conclusions:

- Stresses are generally at or below allowables (ASME-Grade1 annealed Inconel-625: 27.5ksi, ASTM-B-443).
- Local buckling at NB port nozzle to shell transition  $\lambda = 2.9$  for 0.175" thk. Shell
- $\lambda = 6.7$  for 0.250" thick Shell
- May need to increase shell thickness to a 0.220" minimum thickness to meet code S.F. of 5x